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THE UNIVERSITY OF ALBERTA

AN INVESTIGATION OF THE LOG PERIODIC ANTENNA AS A LEAKY WAVE DEVICE

by

Frederick T. Halsey

A THESIS

SUBMITTED TO THE FACULTY OF GRADUATE STUDIES

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## ABSTRACT

Near field measurements available in the literature show that two waves exist on log periodic antennas. The first, called the transmission line wave originates at the apex, propagates along the antenna and decays in the active region. The second, called the radiation wave originates in the active region, propagates back toward and beyond the apex, and becomes the far field at a suitable distance in front of the antenna.

The hypothesis is made that the effective radiating aperture is located at the front of the active region and is in the plane perpendicular to the antenna structure. The aperture distribution is obtained by considering the radiation wave to behave as a leaky wave in the active region. The far field pattern is found from this aperture distribution using a transform technique. Suitable assumptions are made relating the characteristics of the radiation and transmission line waves in the active region. Thus, the far field pattern can be calculated if the properties of the transmission line wave are known.

The transform used is valid only in the half plane to the front of the antenna. Thus, the theory will predict only the shape of the main lobe since no information is obtained regarding the energy radiated to the rear of the antenna.

The characteristics of the transmission line wave for the log periodic dipole antenna were measured for various parameters of the antenna and these used to calculate the far field patterns.

# INTRODUCTION

The following report was prepared for the purpose of

presenting the results of the investigation conducted by the

author in connection with the study of the

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The following is a summary of the results of the

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These calculated patterns were found to show reasonably good agreement with experimentally obtained patterns over the useable range of parameters for this antenna.





## ACKNOWLEDGEMENT

The author wishes to thank Professor W. Sinnema for numerous helpful and thought provoking discussions which were held during the course of this investigation.



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## 1. INTRODUCTION

Since the concept of frequency independent antennas was put forward by V. H. Rumsey<sup>1</sup> this class of antennas has received considerable attention in the literature. The logarithmically periodic (or log periodic) class of antennas introduced by R. H. DuHamel was one of the outgrowths of this interest.<sup>2</sup>

In general, the investigation of the properties of log periodic antennas has been carried out experimentally. However, in order to understand their radiation mechanism and thus obtain optimum design principles, a theoretical analysis is clearly required. R. L. Carrel has made an excellent analysis of the log periodic dipole antenna.<sup>3</sup> Briefly, he considered the self and mutual impedances of the elements making up the structure, and with the aid of a computer, was able to predict the behavior of this antenna with a good degree of accuracy. While the theoretical considerations of this approach are straight forward, the numerical solution of the resulting equations is formidable and would be practically impossible to obtain without the use of a computer. P. E. Mayes, G. A. Deschamps and W. T. Patton have explained the operation of unidirectional periodic structures in terms of backward wave radiation.<sup>4</sup> These antennas can be converted to log periodic antennas by tapering the elements. They have shown that the radiation patterns for the two types of structures are similar. It is their intention to consider the log periodic antenna as a slowly varying periodic structure and apply the theory of the

There are several reasons for the interest in the study of the structure of the human brain. The first reason is that the brain is the most important organ in the human body. It is the organ that controls all the functions of the body. The second reason is that the brain is the most complex organ in the human body. It is the organ that is responsible for all the thoughts, feelings, and actions of the human being. The third reason is that the brain is the most vulnerable organ in the human body. It is the organ that is most susceptible to injury and disease. The study of the structure of the human brain is therefore of great importance to the medical profession and to the general public.

The purpose of this book is to provide a comprehensive overview of the structure of the human brain. It will discuss the various parts of the brain, their functions, and the ways in which they are interconnected. It will also discuss the ways in which the brain is affected by injury and disease, and the ways in which it can be protected and treated. The book is intended for students of the medical profession, for researchers in the field of neuroscience, and for anyone who is interested in the human brain.

The book is divided into three main parts. The first part, which is the largest, is devoted to the anatomy of the brain. It describes the various parts of the brain, their locations, and their functions. It also discusses the ways in which the brain is protected by the skull and the meninges, and the ways in which it is supplied with blood and oxygen. The second part is devoted to the physiology of the brain. It discusses the ways in which the brain controls the functions of the body, and the ways in which it is affected by various factors such as stress, fatigue, and drugs. The third part is devoted to the pathology of the brain. It discusses the ways in which the brain can be affected by injury and disease, and the ways in which it can be treated.

The book is written in a clear and concise style, and it is illustrated with numerous diagrams and photographs. It is intended to be a useful reference for students and researchers, and it is also intended to be an interesting read for anyone who is interested in the human brain.



periodic structure to it.

The approach taken in this paper is to determine the far field radiation patterns for log periodic antennas from the near fields existing on the antenna.

Near field measurements available in the literature show that two waves exist on log periodic antennas.<sup>5</sup> The first, called the transmission line wave originates at the apex, propagates along the antenna, and decays in the active region. The second, called the radiation wave originates in the active region, propagates back toward and beyond the apex, and becomes the far field at a suitable distance in front of the antenna.

The hypothesis is made that the effective radiating aperture is located at the front of the active region and is in the plane perpendicular to the antenna structure. The aperture distribution is obtained by considering the radiation wave to behave as a leaky wave in the active region. The far field is found from this aperture distribution using a transform technique. Assumptions relating the characteristics of the radiation and transmission line waves are made by considering the current distribution on the elements in the active region. Thus, the far field radiation patterns can be calculated if the properties of the transmission line field are known.

In this paper, the properties of the transmission line wave for the log periodic dipole antenna are obtained experimentally. A theoretical analysis of these properties is being carried out in





conjunction with the work of Mayes, Deschamps and Patton previously described. A preliminary study of a class of log periodic circuits has been made by R. Mittra.<sup>6</sup>

The presentation of this paper is as follows: The preceding section briefly reviews the theoretical work, known to this investigator, which has been done on these antennas to date. A review of frequency independent antennas briefly describing some of the better known structures and their properties is presented in Section 2. A detailed formulation of the theory is given in Section 3. The measuring techniques used to obtain the characteristics of the transmission line field are outlined in Section 4. The results for measurements made on a log periodic dipole antenna are presented in Section 5. Calculated patterns using the theory developed in Section 3 are compared to radiation patterns obtained experimentally and the results analysed. Section 6 gives a summary of the paper. Two appendices are attached. The first deals with the solution of an integral obtained in Section 3. The measurements of the transmission line wave along the structure for various parameters of the log periodic dipole antenna are presented graphically in the second appendix. The characteristics of the transmission line wave are obtained from these curves.









## 2. A REVIEW OF FREQUENCY INDEPENDENT ANTENNAS

The logarithmically periodic antenna belongs to a general class of antennas usually referred to as "frequency independent antennas". In practice, this class of antennas can be built to give broadband performance over frequency ranges as great as 40:1 or more, provided the structure is carefully modelled in accordance with the basic design principles. Admittedly, these are not the only antennas which can be designed to cover a wide band of frequencies. For example, the discone antenna, the unbalanced conical helix fed against ground, and the Archimedean spiral, can be designed to cover frequency ranges of 4:1, 5:1, and 10:1 respectively.<sup>7</sup> However, the name "frequency independent antennas" is justified since these are the only structures possessing a theoretical independence of frequency. Since their upper and lower frequency limits are determined by independent parameters, the bandwidth of a specific antenna can be increased almost without limit simply by adding more sections to the antenna.

### 2.1 Basic Formulation

The concept that an antenna specified by angles alone would have characteristics independent of frequency was originally proposed by V. H. Rumsey.<sup>1</sup> The following is a brief résumé of a simplified analysis<sup>8</sup> paralleling Rumsey's earlier work.

Consider an antenna whose terminals are indefinitely close to the origin of a spherical coordinate system, being symmetrically



disposed along the  $\Theta = 0, \pi$  axis. Let the surface of this antenna be described by

$$\mathcal{L} = F(\Theta, \phi) \quad (1)$$

For an antenna of finite thickness, several branches of  $F(\Theta, \phi)$  can be taken, corresponding to the inner and outer surfaces.

Suppose it is desired to scale this antenna to a new frequency which is  $K$  times lower than the original frequency. The antenna must be made  $K$  times larger, resulting in a surface

$$\mathcal{L}' = K F(\Theta, \phi) \quad (2)$$

in which  $K$  depends neither on  $\Theta$  nor on  $\phi$ .

Imagine that when this is done the new surface is congruent (identical) to the original. This implies that both surfaces are infinite. It can be seen that congruence, if it occurs, can be established through a rotation in  $\phi$ . Translation is barred since both surfaces have their terminals at the same origin and rotation in  $\Theta$  is barred because both pairs of terminals are symmetrically disposed along the  $\Theta = 0, \pi$  axis. Thus, for congruence,

$$K F(\Theta, \phi) = F(\Theta, \phi + C) \quad (3)$$

in which  $C$  is the angle through which the second antenna must be rotated in order to achieve congruence with the first.  $C$  depends on  $K$ , but neither depends on  $\Theta$  or  $\phi$ .

Congruency implies that the original antenna would perform exactly the same at both frequencies, except for a rotation,  $C$ , in the azimuthal coordinate of its radiation pattern. If it develops that  $K$  is unrestricted, i.e.,







$$0 \leq K \leq \infty$$

(4)

then the original antenna must have pattern and impedance characteristics which are independent of frequency. The pattern may rotate in  $\phi$  with frequency due to the parameter  $C$ , but its shape will be unaltered.

The nature of  $F(\theta, \phi)$  may be deduced by equating the two equations obtained from differentiating (3) with respect to  $C$  and  $\phi$ . This gives, with the aid of (1),

$$\frac{1}{K} \frac{dK}{dC} = \frac{1}{\mathcal{L}} \frac{\partial \mathcal{L}}{\partial \phi} \quad (5)$$

Since the left side of (5) is independent of  $\theta$  and  $\phi$ , it follows that

$$\mathcal{L} = F(\theta, \phi) = e^{a\phi} f(\theta) \quad (6)$$

$$\text{where } a = \frac{1}{K} \frac{dK}{dC}$$

is a general solution, in which  $a$  is a parameter and  $f(\theta)$  is a completely arbitrary function.

Equation (6) was first derived by V. H. Rumsey.<sup>1</sup> Any antenna shape whose surfaces can be described by functions of the form (6) should be theoretically independent of frequency. This equation requires an infinite structure in order to achieve frequency independence over an infinite spectrum of frequencies. However, it was found experimentally for certain structures that the current distribution along the antenna falls to a negligible value at some number of wavelengths from the feed point. Therefore, the antenna



can be truncated at this point without affecting the current distribution, and therefore, the characteristics of the antenna. The frequency for which this is done represents the low frequency limit of the antenna since the current distribution moves towards the feed point as the frequency is increased. The high frequency limit of the antenna occurs when the feed point ceases to look like a "point" due to the physical impossibility of modelling the structure to an indeterminately small size. As previously stated, the bandwidth can be increased simply by continuing the structure until the desired frequency is included within the range of the antenna.

## 2.2 The Equiangular Spiral Antenna

The equiangular spiral antenna was the first antenna of a practical size to exhibit the characteristics of the infinite structure.<sup>9</sup> The equation of the equiangular spiral may be obtained from (6) by letting

$$\frac{df(\theta)}{d\theta} = f'(\theta) = A \delta(\theta - \frac{\pi}{2}) \quad (7)$$

where  $A$  is an arbitrary positive constant and  $\delta$  is the Dirac delta function. Substituting (7) in (6) yields

$$r = r_0 e^{\alpha(\phi - \phi_0)} \quad (8)$$

in the plane  $\theta = \frac{\pi}{2}$ ,  $r$  being zero elsewhere.  $r_0 e^{-\alpha\phi_0}$  is a substitution for  $A$ . Expression (8) is the equation of a planar equiangular (or logarithmic) spiral. If the angle  $\phi$  is increased by one full turn, the radius vector,  $r$  is increased by the factor  $e^{2\pi\alpha}$ . Hence, each turn of the spiral is identical with every







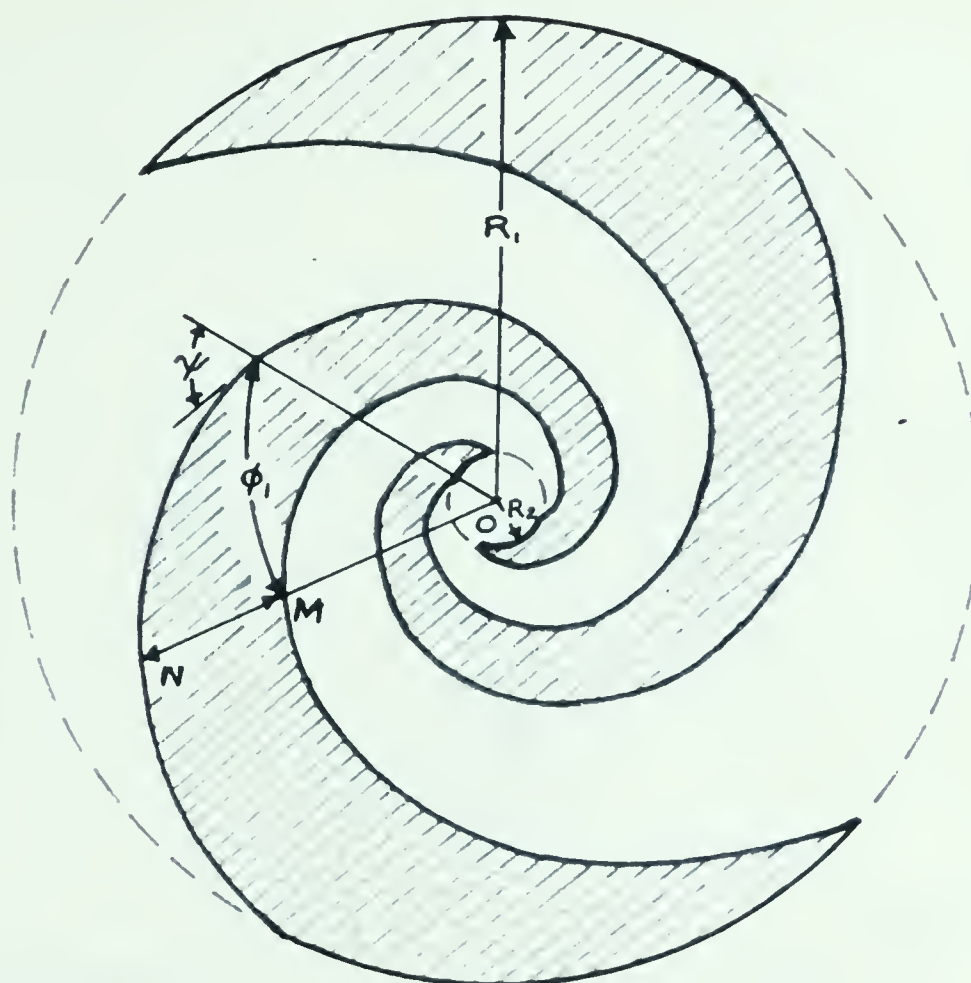
other turn except for a constant multiplier. The angle,  $\psi$ , between the radius vector, and the tangent to the equiangular spiral gives the rate of spiral and is a constant for a particular curve.

Since  $A$  is arbitrary, the choice can be made to fix  $\lambda_0$  and leave  $\phi_0$  as a parameter. If  $\phi_0$  is allowed to assume all values from  $0$  to  $\phi_1$  and all values from  $\pi$  to  $\pi + \phi_1$ , the antenna of Figure 1 results. Since (8) yields an infinite structure, the minimum and maximum radii,  $R_2$  and  $R_1$ , must be specified for a practical antenna. These correspond to the upper and lower frequency limits of the antenna, respectively. Typical values for  $R_2$  and  $R_1$  are one eighth and one half wavelengths, respectively, at the upper and lower frequencies.

Two forms of this antenna have been used, the plane conductor antenna made of metallic arms suspended in free space, and the slot antenna consisting of spiral slots cut in a large conducting sheet.

The antenna may be fed at the centre by a balanced line. However, this causes some distortion of the fields because of the feed line. A preferred method is to imbed the coaxial feed cable in one arm of the spiral and connect the centre conductor to the other arm of the spiral as shown in Figure 2. This results in a perfectly balanced feed system. This method of feed, often referred to as the "infinite balun", is made possible by the rapid attenuation of the currents on the arms. It is the only form of the

• 1998



$\psi$  =Rate of Spiral

$$r_1 = OM = r_0 e^{a(\phi - \phi_1)}$$

$$r_2 = ON = r_0 e^{a\phi}$$

$$\frac{r_2}{r_1} = e^{a\phi_1}$$

FIGURE 1

OUTLINE AND BASIC PARAMETERS  
OF THE TWO ARM EQUIANGULAR SPIRAL





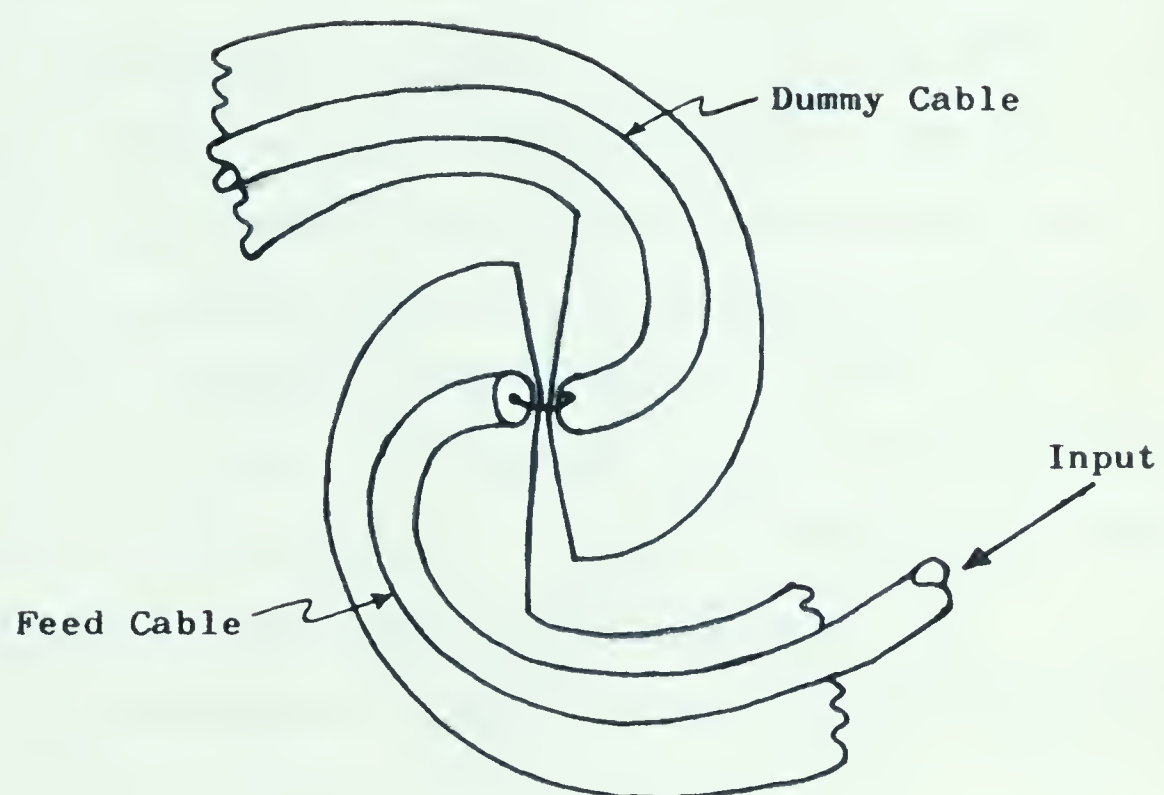


FIGURE 2

FEED REGION FOR A TYPICAL EQUIANGULAR SPIRAL  
SHOWING THE "INFINITE BALUN CONSTRUCTION"



presently known which permits full utilization of the "infinite" frequency properties of this antenna. To maintain symmetry, a dummy cable is sometimes soldered to the other arm of the spiral.

This antenna produces two equal broad lobes to the front and back of the structure, typical beamwidths being of the order sixty to one hundred degrees. The patterns do not change much with the parameters  $a$  and  $\phi_1$ . They are nearly circular for high rates of spiral, but show some assymetry for low rates of spiral which produces a rotation effect as the frequency is varied. The field is practically circularly polarized over the operating range of the antenna.

A typical input impedance of 120 ohms with a VSWR of less than 2:1 over the frequency range is obtained for  $\phi_1 = 90^\circ$ .<sup>10</sup>

B. R. S. Cheo, V. H. Rumsey, and W. J. Welch have made a theoretical analysis of this antenna.<sup>11</sup> Design data is available.<sup>10</sup>

### 2.3 The Conical Equiangular Spiral Antenna

A slightly less restrictive choice of  $f'(\theta)$  is given by

$$f'(\theta) = A \delta(\theta - \theta_0) \quad (9)$$

where  $\theta_0$  is any angle in the range  $0 \leq \theta_0 \leq \frac{\pi}{2}$ . Substituting in (6) yields

$$r = r_0 e^{a_1(\phi - \phi_0)} \quad (10)$$

where  $A = r_0 e^{-a_1 \phi_0}$  for the cone of revolution  $\theta = \theta_0$ ,  $r$  being zero elsewhere. This is the equation of an equiangular spiral projected on to a cone of apex angle  $2\theta_0$ . As was done for the equiangular spiral antenna in the previous subsection the choice is made to fix  $r_0$  and allow  $\phi_0$  to assume all values from 0 to  $\phi_1$  and all values from  $\pi$  to  $\pi + \phi_1$ . The outline for this particular type of

presently known solid bodies will withstand the collision  
 frequency properties of this system. In addition, however,  
 direct tests in laboratory conditions for the case of the  
 solid system require two sets of data from the first and  
 back of the structure, typical mechanical tests of the system under  
 one hundred degrees. The system is not known with the  
 parameters  $\Phi$  and  $\Psi$ . Thus we must assume the high order of  
 approximation, but also we are assuming for the case of solid system  
 a relation effect on the system is given. The first is  
 essentially completely isolated from the surrounding space of the system.  
 A typical inner structure of the system is given in the form of  
 2:1 over the frequency range is defined by  $\frac{\omega}{\omega_0} = 1.1$ .  
 The system is given by  $\omega_0 = 1.1$  and  $\omega = 1.1$  with the same  
 theoretical analysis of this system. The system is given by

### 2.2 The Linear Nonlinear System

A slightly less realistic model of a linear system is

$$\ddot{x} + \delta \dot{x} + \omega_0^2 x = \omega_0^2 A \cos \omega t$$

where  $\omega_0$  is the natural frequency of the system,  $\delta$  is the damping coefficient, and  $A$  is the amplitude of the forcing function.

yield

$$x = \frac{A}{\sqrt{1 - \delta^2/\omega_0^2}} \cos(\omega t - \phi)$$

where  $\phi = \arctan \frac{\delta \omega}{\omega_0^2 - \omega^2}$  for the case of resonance  $\omega = \omega_0$ , it is noted that  
 elsewhere. This is the condition of an undamped system which  
 on to a case of zero angle  $\phi = 0$ . It is also for the resonance  
 which means in the linear approximation the case is seen in the  
 $A_0$  and also  $\phi_0$  to remain the same for  $\omega = \omega_0$  and  $\delta = 0$ .  
 To get the value of  $\phi_0$  for this particular type of



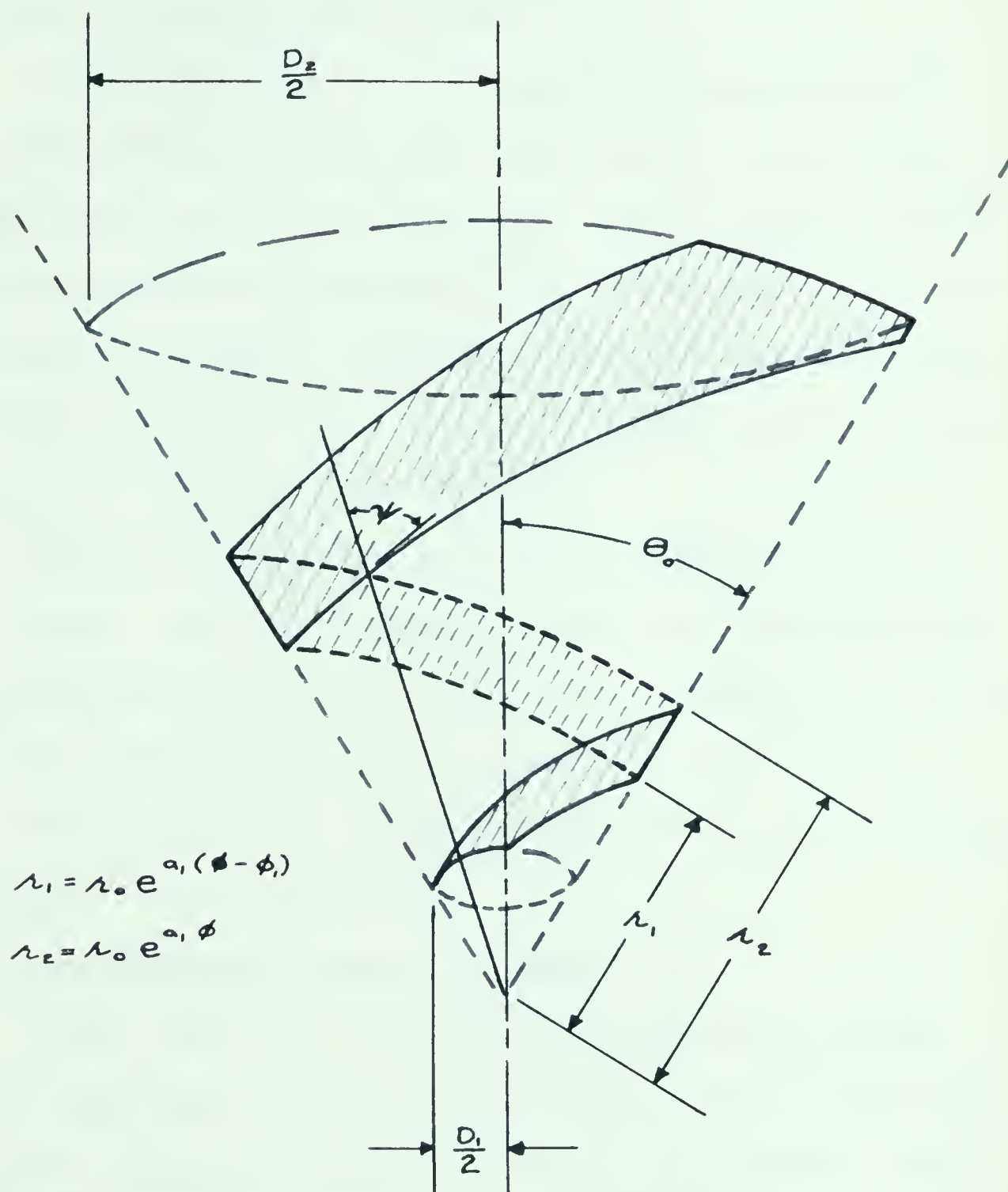


FIGURE 3

OUTLINE AND BASIC PARAMETERS OF  
THE CONICAL EQUIANGULAR SPIRAL ANTENNA  
(ONE ARM ONLY SHOWN)



antenna, shown in Figure 3, represents the projection of the planar spiral of Figure 1 on to the conical surface  $\Theta = \Theta_0$ . For the sake of clarity, only one arm is shown.

J. D. Dyson studied this antenna experimentally.<sup>12</sup> He found that the additional parameter,  $\Theta_0$ , exerts a considerable influence on the shape of the radiation field. For the proper choice of  $\Theta_0$  the pattern becomes unidirectional with the major lobe in the direction of the apex. The beamwidth varies widely with the parameters, typical values being  $60^\circ$  to  $200^\circ$  when  $\psi$  is varied from  $82^\circ$  to  $45^\circ$  ( $\Theta_0 = 10^\circ$  and  $\phi_1 \approx 94^\circ$ ).<sup>13</sup>

The "infinite balun" previously described is used for the feed system. The mean impedance ranges from 153 to 130 ohms when the apex angle is varied from  $60^\circ$  to  $20^\circ$ . VSWR's of less than two to one are obtained over a six to one frequency range.

The design of this antenna is similar to that of the planar equiangular spiral antenna.<sup>10</sup>

#### 2.4 Logarithmically Periodic Antennas

Since  $\Theta_0$  in (9) is fixed for a particular antenna, the conical equiangular spiral antenna is essentially a two dimensional structure wrapped on a conical surface. If, however,  $\Theta_0$  is allowed to assume all values from  $\Theta_1$  to  $\Theta_2$  ( $0 \leq \Theta_1 < \Theta_2 \leq \frac{\pi}{2}$ ) the three dimensional conical screw results.

R. H. DuHamel conceived the idea of placing two conical screws apex to apex, collinearly, and taking the cut,  $\phi = \text{constant}$ .<sup>2</sup> This yields the planar antenna of Figure 4. This concept is extremely

...; hence, in Table 1, ... of ... ..  
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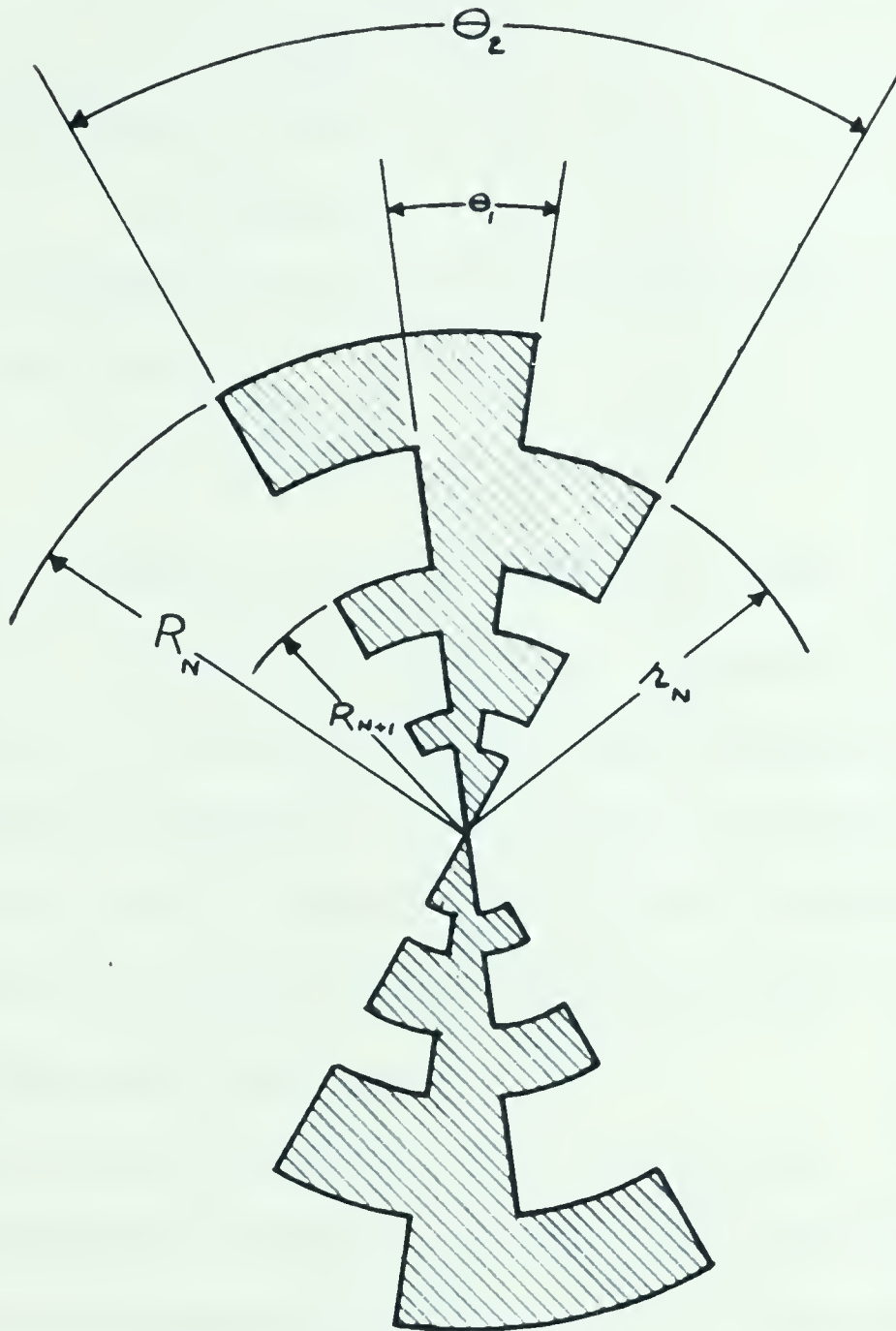


FIGURE 4

THE PLANAR SHEET CIRCULAR TOOTH  
LOGARITHMICALLY PERIODIC ANTENNA



valuable since it provides a linearly polarized frequency independent antenna. The shape of these antennas can be described by

$$L = f(\theta) \quad (11)$$

A new antenna scaled to give

$$L = K f(\theta) \quad (12)$$

will only be congruent to the original antenna for a discrete set of values for  $K$  given by

$$K = e^{2\pi n \alpha_1} \quad (13)$$

where  $n$  is an integer and  $\alpha_1$  is defined in (10). Therefore, the properties of the antenna will repeat with period  $2\pi\alpha_1$ . Hence, structures of this nature are called logarithmically periodic (or log periodic) antennas. If the variation of the impedance and pattern is small over a period, and therefore, all periods because of the repetitive characteristics, then the result is essentially a frequency independent antenna.

There is an infinite variety of choices for  $f(\theta)$  and hence a corresponding infinite variety of log periodic structures. A theoretical study of the best choice for  $f(\theta)$  seems to be nearly impossible and to date, the initial evaluation of these structures has been done experimentally. Some of the more successful structures are shown in Figure 5.

The parameters for this class of antennas are shown in Figure 6. The symbol  $\theta$  is used to define the apex angle since the symbol  $\alpha$ , normally used in the literature, will be used to

whereas when it is a function of the position of the system, it is a function of the position of the system. The type of the system can be described by

$$(11) \quad \mathcal{L} = \mathcal{L}(x, \dot{x}, t)$$

A new minimum is found to exist

$$(12) \quad \mathcal{L} = \mathcal{L}(x, \dot{x}, t)$$

will only be a minimum if the system is in a state of

of order  $n$  or

$$(13) \quad \mathcal{L} = \mathcal{L}(x, \dot{x}, t)$$

where  $n$  is an integer and  $0 \leq n \leq 100$ . Therefore, the properties of the system will change with time (27). The properties of the system are called *nonlinear* properties for the periodic motion. It is the variation of the system and motion is small over a period, and then the system is called *linear* at the repetition of the motion, and the system is called *nonlinear* at the repetition of the motion.

There is an infinite number of systems for  $n \leq 100$ . Since a corresponding infinite number of the systems exist, a theoretical study of the system for  $n \leq 100$  seems to be impossible and to exist. The infinite number of the systems has been shown experimentally. Some of the experimental results are shown in figure 1.

The properties of the system at different times are shown in figure 1. The system is in a state of order  $n$  at the time  $t$ . The system is in a state of order  $n$  at the time  $t$ . The system is in a state of order  $n$  at the time  $t$ .



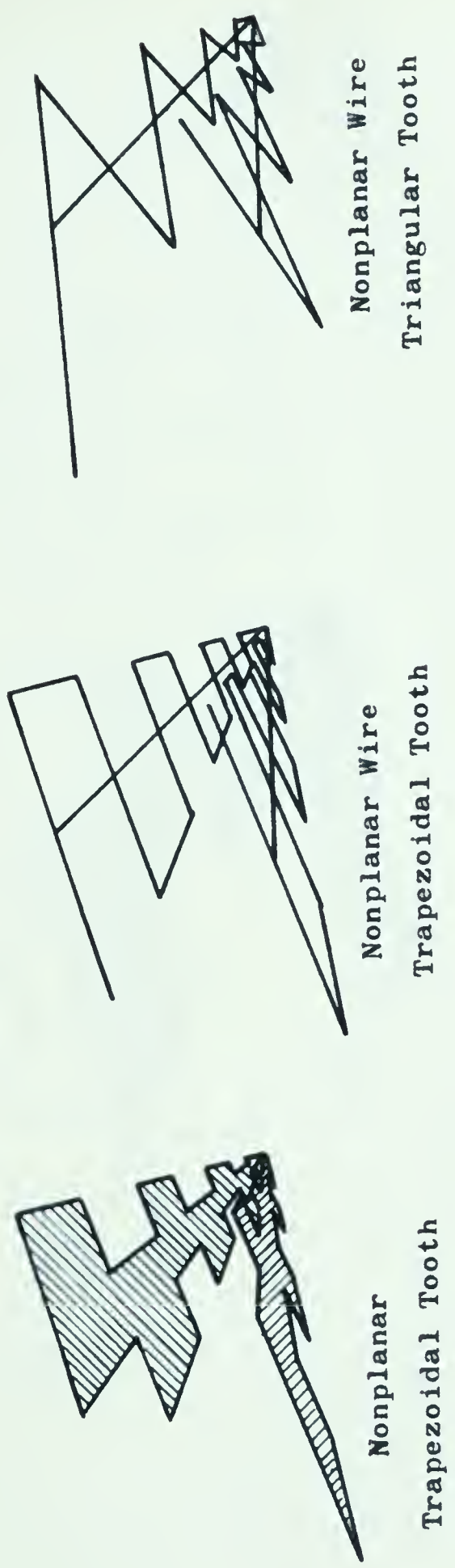
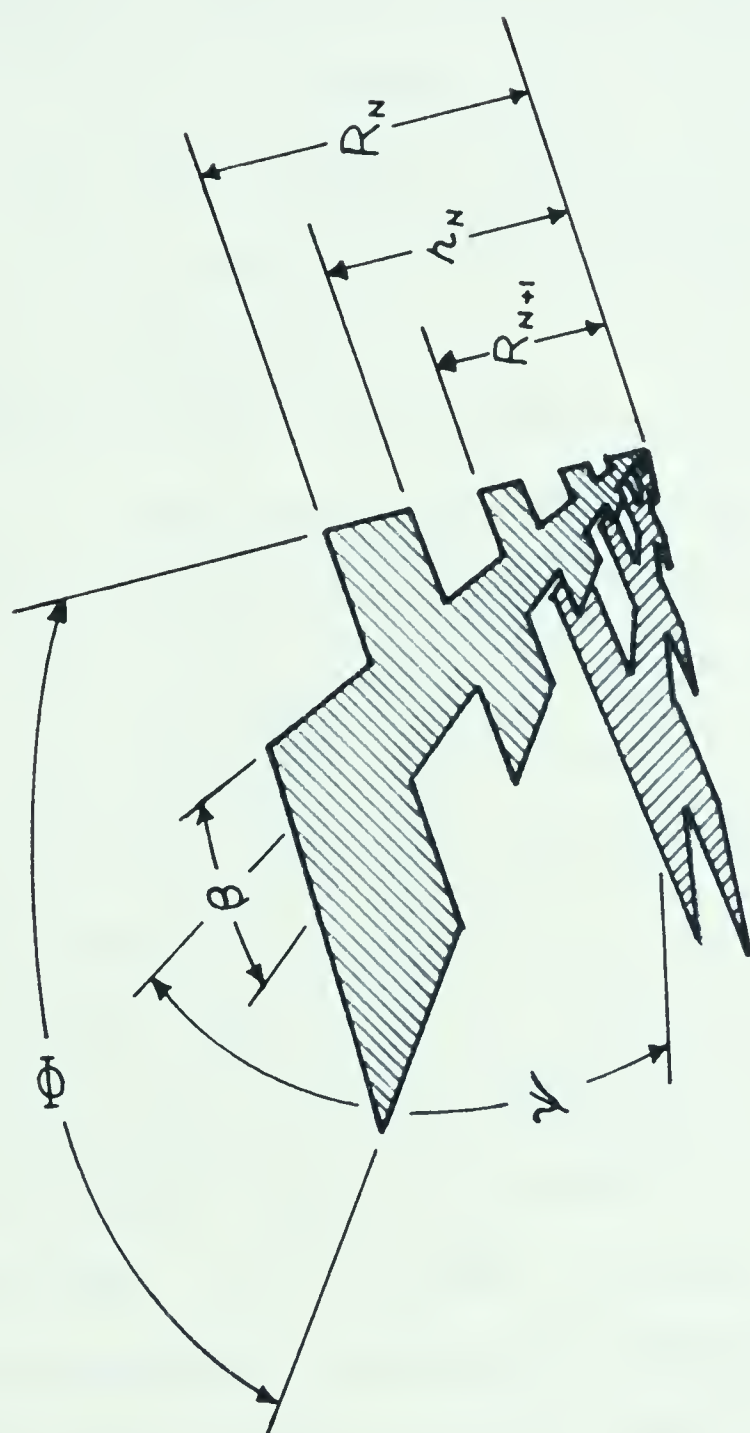


FIGURE 5  
SEVERAL TYPES OF LOG PERIODIC ANTENNAS





$$\gamma' = \frac{R_{N+1}}{R_N}$$

$$r_N = \sqrt{\gamma'} R_N$$

FIGURE 6

BASIC PARAMETERS OF THE LOG PERIODIC ANTENNA





denote the attenuation constant introduced later. The geometric ratio is defined as

$$\tau' = e^{2\pi\alpha_1} = \frac{R_{N+1}}{R_N} \quad (14)$$

Since the radiation from these antennas is approximately centred on the teeth which are near a quarter wavelength in length, the upper and lower frequency limits of these antennas occur when the shortest and longest teeth are approximately a quarter wavelength long respectively.

Theoretically, the thickness of the sheets and the diameters of the wires should increase linearly with distance from the apex. In practice, this is not necessary for bandwidths less than five to one.<sup>10</sup> With tapering, bandwidths of twenty to one are easily achieved, and bandwidths of one hundred to one are feasible if the theoretical design is closely approximated.

The radiation patterns of the log periodic antennas shown in Figure 5 are similar for the same design parameters,  $\tau'$ ,  $\theta$  and  $\psi$ . They give a unidirectional pattern pointing towards the apex. In general, higher gains are obtained by increasing  $\tau'$  and  $\psi$  and decreasing  $\theta$ . If  $\tau'$  is made too small, the patterns will break up excessively over a period. For the parameters  $\tau' = 0.707$ ,  $\theta = \psi = 45^\circ$ , a gain of 9.45 db. over an isotropic radiator is obtained for a wire tooth logarithmically periodic antenna. This particular antenna has a beamwidth of  $66^\circ$  for both the E and H planes, and a maximum sidelobe level of - 12.3 db.<sup>14</sup>

being the same as the one in the previous section.

Let us now consider

$$\frac{1}{\Gamma} \left( \frac{1}{\Gamma} \right)^{\frac{1}{\Gamma-1}} = \frac{1}{\Gamma} \left( \frac{1}{\Gamma} \right)^{\frac{1}{\Gamma-1}}$$

(11)

Since the function  $f(x) = x^{1/\Gamma}$  is concave for  $\Gamma > 1$ ,

we have by Jensen's inequality

and by the definition of  $\Gamma$  we have

for all  $x > 0$  and  $\Gamma > 1$ .

It follows that

hence, the function  $f(x) = x^{1/\Gamma}$  is concave for  $\Gamma > 1$ .

Of the other hand, let us consider the function  $f(x) = x^{1/\Gamma}$  for  $\Gamma < 1$ .

In this case,  $f(x)$  is convex for  $\Gamma < 1$  and we have

hence, the function  $f(x) = x^{1/\Gamma}$  is convex for  $\Gamma < 1$ .

Moreover, we have

hence, the function  $f(x) = x^{1/\Gamma}$  is concave for  $\Gamma > 1$ .

The function  $f(x) = x^{1/\Gamma}$  is concave for  $\Gamma > 1$  and

is convex for  $\Gamma < 1$ . Hence, the function  $f(x) = x^{1/\Gamma}$  is

convex for  $\Gamma < 1$  and concave for  $\Gamma > 1$ .

In general, we have

hence, the function  $f(x) = x^{1/\Gamma}$  is convex for  $\Gamma < 1$  and

concave for  $\Gamma > 1$ .

Let us now consider the function  $f(x) = x^{1/\Gamma}$  for  $\Gamma > 1$ .

Since the function  $f(x) = x^{1/\Gamma}$  is concave for  $\Gamma > 1$ ,

we have by Jensen's inequality

hence, the function  $f(x) = x^{1/\Gamma}$  is concave for  $\Gamma > 1$ .

These antennas are fed at the apex either by a balanced line or the infinite balun structure previously described. Because of the left-right asymmetry of these structures, the impedance has a periodicity of  $\frac{1}{2} \ln(\sqrt{\tau})$  compared to a periodicity of  $\ln(\sqrt{\tau})$  for the radiation patterns.<sup>10</sup> The input impedance varies widely with the various parameters, and as might be expected, the variation in the input impedance between various geometries with the same design parameters is greater than the variation in the radiation patterns.

Some design data is available in the literature for these antennas.<sup>10,14</sup>

The log periodic dipole antenna, shown in Figure 7 consists essentially of an array of dipoles whose lengths and spacings are arranged in a log periodic manner. This antenna has been investigated, both experimentally<sup>15</sup> and theoretically.<sup>3</sup> Design data, based on the theoretical analysis is available.<sup>3</sup> This antenna can be derived from the antenna of Figure 6 by letting the width of the teeth and the angles  $\gamma$  and  $\theta$  approach zero. The relation between the geometric ratios of the two antennas is given by

$$\tau = \sqrt{\tau'} \quad (15)$$

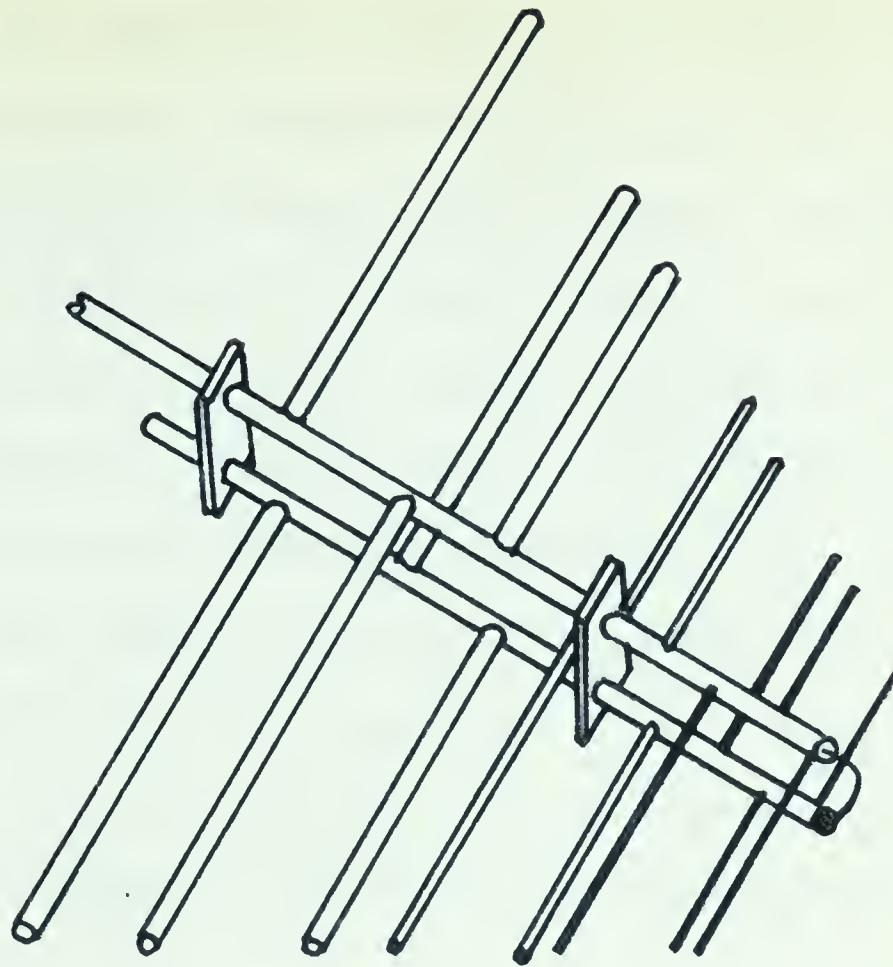
The radiation and impedance characteristics of this antenna are similar to the other types of log periodic antennas.

The log periodic V antenna is an adaptation of the log periodic dipole antenna which takes advantage of the modal and directional characteristics of the conventional V antenna.<sup>16</sup> In the

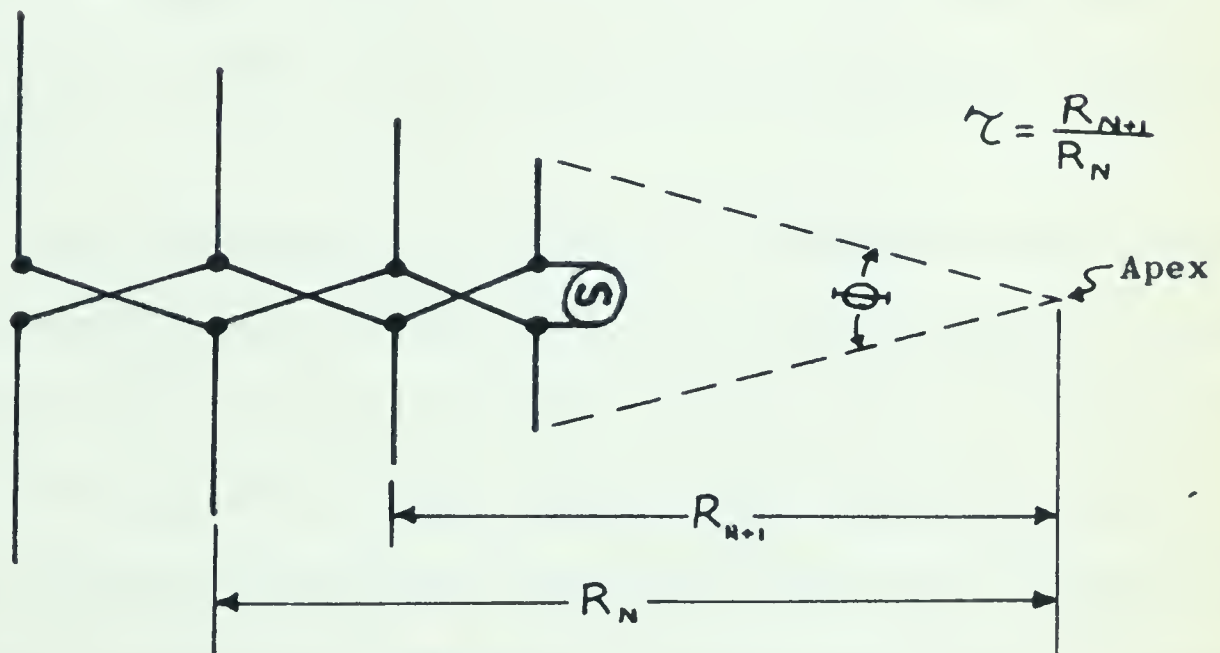








Sketch



Schematic

FIGURE 7THE LOG PERIODIC DIPOLE ANTENNA



lowest order mode, where the energy is radiated from the elements which are approximately half a wavelength long, its characteristics are similar to the log periodic dipole antenna. As the frequency is increased, the radiating region moves toward the apex. If the antenna is designed so that the longest elements are one and a half wavelengths long when the short elements are a half a wavelength long, then as the half wavelength radiating region moves off the antenna, the next higher mode consisting of the elements one and a half wavelengths long will be excited. As the frequency is increased further, this region of radiation will move toward the apex and will eventually be replaced by the two and a half wavelength mode of operation.

Gains as high as 18 db. over an isotropic source have been obtained in the higher modes of operation.<sup>17</sup> The main disadvantage of this antenna is that its characteristics deteriorate between modes.

## 2.5 Summary

The above description of frequency independent antennas has been, by necessity, quite brief and covers only the basic structures.

These antennas may be arrayed to give either an omnidirectional pattern or a high gain for both circular and linear polarization.<sup>10,13,18</sup>

The range of frequencies over which these antennas may be used is nearly unlimited. The lowest limit occurs when the elements become too large and hence difficult to build and support. The highest frequency boundary is usually determined by the accuracy with which the geometry of the structure can be modelled. In some





applications, the high frequency limit may be determined by the power which the antenna is required to handle.

Numerous applications have been made of the principles outlined above to construct wideband antennas for special applications. Reference (7), in particular, contains a list of over seventy-five publications in its bibliography, many of which investigate specific applications for this class of antennas.

Application, 400 West Broadway, New York, N.Y. 10014

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### SECTION 3

THE UNIVERSITY OF CHICAGO PRESS

$$A_1 + A_2 + \dots + A_n = A$$

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$$A_1 + A_2 + \dots + A_n = A$$

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$$A_1 + A_2 + \dots + A_n = A$$





### 3. DERIVATION OF THE RADIATION PATTERNS FOR LOG PERIODIC ANTENNAS

#### 3.1 Review of Surface and Leaky Waves<sup>19</sup>

Surface and leaky wave antennas are part of the larger class of travelling wave antennas. A surface wave antenna supports a surface or trapped wave and does not radiate except at a discontinuity, for example, the termination of the structure. Therefore, the wave number along the structure is real. Conversely, the radiation from a leaky wave antenna occurs at all points along the structure so that the wave number along the structure is complex, containing an attenuation factor.

The relationship governing the parameters is the separability condition for the wave equation which is, in cartesian coordinates,

$$k_x^2 + k_y^2 + k_z^2 = k_o^2 \quad (16)$$

where the  $k$ 's, called the wave numbers, are generally complex.

For example,

$$\begin{aligned} k_i &= -j\gamma_i = \beta_i - j\alpha_i \\ \gamma_i &= jk_i = \alpha_i + j\beta_i \end{aligned} \quad (17)$$

where  $\gamma_i$  is the complex propagation constant,  $\beta_i$  is the phase constant (radians per wavelength) and  $\alpha_i$  is the attenuation constant (nepers per wavelength), all in the  $\hat{z}$  direction. The wave number,

$k_o$ , belongs to the medium in which the wave travels, and is given by

$$k_o = \omega \sqrt{\epsilon \mu} \quad (18)$$



where  $\omega$  is the angular frequency,  $\epsilon$  and  $\mu$  are the dielectric and permeability constants of the medium respectively. In air,  $k_0$  is real.

Using this notation, the variation of the field strength in the  $\hat{i}_x$  direction is given by

$$E_i = E_0 e^{j(-k_i \hat{i} + \omega t)} = E_0 e^{-\gamma_i \hat{i} + j\omega t} \quad (19)$$

Sometimes the time dependent term,  $e^{j\omega t}$ , is suppressed to shorten the expressions. If this is done, its presence is implied.

Consider a surface of infinite extent in the  $y$  direction supporting a surface wave propagating parallel to the surface in the  $z$  direction. For these conditions

$$k_y = 0 \quad k_x = -j\alpha_x \quad (20)$$

since a surface wave decays exponentially away from an infinite surface. Substituting in (16) and equating the real and imaginary terms gives, taking  $k_z = \beta_z - j\alpha_z$ ,

$$\beta_z^2 = k_0^2 + \alpha_x^2 \quad \alpha_z = 0 \quad (21)$$

This means the signal is not attenuated as it propagates along the surface, and thus does not radiate.

If the above surface supports a leaky wave propagating in the  $z$  direction, the conditions become

$$k_y = 0 \quad k_z = \beta_z - j\alpha_z \quad (22)$$

Substituting in (16) gives, taking  $k_x = \beta_x - j\alpha_x$ ,

$$\begin{aligned} \beta_x^2 - \alpha_x^2 + \beta_z^2 - \alpha_z^2 &= k_0^2 \\ \alpha_x \beta_x + \alpha_z \beta_z &= 0 \end{aligned} \quad (23)$$



where  $\mu$  is the number of particles,  $\sigma$  is the cross-section,  $\rho$  is the density, and  $\tau$  is the time interval. The number of particles is given by  $N = \mu \rho \tau$ .

During this interval, the number of particles is given by  $N = \mu \rho \tau$ .

$$N = \mu \rho \tau \quad (1)$$

Substituting the value of  $N$  in equation (1), we get  $N = \mu \rho \tau$ . The number of particles is given by  $N = \mu \rho \tau$ . The number of particles is given by  $N = \mu \rho \tau$ . The number of particles is given by  $N = \mu \rho \tau$ .

$$N = \mu \rho \tau \quad (2)$$

Substituting the value of  $N$  in equation (2), we get  $N = \mu \rho \tau$ . The number of particles is given by  $N = \mu \rho \tau$ . The number of particles is given by  $N = \mu \rho \tau$ .

$$N = \mu \rho \tau \quad (3)$$

The above equation shows that the number of particles is given by  $N = \mu \rho \tau$ . The number of particles is given by  $N = \mu \rho \tau$ . The number of particles is given by  $N = \mu \rho \tau$ .

$$N = \mu \rho \tau \quad (4)$$

$$N = \mu \rho \tau \quad (5)$$



Since  $\alpha_z$  and  $\beta_z$  are of equal sign for attenuation of the wave in the direction of propagation along the surface and  $\beta_x$  is positive for propagation away from the surface,  $\alpha_x$  must be negative. Therefore, the leaky wave field propagates with increasing amplitude away from the surface, becoming infinite at infinity in the direction transverse to the surface. In spite of this improper transverse behavior, leaky waves can be used to provide a valid representation of the fields for certain structures. To date, most antennas analysed using leaky waves have been open waveguide structures.

### 3.2 Interpretation of Near Field Measurements Made on a Log Periodic Antenna

R. L. Bell, C. T. Elfving, and R. E. Franks have made near field measurements on a triangular tooth antenna.<sup>5</sup> They found that two waves exist on these antennas.

The first, called the "transmission line wave" was found to be the transverse electromagnetic (TEM) two wire transmission line wave. This wave originates at the apex and propagates as a slow wave along the antenna without attenuation until it reaches the area where the teeth are approximately one quarter wavelength long. In this region, the wave is exponentially attenuated at a high rate and is thus effectively terminated.

Measurements of the currents on the teeth in this region were made and used to calculate the far field patterns. These

Since  $\omega_0$  and  $\omega_1$  are the angular frequencies of the wave in the direction of propagation along the surface and  $\tilde{\omega}$  is the frequency for propagation away from the surface, we must be cautious. Therefore, the waves were distinguished with localized oscillations away from the surface, localized surface waves in the direction perpendicular to the surface. In spite of this cautionary statement, many waves can be used to study a wide representation of the linear and cubic properties. In fact, some surface waves being body waves have been used previously.

### 3.2 Interference of Two Field Components Due to a

#### Periodic Surface

3.2.1. Let  $\omega_0$  and  $\omega_1$  be the angular frequencies of the wave in the direction of propagation along the surface and  $\tilde{\omega}$  is the frequency for propagation away from the surface. Therefore, the waves were distinguished with localized oscillations away from the surface, localized surface waves in the direction perpendicular to the surface. In spite of this cautionary statement, many waves can be used to study a wide representation of the linear and cubic properties. In fact, some surface waves being body waves have been used previously.

The linear and cubic properties of the surface wave in the direction of propagation along the surface are distinguished with localized oscillations away from the surface, localized surface waves in the direction perpendicular to the surface. In spite of this cautionary statement, many waves can be used to study a wide representation of the linear and cubic properties. In fact, some surface waves being body waves have been used previously.

3.2.2. Let  $\omega_0$  and  $\omega_1$  be the angular frequencies of the wave in the direction of propagation along the surface and  $\tilde{\omega}$  is the frequency for propagation away from the surface. Therefore, the waves were distinguished with localized oscillations away from the surface, localized surface waves in the direction perpendicular to the surface. In spite of this cautionary statement, many waves can be used to study a wide representation of the linear and cubic properties. In fact, some surface waves being body waves have been used previously.



agreed with the measured patterns indicating that the energy was radiated from those elements. Therefore, this region was designated the "active region" of the antenna.

The second wave, named the "radiation wave", originates in the active region, and propagates toward the apex. Inside the active region, it builds up exponentially in the direction of propagation. Outside this region, it propagates with the speed of light toward and beyond the apex, its magnitude decreasing inversely with distance in front of the antenna. This wave becomes the far field at a suitable distance in front of the antenna.

Contour plots for this antenna have been published<sup>5</sup> and are reproduced in Figure 8. To aid in interpreting the radiation wave as a leaky wave, the near field contours for a Yagi-Uda antenna six wavelengths in length have been reproduced from the literature in Figure 9.<sup>20</sup>

It has been shown that a Yagi-Uda structure will support a travelling wave and that a long Yagi-Uda antenna can be treated as a surface wave antenna.<sup>21,22</sup> The hypothesis was made that the cross-sectional area of the field at the end of the antenna represents the effective aperture for the antenna. This approach assumes no direct radiation from the feed and no reflection from the end of the antenna. If the antenna is long enough for the surface wave to become established, the above solution represents the dominant mode of radiation.

As can be seen from Figure 9, the Yagi-Uda antenna supports





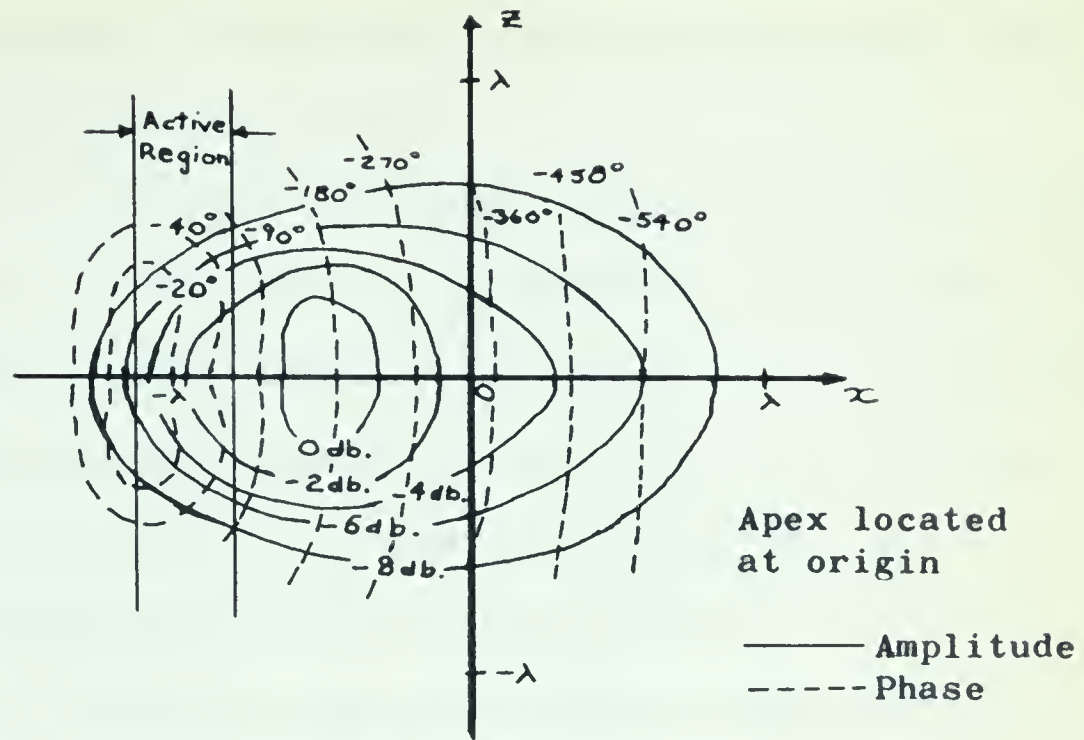


FIGURE 8

THE RADIATION WAVE FOR A LOG PERIODIC ANTENNA

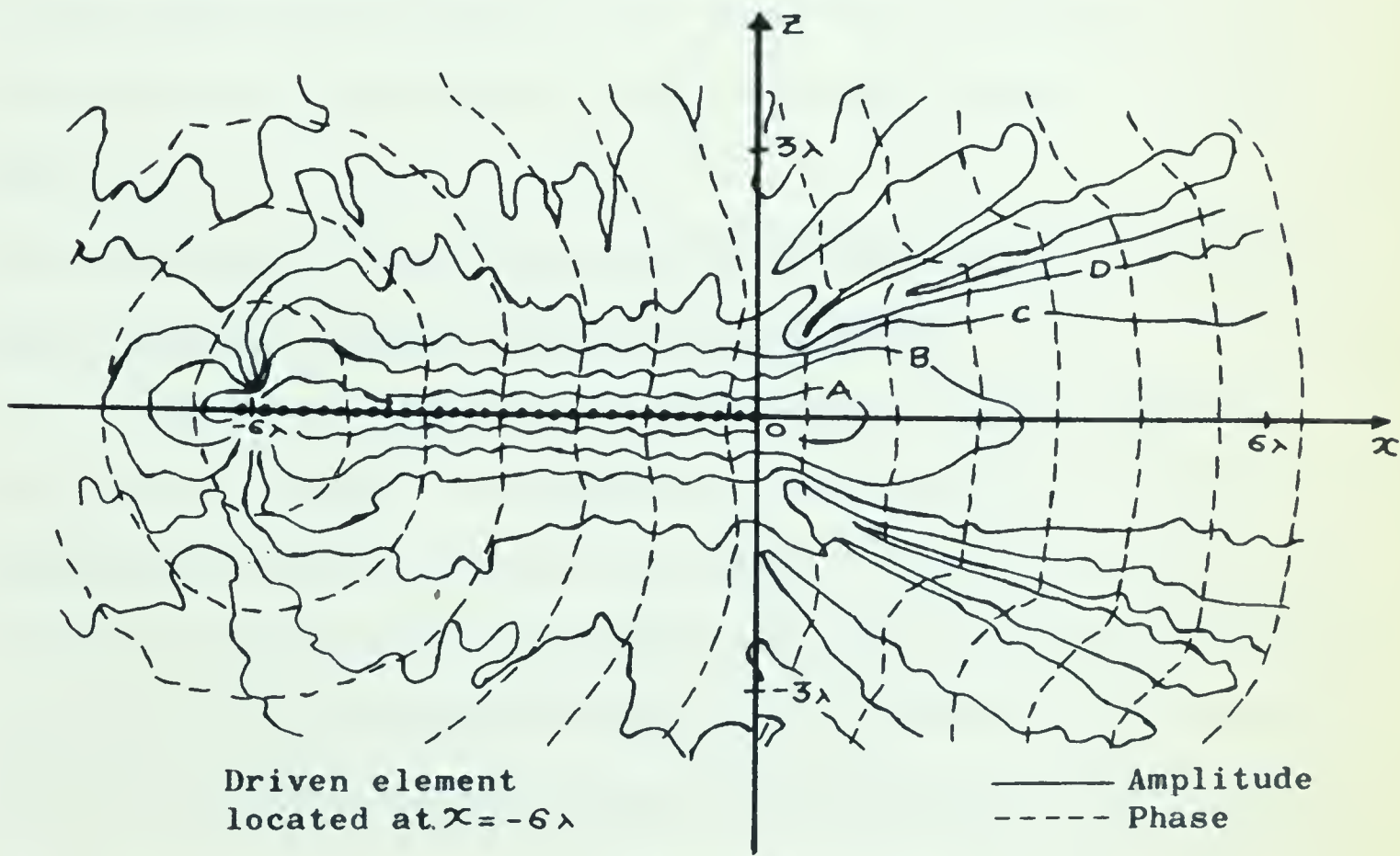


FIGURE 9

NEAR FIELD CONTOURS FOR A YAGI-UDA ANTENNA



a surface wave which forms an equi-phase aperture distribution at the end of the antenna. In front of the antenna, the wave begins to take the form of a diverging wave.

Comparison of the radiation wave for the log periodic antenna, shown in Figure 8, to the field of the Yagi-Uda antenna shows that the radiation wave does not behave as a surface wave in the active region. Instead, the magnitude of this wave increases along the centreline of the antenna in the direction toward the apex and decays in the perpendicular direction away from the centreline of the structure. Its phase progression has two components. The first is along the centreline of the antenna in the direction toward the apex and the second is perpendicular to the centreline in the direction away from the antenna. This behavior is the same as that described on Page 24 for a leaky wave. Therefore, the radiation wave can be treated as a leaky wave inside the active region.

### 3.3 Derivation of the Radiation Patterns for the Log Periodic Dipole Antenna ( $\gamma = 0$ ) in Terms of Leaky Waves

It has been shown in the literature that the radiation patterns for a leaky wave antenna can be calculated from the approximate aperture distribution which is determined only by the leakage of energy along the structure.<sup>23,24</sup> In this section the approximate radiation pattern for the log periodic dipole antenna will be derived using this concept. The hypothesis will be made that the effective radiating aperture is located at the front of





the active region in the plane perpendicular to the antenna structure and that the radiation wave, which passes through this plane, makes up the aperture field.

Consider the log periodic dipole antenna shown in Figure 10. The antenna is oriented along the  $x$  axis with its elements parallel to the  $y$  axis (into and out of page). The active region covers the range  $0 \leq x \leq a$ . For mathematical expediency, the assumption is made that

$$k_y = 0 \quad (24)$$

This is true if the structure is infinite in the  $y$  direction. Physically an infinite structure can be obtained by arraying other identical antennas along the  $y$  axis. If there is no mutual coupling between the antennas, the shape of the radiation pattern in the  $x-z$  plane will not be affected, and so the calculated radiation pattern in the  $x-z$  plane obtained using this approach will be a good approximation to the actual patterns. Thus the problem has been effectively reduced from three dimensions to two dimensions. This simplification allows the fields to be described by exponential rather than Bessel functions.

It was shown in the previous subsection that the radiation wave for a log periodic antenna can be treated as a leaky wave inside the active region. Therefore, the following conditions hold in the region bounded by  $0 \leq x \leq a$  and  $0 \leq z \leq \infty$ ,

$$\begin{aligned} k_x &= -j\gamma_x = \beta_x + j\alpha_x = -j\gamma_R = \beta_R + j\alpha_R \\ k_z &= -j\gamma_z = \beta_z - j\alpha_z \end{aligned} \quad (25)$$



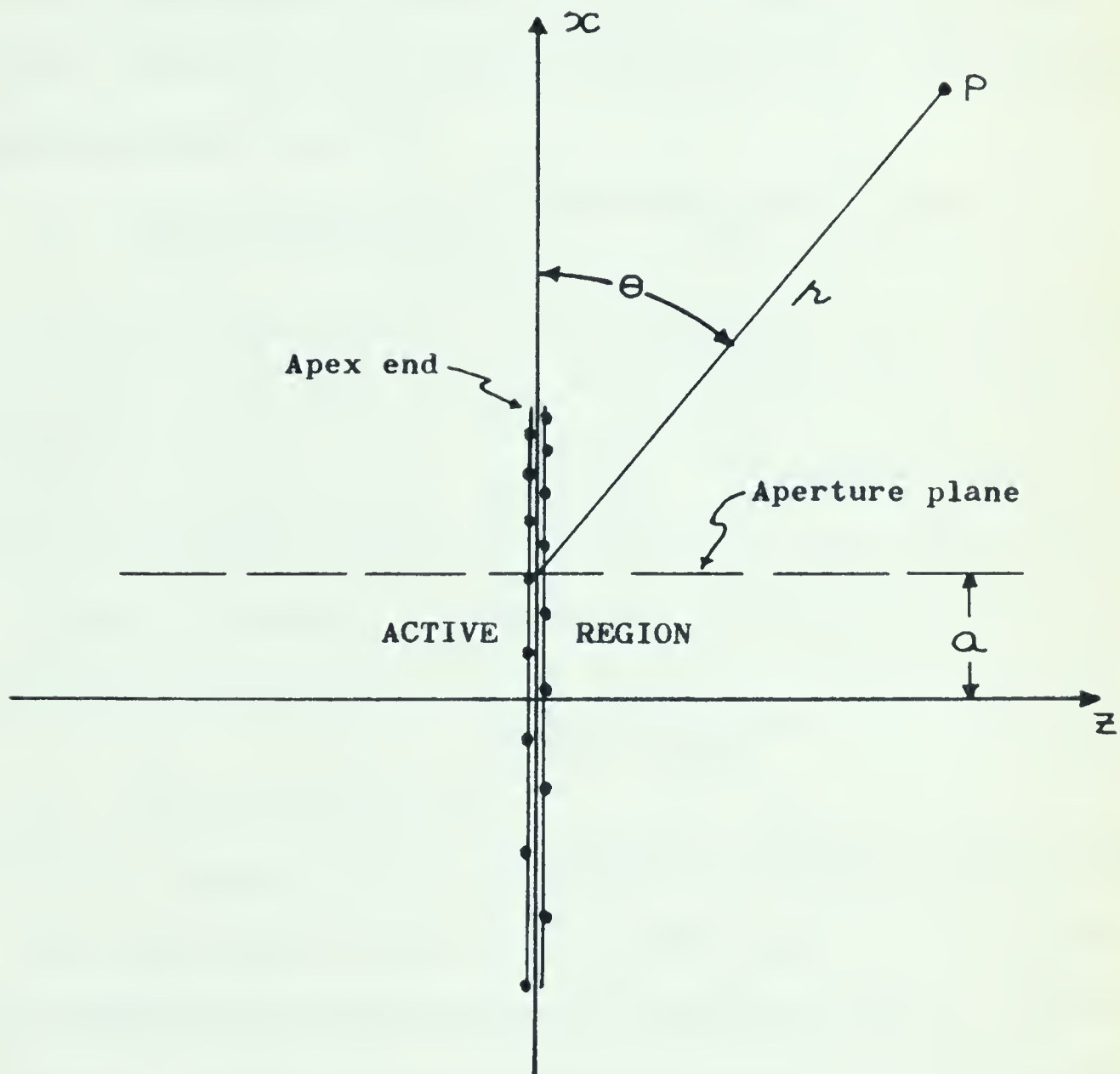


FIGURE 10  
COORDINATE SYSTEM USED TO  
CALCULATE THE RADIATION PATTERNS  
FOR THE LOG PERIODIC DIPOLE ANTENNA





assuming  $k_y = 0$ .  $\gamma_R (= \gamma_z)$  is the complex propagation along the centreline of the antenna and thus describes the properties of the radiation wave in the direction along the boom. Equation (25) is readily obtained from Figures 8 and 10. The electric field intensity, parallel to the antenna elements in the active region, is obtained from (19) and (25)

$$E_y(x, z) = E_o e^{-z(\alpha_z + j\beta_z)} e^{x(\alpha_x - j\beta_x)} \quad (26)$$

where the time dependence,  $e^{j\omega t}$ , is suppressed.  $E_o$ , the amplitude of the fields at  $x = z = 0$  is not needed since only the shape of the radiation fields will be considered. If required, it could be computed by equating the total input power to the radiated power.

Since the antenna is symmetrical about the  $x$  axis, the component of the electric field is given by

$$E_y(x, z) = E_o e^{-|z|\gamma_z - x\gamma_x} \quad (27)$$

in the active region, bounded by  $0 \leq x \leq a$ ,  $-\infty \leq y \leq \infty$ , and  $-\infty \leq z \leq \infty$ .

The hypothesis is made that the aperture is in the plane  $x = a$ ,  $-\infty \leq y \leq \infty$  and  $-\infty \leq z \leq \infty$  which is perpendicular to the axis of the antenna. This parallels the approach used for the long Yagi-Uda antenna previously described. Therefore, the aperture distribution is given by

$$E_y(a, z) = E_o e^{-|z|\gamma_z} e^{-a\gamma_x} = E_a e^{-|z|\gamma_z} \quad (28)$$



For  $x > a$  the electric field  $E_y(x, z)$  may be found using the bilateral Laplace transform<sup>23</sup> (equivalent to a Fourier transform<sup>25</sup>)

$$E_y(x, z) = \int_c E_y(\gamma'_z) e^{-\gamma'_x x - \gamma'_z z} d\gamma'_z \quad (29)$$

and

$$E_y(\gamma'_z) = \frac{1}{2\pi j} \int_{-\infty}^{\infty} E_y(a, z) e^{\gamma'_x a + \gamma'_z z} dz \quad (30)$$

where  $\gamma'^2_x = -k_o^2 - \gamma'^2_z$  and  $E_y(a, z)$  is given by (28).

In the following analysis, the parameters describing the radiation wave in the active region,  $\gamma_z$  and  $\gamma_x$  (or  $\gamma_R$ ) will be treated as constants. This is not necessarily true for the actual antenna since there is some variation of the far field radiation patterns as the frequency is swept over a period of the antenna. It was observed during the measurements presented in Appendix B that the position of the standing wave on the transmission line field shifted through  $360^\circ$  as the frequency was swept over a period. Therefore, it is reasonable to assume that the position of the variations in the radiation wave making up the aperture distribution will also vary periodically with frequency and will cause the periodic variations in the far field patterns. However, the variations of the far field patterns, and thus, the effect of the variations in the radiation wave, are small over the useable range of the antenna's parameters so that the patterns obtained by assuming  $\gamma_z$  and  $\gamma_x$  (or  $\gamma_R$ ) are constants will represent the average shape of the radiation patterns. Substituting (28) into (30) gives<sup>23</sup>







$$\begin{aligned}
E_y(\gamma'_z) &= \frac{1}{2\pi j} \int_{-\infty}^{\infty} E_a e^{-|z|\gamma_z} e^{\gamma'_x a + \gamma'_z z} dz \\
&= \frac{E_a e^{\gamma'_x a}}{2\pi j} \left[ \int_{-\infty}^0 e^{(\gamma_z + \gamma'_z)z} dz + \int_0^{\infty} e^{(-\gamma_z + \gamma'_z)z} dz \right] \\
&= \frac{E_a e^{\gamma'_x a}}{2\pi j} \left[ \frac{1}{\gamma_z + \gamma'_z} - \frac{1}{-\gamma_z + \gamma'_z} \right]
\end{aligned} \tag{31}$$

$$E_y(\gamma'_z) = \frac{E_a \gamma_z e^{\gamma'_x a}}{j\pi [\gamma_z^2 - \gamma'^2_z]} \tag{32}$$

In (32),  $\gamma'_x = j(k_o^2 + \gamma'^2_z)^{1/2}$ , but  $\gamma_z = \alpha_z + j\beta_z$  is a constant.

Therefore, substituting (32) into (29) gives

$$E_y(x, z) = \frac{E_a \gamma_z}{j\pi} \int_c \frac{e^{-\gamma'_x(x-a) - \gamma'_z z}}{\gamma_z^2 - \gamma'^2_z} d\gamma'_z \tag{33}$$

This integral may be solved using the saddle-point method of approximation outlined in Appendix A.<sup>23,26</sup> In order to simplify the mechanics of using this method, the complex  $\gamma'_z$  plane is transformed into the complex  $\phi'$  plane using the transformation

$$\gamma'_z = jk_o \sin \phi' \tag{34}$$

where

$$\gamma'_z = \alpha'_z + j\beta'_z \quad \phi' = \sigma' + j\tau' \tag{35}$$

Also, the rectangular coordinates  $x, z$  are changed to the cylindrical coordinates  $r, \theta$  where  $\theta$  is measured from the  $x$  axis.

Therefore, (33) becomes

$$E_y(r, \theta) = \frac{E_a \gamma_z k_o}{\pi} \int_{c_0} \frac{e^{-jk_o r \cos(\phi' - \theta)} \cos \phi'}{k_o^2 \sin^2 \phi' + \gamma_z^2} d\phi' \tag{36}$$

(10)

$$x^2 + y^2 = r^2 \Rightarrow x^2 = r^2 - y^2 \Rightarrow \frac{d}{dt} x^2 = \frac{d}{dt} (r^2 - y^2) = 2r \frac{dr}{dt} - 2y \frac{dy}{dt}$$

$$\left( \frac{d}{dt} x^2 \right) = 2r \frac{dr}{dt} - 2y \frac{dy}{dt} = 0$$

$$\left( \frac{d}{dt} x^2 \right) = 2r \frac{dr}{dt} - 2y \frac{dy}{dt} = 0$$

(11)

$$\frac{d}{dt} \left( \frac{x^2}{2} + \frac{y^2}{2} \right) = 0$$

Let  $x = r \cos \theta$  and  $y = r \sin \theta$ . Then  $\frac{d}{dt} \left( \frac{x^2}{2} + \frac{y^2}{2} \right) = \frac{d}{dt} \left( \frac{r^2}{2} \right) = r \frac{dr}{dt} = 0$

(12)

$$\frac{d}{dt} \left( \frac{x^2}{2} + \frac{y^2}{2} \right) = 0 \Rightarrow \frac{d}{dt} \left( \frac{r^2}{2} \right) = 0 \Rightarrow r \frac{dr}{dt} = 0$$

The above result shows that the radius  $r$  is constant. This is the case for a circular orbit. The angular momentum  $L$  is also constant. The total energy  $E$  is also constant. The motion is periodic with period  $T$ .

(13)

$$\frac{d}{dt} \left( \frac{x^2}{2} + \frac{y^2}{2} \right) = 0$$

(14)

$$\frac{d}{dt} \left( \frac{x^2}{2} + \frac{y^2}{2} \right) = 0$$

Also, the angular momentum  $L$  is constant. The total energy  $E$  is also constant. The motion is periodic with period  $T$ .

(15)

$$\frac{d}{dt} \left( \frac{x^2}{2} + \frac{y^2}{2} \right) = 0$$

The contour  $C_0$  in the complex  $\phi'$  plane may be deformed into a steepest descent contour, SDC, passing through the saddle-point at  $\phi' = \theta$ . The integral may now be evaluated asymptotically to yield the far field radiation pattern, given by

$$E_y(r, \theta) = E_a \gamma_z k_0 \left( \frac{2}{\pi k_0 r} \right)^{1/2} e^{-j(k_0 r - \pi/4)} f_1(\theta) \quad (37)$$

where the pattern function,  $f_1(\theta)$ , is

$$f_1(\theta) = \frac{\cos \theta}{k_0^2 \sin^2 \theta + \gamma_z^2} \quad (38)$$

The details for the above transformation and integration are given in Appendix A.

A slightly more convenient form of (38) can be obtained in order to simplify the calculation of the radiation fields. The relation

$$\gamma_x^2 + \gamma_z^2 + k_0^2 = 0 \quad (39)$$

can be obtained from (16) and (17) on Page 23. Substituting (39) in (38) yields

$$f_1(\theta) = \frac{-\cos \theta}{k_0^2 \cos^2 \theta + \gamma_x^2} = \frac{-\cos \theta}{k_0^2 \cos^2 \theta + \gamma_R^2} \quad (40)$$

since  $\cos^2 \theta + \sin^2 \theta = 1$ . The angle  $\theta$  is measured from the  $x$  axis as shown in Figure 10.

In the above derivation, the radiation patterns of the individual elements making up the antenna were not considered. Hence, (40) represents the group pattern for the log periodic dipole antenna. Since the radiation pattern of a dipole is isotropic in the H plane, (40) also represents the H plane radiation pattern for







the log periodic dipole antenna. To obtain the E field pattern, the unit patterns of the individual elements must be considered. An exact solution for these unit patterns would be very difficult to obtain because the lengths of the elements vary and mutual coupling must be accounted for. However, measurements made on a wire triangular tooth log periodic antenna indicated that the current distribution over each tooth is approximately sinusoidal so that mutual coupling between the teeth has little effect on the shape of the current distribution on the individual teeth.<sup>5</sup> Since the active region is approximately centred on the elements which are near a half wavelength long and the radiation patterns for dipoles near this length are practically identical to the pattern of a half wave dipole, the E field pattern for this antenna can be approximated by multiplying the group pattern by the unit pattern for a half wave dipole. The pattern function of a half wave dipole with a sinusoidal distribution is given by<sup>27</sup>

$$f_d(\theta) = \frac{\cos(\pi/2 \sin \theta)}{\cos \theta} \quad (41)$$

Therefore, the pattern function for the E field of the log periodic dipole antenna is given by

$$f_2(\theta) = \frac{-\cos(\pi/2 \sin \theta)}{k_o^2 \cos^2 \theta + \gamma_R^2} \quad (42)$$

### 3.4 Generalization to Other Types of Log Periodic Antennas ( $\psi \neq 0$ )

Although it is necessary to feed two half structures, or booms, of the log periodic antennas against each other or against ground in order to obtain wideband operation, R. H. DuHamel and



D. G. Berry have shown, using special measuring techniques, that each half structure produces a unidirectional beam pointing in the direction of its centreline.<sup>18</sup> In order to explain the variation of the H field radiation pattern for the complete antenna as the angle,  $\psi$ , between the booms is changed, a reasonable approach would be to assume that each boom radiates independently of the other. Therefore, the overall pattern of the complete antenna can be calculated if the direction of each half structure and the position of the centre of radiation on each boom is known. Although this approach neglects the effects of mutual coupling between the booms, it has still been found to give good results.<sup>10,18</sup>

The H field radiation pattern for a log periodic antenna may be calculated by considering Figure 11.<sup>18</sup> In space, the total radiation pattern is given by the vector addition of the radiation fields from the booms. If the phase reference point is taken to be the apex of the antenna, then the pattern function for the H field is given to be

$$f_3(\theta) = f_1(\theta - \psi/2) e^{-j2\pi l_1} + f_1(\theta + \psi/2) e^{-j2\pi l_2} \quad (43)$$

The lengths,  $l_1$ , and  $l_2$ , are given by

$$\begin{aligned} l_1 &= d \cos(\theta - \psi/2) \\ l_2 &= d \cos(\theta + \psi/2) \end{aligned} \quad (44)$$

where  $d$  is the distance from the apex of the antenna to the centre of radiation on each boom, measured in wavelengths. For the special case where  $\psi = 0$ , the patterns for the overall antenna should be identical to those of each individual boom, providing the effects of











mutual coupling are ignored. It has been observed that the radiation patterns of the log periodic dipole antenna ( $\gamma = 0$ ) and the wire trapezoidal tooth half structure with equivalent parameters do agree closely.<sup>10</sup> Therefore, the pattern function,  $f_1(\theta \pm \frac{\gamma}{2})$ , in (43) for each boom is given by (40) of the previous subsection making the substitution  $\theta \pm \gamma/2$  for  $\theta$ .

In the previous subsection, the hypothesis was made that the "effective aperture" for these antennas occurs at the front of the active region, which is roughly centred on the elements one half wavelength long. However, in order to apply (43), the exact location of the phase centre of the radiation from each boom must be known. This information is available in the literature and is reproduced in Figure 12 for the wire trapezoidal tooth structure.<sup>10</sup> It was found that  $d$  (measured in wavelengths) is essentially independent of frequency. For apex angles less than  $60^\circ$  and in the range of geometric ratios for which the patterns do not break up excessively over a period,  $d$  is also independent of the geometric ratio. Thus, Figure 12 can be applied to any antenna whose parameters meet the above restrictions.

It has been observed that the E field beamwidth is reasonably constant as the angle,  $\gamma$ , between the booms is varied from  $0^\circ$  to  $90^\circ$  and so (42) can be used to calculate the E field pattern for a log periodic antenna if  $\gamma$  is not too large.<sup>10</sup>

### 3.5 Relationships Between the Radiation and Transmission Line Waves

It has been shown that the transmission line wave terminates in the active region, and the radiation wave originates therein.





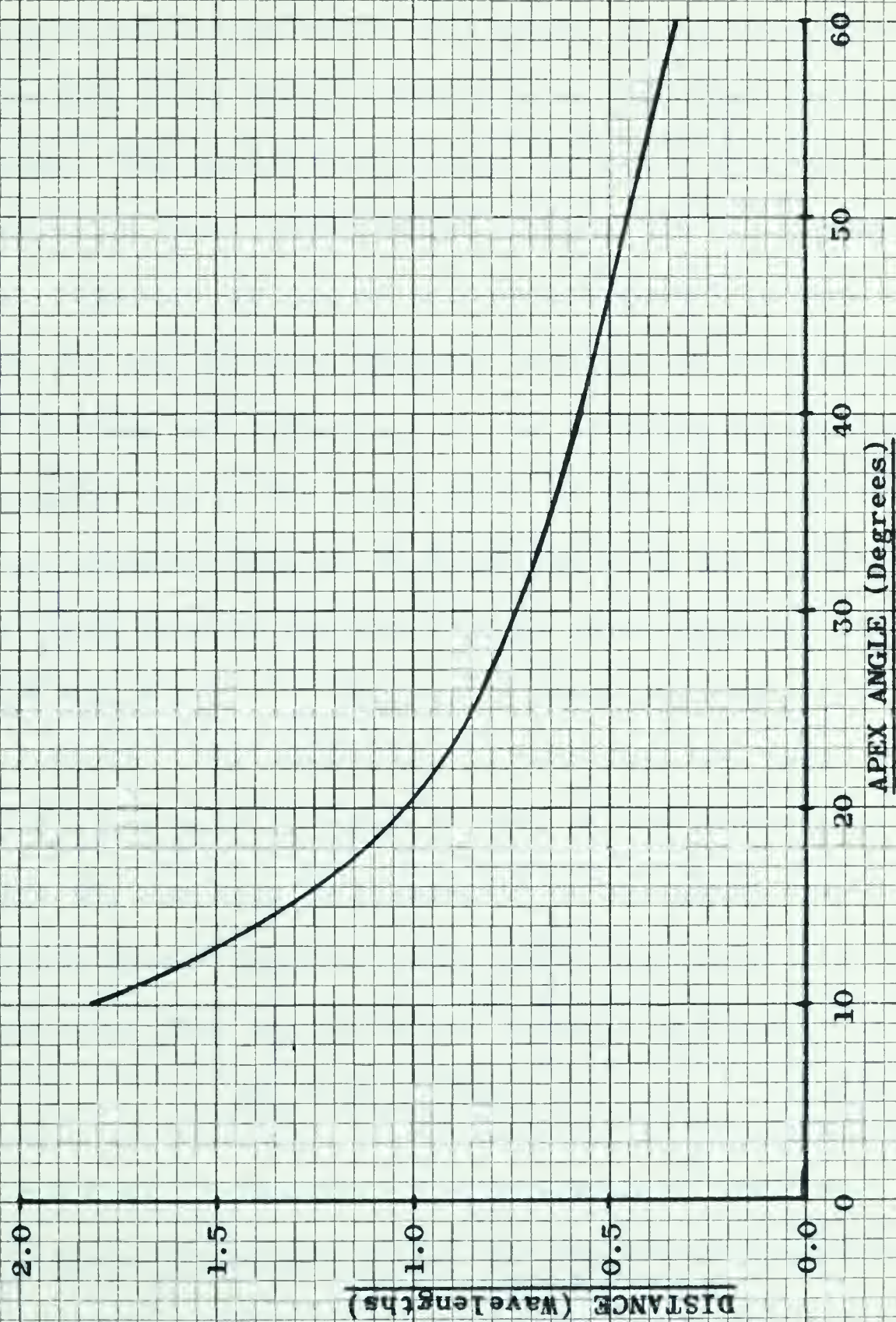


FIGURE 12

DISTANCE ( $d$ ) FROM APEX TO PHASE CENTRE  
FOR A WIRE TRAPEZOIDAL TOOTH LOG PERIODIC HALF STRUCTURE





Therefore, the current distribution on the teeth in this region serves as the common boundary between the two waves.<sup>5</sup>

To simplify the problem of finding the relations between the radiation and transmission line waves, it is convenient to consider an end fire array whose elements have identical input impedances and spacings. For normal excitation, the phase progression along the transmission line field and the currents in the elements would be equal. However, as can be seen from Figure 7 on Page 20, the feed system for log periodic antennas produces an additional  $180^\circ$  phase shift between the elements. If the elements are spaced  $\delta$  wavelengths apart, the phase shift of the transmission line wave between elements will be  $\beta_T \delta$  radians with the result that the phase shift between the currents on these elements will be given by  $\beta_R \delta = \beta_T \delta - \pi$ .<sup>4</sup>  $\beta_R$  is the phase constant of the currents on the elements. The assumption is made that the phase progression of the radiation wave is equal to that of the currents on the elements so that  $\beta_R$  can also be considered as the phase constant of the radiation wave. Therefore, the phase relationship between the transmission line and radiation waves is given by

$$\beta_R = \beta_T - \pi/\delta \quad (45)$$

for an array of elements having equal input impedances and spacings.

Since the active region for log periodic antennas is short, the variation in the spacings between the elements will be small and an average value can be used.<sup>4</sup> If  $\delta$  is taken to be the





spacing between the half wavelength element and the next closer element to the apex of the antenna, the result<sup>3</sup>

$$\sigma = \frac{1 - \tau}{4 \tan \frac{\theta_1}{2}} \quad (46)$$

is obtained from the geometry of the log periodic dipole antenna shown in Figure 7 on Page 20. Using this approximation, it has been found that (46) gives a good representation for the phase shift of the currents on the elements of a log periodic dipole antenna even though the input impedances of the elements do vary over the active region.<sup>3</sup>

For successful log periodic antennas  $\sigma$  is normally small so that  $\beta_R$  is negative with respect to  $\beta_T$ , and the radiation wave propagates back toward the apex. This explains why log periodic antennas behave as backward wave structures.<sup>4</sup>

In the active region, it can be seen from measurements presented in the literature that the currents on the elements build up very quickly near the front of the active region and then decay at nearly the same rate as the transmission line wave toward the rear of the region.<sup>3</sup> Since these currents act as the source of the radiation wave the phase progression and distribution of the wave will be similar to that of the currents in the elements. Since the phase progression of the elements is toward the apex, the wave will appear to start at the rear of the active region and propagate toward the apex, building up at the same rate as the currents on the elements. Near the front of the active region, the currents on the

regarding between the cell constants measured on the two sides of the specimen is the same as the ratio of the cell constants.

$$\frac{1}{\epsilon} = \frac{1}{\epsilon_0} + \frac{1}{\epsilon_1}$$

is obtained from the summation of the cell constants. It is known in Figure 1 that  $\epsilon_0$  is the cell constant, it is

known from (1) that  $\epsilon_1$  is the cell constant for the same

ratio of the constants on the sides of the specimen.

Therefore, the ratio of the cell constants on the two sides of the

specimen is the same as the ratio of the cell constants.

For convenience in calculation, it is assumed that

the ratio  $\epsilon_1/\epsilon_0$  is equal to the ratio of the cell constants.

When the specimen is placed in the cell, the cell constants

measured on the two sides of the specimen are equal.

In the case of the specimen, it can be seen that

measured in the specimen, the ratio of the cell constants is

the ratio of the cell constants of the specimen and the ratio

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elements drop off rapidly, and have little effect on the radiation wave. Once established, the wave continues to propagate toward and beyond the apex, and becomes the far field at a suitable distance in front of the antenna. Therefore, the decay of and build up of the transmission line and radiation waves, respectively, are approximately equal and the relationships between the two waves are given by

$$\alpha_R = \alpha_T \quad (47)$$

and (45) to a first order approximation.  $\alpha_R$  and  $\alpha_T$  are the attenuation constants for the radiation and transmission line waves, respectively, in the active region.

### 3.6 Summary

It was shown in this section that the radiation mechanism of the log periodic class of antennas can be analysed in terms of leaky waves.

Three basic assumptions were made which greatly simplified the analysis. The first was that each half structure radiates independently of the other. Although this approach ignores the effects of mutual coupling between the booms, it has been found to give a good approximation to the measured patterns.<sup>18</sup> Therefore, only one boom need be considered in the leaky wave analysis. The overall pattern can then be obtained by adding the radiation from the two booms vectorially in space. The second simplification made was to assume  $k_y = 0$ , therefore, reducing the problem to two dimensions.



elementary step, etc. etc., and then the system is in a steady state. These conditions are then substituted in the equations of motion and the system is solved for the steady state. The results are then compared with the results of the other method. The results are then compared with the results of the other method. The results are then compared with the results of the other method.

(11)

$$\frac{d^2x}{dt^2} = -\frac{g}{L}x$$

and (12) is a first order approximation. The results are then compared with the results of the other method. The results are then compared with the results of the other method. The results are then compared with the results of the other method.

### 2.4. Summary

It was shown in (11) and (12) that the results of the other method are in good agreement with the results of the other method. The results are then compared with the results of the other method. The results are then compared with the results of the other method. The results are then compared with the results of the other method.



This allowed the fields to be described in terms of exponential functions. The third premise was that the radiating aperture for each half structure occurs at the front of the active region and is in the plane perpendicular to the boom. The radiation wave, which passes through this plane, was assumed to make up the aperture field distribution.

The far field pattern for each boom may be obtained from the aperture distribution using a bilateral Laplace transform. This transform is valid in the half plane  $x > a$ , and therefore, the solutions obtained are valid over the range  $-\frac{\pi}{2} < \theta < \frac{\pi}{2}$  and will not predict the radiation in the rear quadrants.

The E and H field pattern functions for each boom or for the log periodic dipole antenna ( $\gamma = 0$ ) were found to be

$$f_2(\theta) = \frac{-\cos(\frac{\pi}{2} \sin \theta)}{k_o^2 \cos^2 \theta + \gamma_R^2} \quad (42)$$

and

$$f_1(\theta) = \frac{-\cos \theta}{k_o^2 \cos^2 \theta + \gamma_R^2} \quad (40)$$

respectively. The azimuth angle,  $\theta$ , is measured from the extended centreline of the boom.  $\gamma_R$  is the complex propagation constant of the radiation wave along the boom and  $k_o$  is the wave number of free space.

The H field pattern function for other types of log periodic antenna ( $\gamma \neq 0$ ) was found to be

$$f_3(\theta) = f_1(\theta - \tau/2) e^{-j2\pi \ell_1} + f_1(\theta + \tau/2) e^{-j2\pi \ell_2} \quad (43)$$



where

$$\begin{aligned} l_1 &= d \cos(\theta - \gamma/2) \\ l_2 &= d \cos(\theta + \gamma/2) \end{aligned} \quad (44)$$

The distance,  $d$ , from the apex of the antenna to the centre of radiation on each boom, measured in wavelengths, may be obtained from Figure 22. The azimuth angle,  $\theta$ , is measured from the extended centreline of the antenna as shown in Figure 11.

The relations between the radiation field and the transmission line field were found to be

$$\alpha_R = \alpha_T \quad (47)$$

and

$$\beta_R = \beta_T - \frac{\pi}{\sigma} \quad (45)$$

where

$$\sigma = \frac{1 - \gamma}{4 \tan \frac{\theta}{2}} \quad (46)$$

for the log periodic dipole antenna, or

$$\sigma = \frac{1 - \sqrt{\gamma}}{4 \tan \frac{\theta}{2}} \quad (48)$$

for the other types of log periodic antenna. (48) is obtained from (46) using (15) on Page 19.

If the angle,  $\gamma$ , between the booms is less than  $90^\circ$ , the pattern function for the E field can be approximated by (42) with reasonable accuracy.<sup>10</sup>

Using the above equations, the shape of the far field patterns can be obtained for the log periodic class of antennas if the properties of the transmission line field are known. These can







be obtained either theoretically or experimentally. In this thesis they will be found experimentally using the measuring procedure outlined in the next section.



Let  $f(x)$  be a function defined on the interval  $[a, b]$ . The function  $f(x)$  is said to be continuous at the point  $x_0$  if for every  $\epsilon > 0$  there exists a  $\delta > 0$  such that for all  $x$  satisfying  $|x - x_0| < \delta$  we have  $|f(x) - f(x_0)| < \epsilon$ .

The function  $f(x)$  is said to be continuous on the interval  $[a, b]$  if it is continuous at every point  $x_0$  of the interval. The function  $f(x)$  is said to be discontinuous at the point  $x_0$  if it is not continuous at  $x_0$ . The function  $f(x)$  is said to be discontinuous on the interval  $[a, b]$  if it is discontinuous at some point  $x_0$  of the interval.

$$\lim_{x \rightarrow x_0} f(x) = L$$

#### SECTION 4

Let  $f(x)$  be a function defined on the interval  $[a, b]$ . The function  $f(x)$  is said to be continuous at the point  $x_0$  if for every  $\epsilon > 0$  there exists a  $\delta > 0$  such that for all  $x$  satisfying  $|x - x_0| < \delta$  we have  $|f(x) - f(x_0)| < \epsilon$ . The function  $f(x)$  is said to be continuous on the interval  $[a, b]$  if it is continuous at every point  $x_0$  of the interval. The function  $f(x)$  is said to be discontinuous at the point  $x_0$  if it is not continuous at  $x_0$ . The function  $f(x)$  is said to be discontinuous on the interval  $[a, b]$  if it is discontinuous at some point  $x_0$  of the interval.

$$f(x) = \begin{cases} x^2 & \text{if } x \neq 0 \\ 1 & \text{if } x = 0 \end{cases}$$

Let  $f(x)$  be a function defined on the interval  $[a, b]$ . The function  $f(x)$  is said to be continuous at the point  $x_0$  if for every  $\epsilon > 0$  there exists a  $\delta > 0$  such that for all  $x$  satisfying  $|x - x_0| < \delta$  we have  $|f(x) - f(x_0)| < \epsilon$ . The function  $f(x)$  is said to be continuous on the interval  $[a, b]$  if it is continuous at every point  $x_0$  of the interval. The function  $f(x)$  is said to be discontinuous at the point  $x_0$  if it is not continuous at  $x_0$ . The function  $f(x)$  is said to be discontinuous on the interval  $[a, b]$  if it is discontinuous at some point  $x_0$  of the interval.





#### 4. MEASURING TECHNIQUES

In the previous section, the far field radiation patterns for the log periodic antenna were derived in terms of the attenuation and phase constants of the transmission line field in the active region of the antenna.

The attenuation constant may be obtained by plotting the relative strength of the transmission line field in decibels versus the distance from the apex of the antenna in wavelengths and taking the slope of the curve in the active region. If this is done, the attenuation constant is given by

$$\alpha_T = - \frac{\text{Slope}}{8.68} \quad (49)$$

where  $\alpha_T$  is expressed in nepers per wavelength. Similarly, the phase constant is found by plotting the relative phase shift of the transmission line field in degrees versus the distance from the apex of the antenna in wavelengths and taking the slope of the resulting curve in the active region. The phase constant is then given by

$$\beta_T = - \frac{\text{Slope}}{57.3} \quad (50)$$

where  $\beta_T$  is expressed in radians per wavelength. Therefore, the relative magnitude and phase of the transmission line field must be measured along the antenna before the radiation patterns can be calculated.



#### 4.1 Requirements of the Measuring System

Since both the transmission line and radiation fields exist along the antenna, the measurement of each individual field is rather difficult. It has been found that both these fields contain components parallel and perpendicular to the elements except in the plane equidistant between the two booms.<sup>5</sup> In this plane, the electric part of the transmission line field, perpendicular to both this plane and the elements, is parallel to the plane containing the booms. The radiation field is parallel to the elements in this plane. Therefore, the sampling probe must have the proper polarization to a high degree of accuracy in order to sample the desired field.

The system used to make the measurements must give accurate results over a wide range of signal levels since the transmission line field decays about 20 db. in the active region.

A desirable, although not essential, property of the measuring system is that it can be used over a reasonably wide frequency range. If the measurements are made on a log periodic dipole antenna having a particular value of geometric ratio,  $\tau$ , and apex angle,  $\theta$ , the same structure can be used for smaller apex angles simply by cutting the elements to the required lengths. Since it is necessary to measure the properties of the transmission line field over a large range of antenna parameters, a wide band measuring system will greatly reduce the time required for assembling the structures.







The final requirement is that the frequency at which the measurements are to be made lies within the ranges 400 mc. to 1,000 mc. and 1,900 mc. to 4,000 mc. in order to use the equipment available. The range between 1,900 mc. and 3,500 mc. is the most attractive since the antenna structures will be of a convenient size to handle, but will not become too small for the required physical tolerances to be maintained.

Direct probing of the fields by a dipole is unsatisfactory since a significant portion of the signal is picked up by the outer conductor of the coaxial lead. This results in the measurement of a very high standing wave caused by the vector addition of the transmission line field and the radiation field picked up by the outer conductor. It is possible to eliminate the induced currents produced by these fields on the outer conductor by using chokes.<sup>19</sup> However, the space between the booms is rather limited so that this method is not too practical unless a relatively low frequency is used.<sup>5</sup>

Since only one boom is used for the feed system, the other boom may be slotted and the cable to the probe run inside the boom.<sup>3</sup> When the angle between the booms is zero, such as for the log periodic dipole antenna, the probe is very nearly symmetrically located between the booms and good results are obtained. This requires that the inner diameter of the boom be large enough to allow the feed cable to move freely and effectively limits the upper frequency at which the measurements can be made to about 500 mc.

The total treatment is about the equivalent of 1000 cc.

approximately 100 cc. of water from which the water was removed.

1,000 cc. and 1,500 cc. in order to see the difference

available. The total between 1,000 cc. and 1,500 cc. is the same

as before since the same amount will be at a constant level

in water, but will not become too small for the nitrogen

to be removed.

There is a problem of the water to be used in the

since a significant portion of the water is used in the

concentration of the water used. This results in the

a very high water level which is the result of the

transformation from liquid to the solid state and the

water content. It is possible to estimate the

produced by those living on the water content of water

however, the water content of the water is not

which is not too small to be a serious

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Since only the water is used for the water

from which the water is used and the water is

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Therefore, this method is not entirely satisfactory.

#### 4.2. Description of Measuring System

The method finally settled on was to mount a single boom over a horizontal ground plane and use a vertical probe extending through the ground plane to make the measurements. Since the ground plane is located symmetrically between the boom and its image, the probe will only sample the transmission line field. This type of probe can be used over a wide frequency range. No currents will be induced in the feed cable as it is located below the ground plane.

Placing a single half structure over an infinite ground plane is analogous to rotating one of the booms about its centreline through an angle of  $180^\circ$  since the image antenna is the mirror image of the actual boom. The phase of the energy radiated by the rotated antenna is effectively reversed with the result that a null occurs along the extended centreline of the boom (or the ground plane) for the far field radiation pattern. However, it has been observed that the E field radiation pattern is unchanged when this rotation is made so that the effects of mutual coupling between the booms are small.<sup>14</sup> Therefore, the results obtained using the ground plane should be identical to those obtained for the complete antenna providing the ground plane is infinite in extent. However, it was observed that the magnitude of the standing wave of the transmission line field was larger than that obtained for the complete antenna. The reason for this apparent discrepancy is that the reflection of energy out of the active region from the booms adds in phase when

Therefore, this system is not suitable for use.

#### 4.2. The effect of temperature

The effect of temperature on the rate of reaction was studied by measuring the rate of reaction between hydrogen peroxide and potassium iodide at different temperatures. The results are shown in Table 1. It can be seen from the table that the rate of reaction increases as the temperature increases. This is because the molecules have more energy at higher temperatures and so they are more likely to collide with sufficient energy to overcome the activation energy barrier.

Figure 1 shows a graph of the rate of reaction against temperature. The graph shows that the rate of reaction increases exponentially with temperature. This is because the rate of reaction is proportional to the number of molecules that have sufficient energy to overcome the activation energy barrier. As the temperature increases, the number of molecules with sufficient energy increases exponentially.

The effect of temperature on the rate of reaction was also studied by measuring the rate of reaction between hydrogen peroxide and potassium iodide at different concentrations. The results are shown in Table 2. It can be seen from the table that the rate of reaction increases as the concentration of hydrogen peroxide increases. This is because there are more molecules of hydrogen peroxide available to react with the potassium iodide.

The effect of temperature on the rate of reaction was also studied by measuring the rate of reaction between hydrogen peroxide and potassium iodide at different pressures. The results are shown in Table 3. It can be seen from the table that the rate of reaction increases as the pressure increases. This is because there are more molecules of hydrogen peroxide available to react with the potassium iodide.



they are mirror images. For the normal antenna, the reflections are  $180^\circ$  out of phase and tend to cancel so that the standing wave is reduced. A similar phenomenon has been observed for the variation of the input impedance<sup>14</sup> which is of the same magnitude as the standing wave on the transmission line field.<sup>5</sup> Thus, this method of measuring might be considered as giving a more accurate picture of the actual transmission line field for a single boom than the measurements made on the complete antenna.

The ground plane used was a sheet of aluminum eight feet long and four feet wide. Absorbing material was piled around the edges of the plane to make it appear infinite in extent to the antenna. A photograph of the ground plane and absorbing material is shown in Figure 13. One piece of absorbing material was removed allowing the antenna to be seen from the rear.

A close up photograph of a typical antenna half structure, with parameters  $\tau = 0.81$  and  $\theta = 20^\circ$ , mounted over the ground plane is shown in Figure 14. The half structure is soldered to the centre conductor of a type UG-290/U panel jack mounted on a small aluminum bracket. This is shown in the right hand side of Figure 14. The feed cable from the generator runs up through a hole in the ground plane and is connected to the panel jack, thus, exciting the antenna. Probing holes were drilled in the ground plane at intervals of one half centimeter directly under the boom. The probe can be seen located in the ground plane near the rear of the antenna.







FIGURE 13

GROUND PLANE USED FOR MEASURING

TRANSMISSION LINE CHARACTERISTICS







FIGURE 14

HALF STRUCTURE MOUNTED OVER GROUND PLANE  
SHOWING DETAILS OF FEED SYSTEM



Figure 15 shows a view of the bottom side of the ground plane with the probe assembly in place. The probe is held in place by four aluminum strips bolted to the ground plane and running parallel to the holes. For each pair of strips, the one next to the ground plane is slightly farther from the holes than the top piece so that a small lip is formed which holds the probe assembly in place. The probe can be quickly removed by twisting it clockwise until the plate of the assembly clears the lips and then lifting it out of the hole. To insert the probe in another hole, the procedure is reversed. The feed from the generator is located behind the vertical plate.

The probe assembly is shown in Figure 16. The ruler shows the relative size of the probe in inches. The assembly consists of a short length of RG-142/U rigid coaxial cable, a type UG-89/U connector, a small aluminum plate and a small piece of tin. The cable is soldered to the piece of tin which is in turn bolted to the aluminum plate, thus, holding the assembly together. The connector is placed on the other end of the rigid cable and allows the signal to be coupled to a conventional coaxial cable. To insure maximum rigidity and stability, the connectors were taped together.

The circuit used to make the measurements is shown in Figure 17. A photograph of the circuit is shown in Figure 18. The relative phase along the antenna is determined by adjusting the position of the probe on the slotted line and the variable power divider to achieve a null on the receiver.







**FIGURE 15**

**BOTTOM SIDE OF GROUND PLANE**

**WITH PROBE ASSEMBLY IN PLACE**



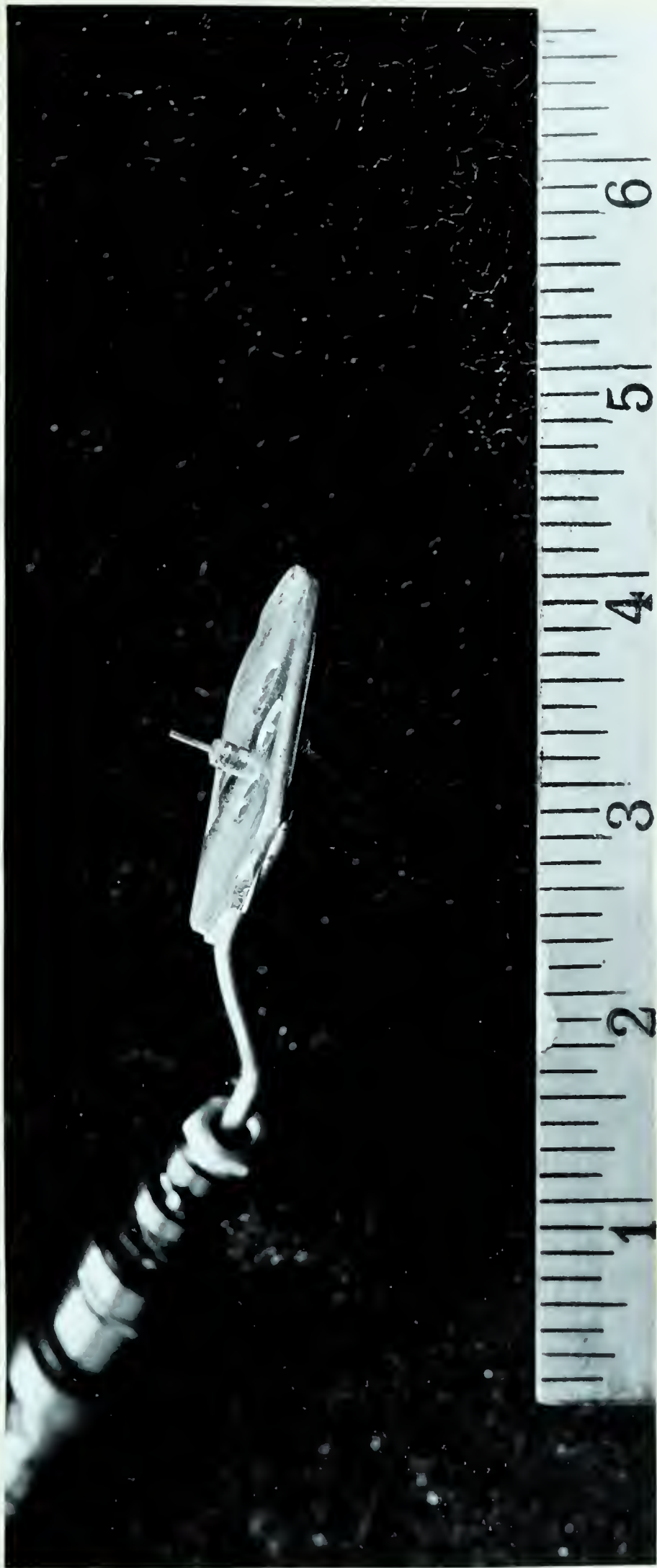


FIGURE 16  
PROBE ASSEMBLY





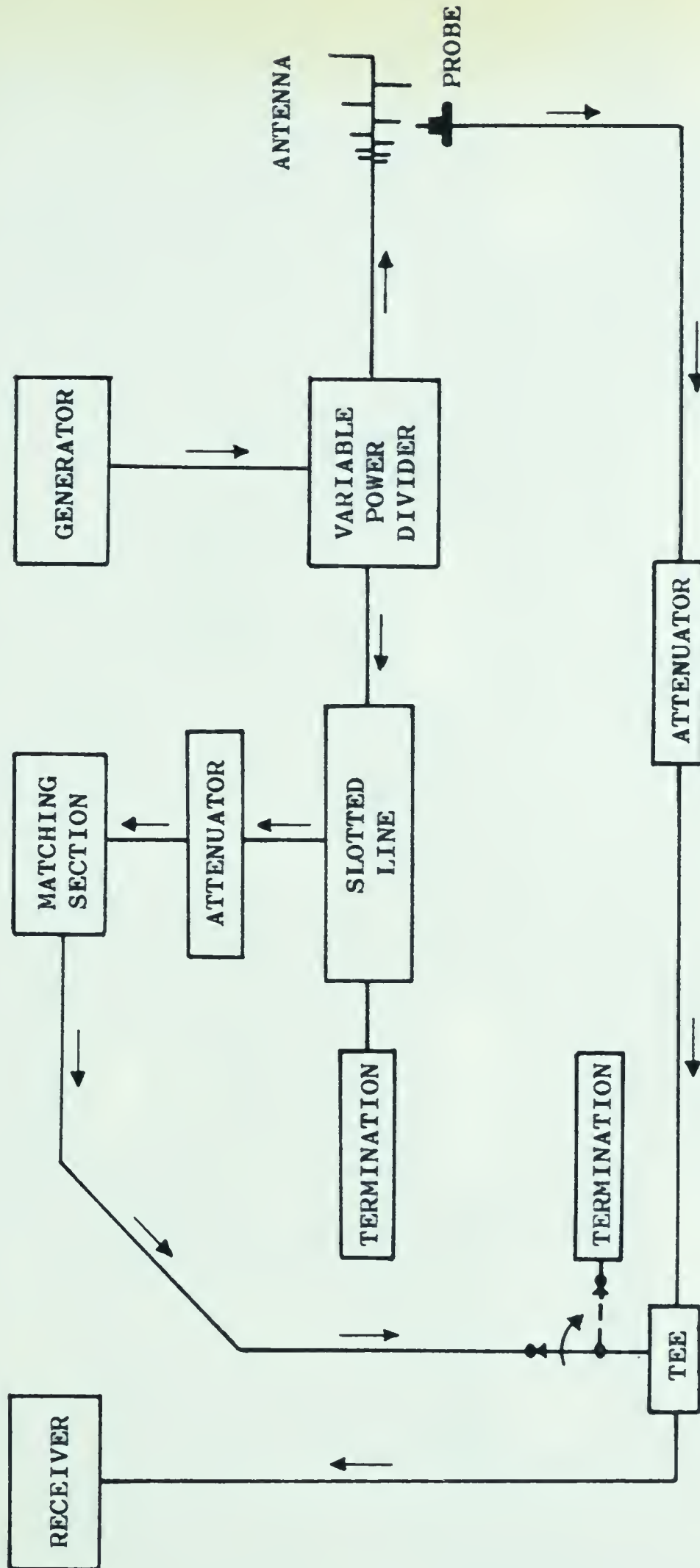
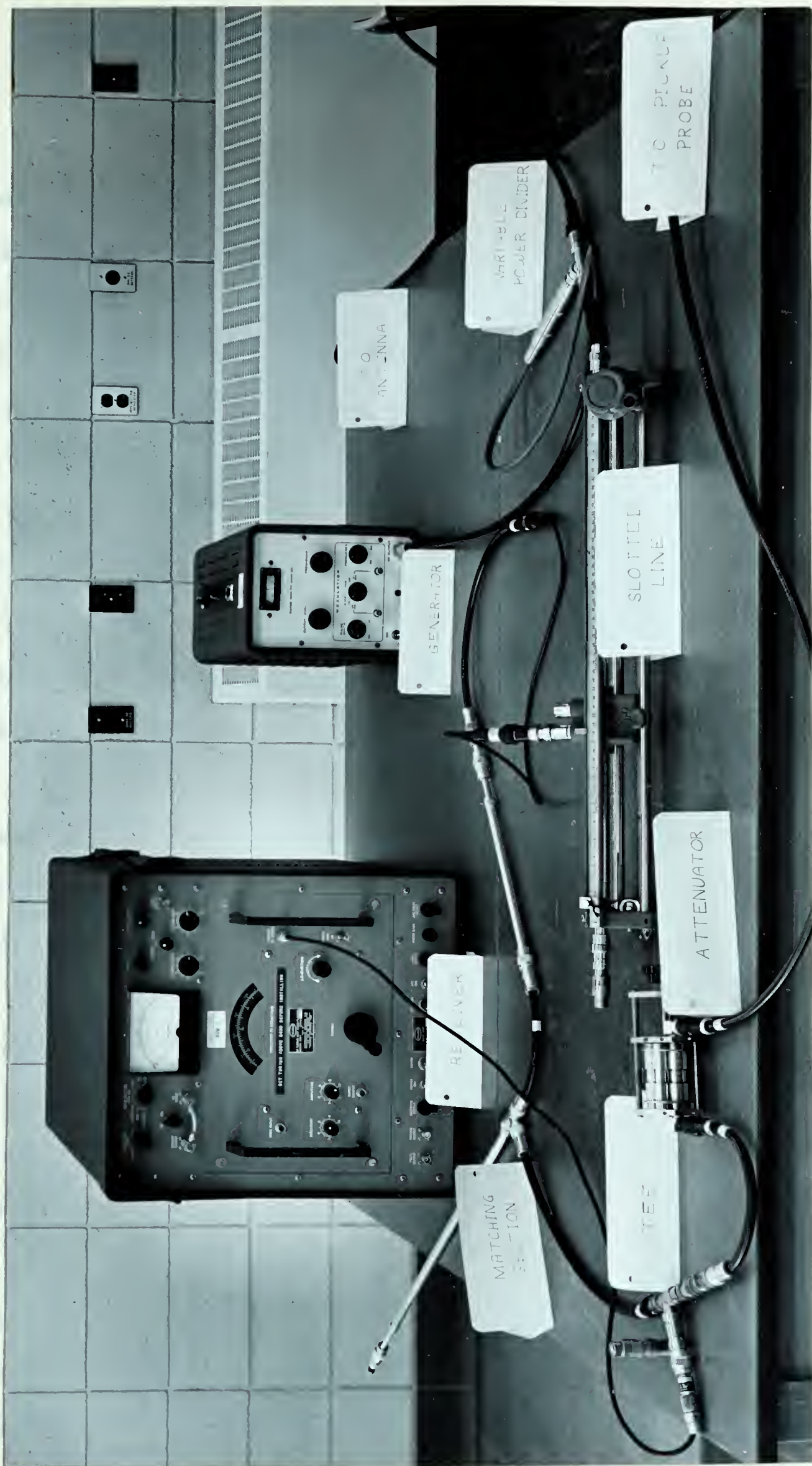


FIGURE 17

CIRCUIT USED FOR PHASE  
AND AMPLITUDE MEASUREMENTS





**FIGURE 18**  
**EQUIPMENT USED FOR THE MEASUREMENTS**





If the slotted line is terminated in its characteristic impedance then the signal will propagate down the line with the speed of light. Therefore, as the slotted line probe is moved along the line, the phase of the signal will vary  $2\pi\ell$  radians where  $\ell$  is the distance the slotted line probe moves, measured in wavelengths. If the signal levels from the pickup probe sampling the antenna fields and from the probe on the slotted line are equal in magnitude at the tee, a very sharp null will occur on the receiver when these two signals are  $180^\circ$  out of phase at the tee. As the sampling probe is moved along the antenna, the distance the probe on the slotted line must be moved to maintain the null condition will give the phase shift of the field on the antenna.

To measure the signal level picked up by the sampling probe, the cable from the slotted line probe was replaced by a standard termination at the tee connector.

#### 4.3 Evaluation of Measuring Circuit

The circuit shown in Figure 17 has several rather critical requirements if accurate results are to be obtained.

Since the variation of the signal level along the antenna exceeds 30 db. and a nulling technique was used to measure the phase, the detecting device must be capable of accurately measuring signal levels over a very wide range. The generator used, a PRD Electronics Corporation Type S712 signal source, has a maximum output of about 100 milliwatts and so the detector must possess high sensitivity. Fortunately, such a receiver, the Polaroid





Electronics Corporation Model R-B1 Microwave Receiver with Plug in Unit RW-T, was available. This receiver will measure signal levels over a 70 db. range. It was found that the maximum signal on the antenna usually gave a deflection of over 50 db. on the receiver so that a reserve exceeding 10 db. was available for making the phase measurements even at the lowest signal levels on the antenna. Very sharp nulls, as deep as 40 db., could be obtained while making the phase measurements in the high signal level region of the antenna. For lower signal levels, the depths of these nulls were determined primarily by the signal level.

Any reflections occurring in the circuit will be constant at a particular frequency and hence will appear as a standing wave for the phase measurements. For example, if the lines leading to the tee at the receiver have a VSWR of 1.1:1 the maximum possible error in the phase reading is  $\pm 5.5^\circ$ , providing both signals are equal in magnitude. It was possible to match one line to the above tolerance using a General Radio type 874-G10 attenuator pad and a Microlab AT-13N Turret Attenuator. The second line could not be matched to this tolerance using the remaining General Radio and Microlab attenuators so that a matching unit consisting of attenuators and a single stub match made up of two Microlab series SR adjustable lines was used. The required matching tolerance was then met at each frequency by adjusting the lengths of the adjustable lines.

Any variation from a linear phase shift, such as caused by





reflections, along the slotted line will also cause an apparent standing wave to appear on the phase measurements. A General Radio type 874-LBA slotted line and a type 874 WS termination was used for this part of the circuit. The standing wave measured at the input of the slotted line was found to be less than 1.1:1 for the frequencies at which the measurements were made. Therefore, the total phase error due to mismatch will not exceed  $\pm 8.5^\circ$  provided the signal levels in the two lines connected to the tee are equal in magnitude.

Since fairly high standing waves were found to exist on the transmission line field, the effects of the above mismatches were hardly noticeable during the course of the measurements.

The above tolerance on the phase measurements will rise rapidly if the magnitude of the signals in the lines connected to the tee are not kept equal. An equally important reason for maintaining equal signal levels in the lines is that the null broadens rapidly if the signals are not nearly equal. Therefore, the inclusion of the variable power divider in the circuit is a necessity since the signal level along the antenna varies over a very wide range. However, the phase delay through the power divider must remain constant as the attenuation is changed if the readings are to have any meaning. The output of the power divider used, a General Radio Type 874-GA adjustable attenuator, can be varied over a range of 120 db., which is more than adequate for the attenuation requirements. Basically it consists of a waveguide





structure operating in the cutoff region of the  $TE_{1,1}$  mode. The output signal is obtained from a pickup loop. Since the waveguide operates in the cutoff region, the signal picked up by the loop varies rapidly for small displacements of the loop, and of equal importance for this application, the phase shift through the device remains constant as the position of the loop is changed.

An attempt was made to calibrate the overall system. This was done by replacing the antenna with a length of brass welding rod short circuited by a vertical aluminum plate placed at the end of the rod in the plane perpendicular to the rod. Since there were no elements on the rod, the signal propagated down the rod with the speed of light, and was reflected by the plate causing a high standing wave to exist along the rod. The results are shown in Figures 19 and 20. The measured standing wave ratio along the line was about 9:1. The theoretical curve for this standing wave ratio was drawn to assist in evaluating the measurements. Since some perturbation of the fields were observed at the short circuiting plate, the zero position for the horizontal ordinate was displaced to the null one half wavelength from the plate. As can be seen, the results are satisfactory for both the magnitude and phase measurements.

During the early stages of the measurements, some difficulty was encountered due to direct radiation from the feed. The feed system shown in Figure 14, on Page 52, was evolved and found to give satisfactory results. Figure 21 shows the signal level of the





FIGURE 19

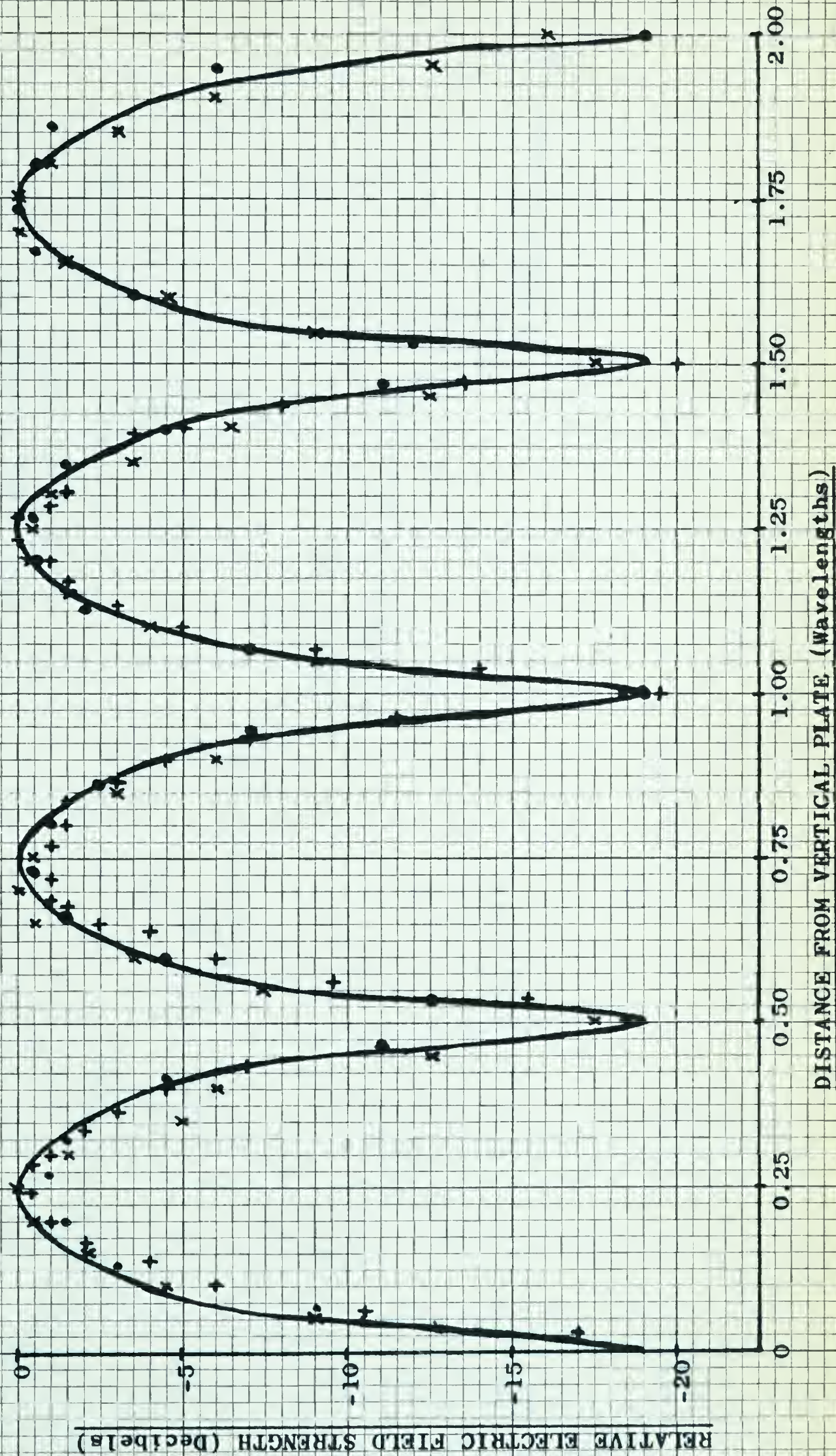
SYSTEM CALIBRATION FOR  
FIELD STRENGTH MEASUREMENTS

— Theoretical (VSWR=9:1)

+ + + + 2000 Mc.

x x x x 3000 Mc.

• • • • 4000 Mc.







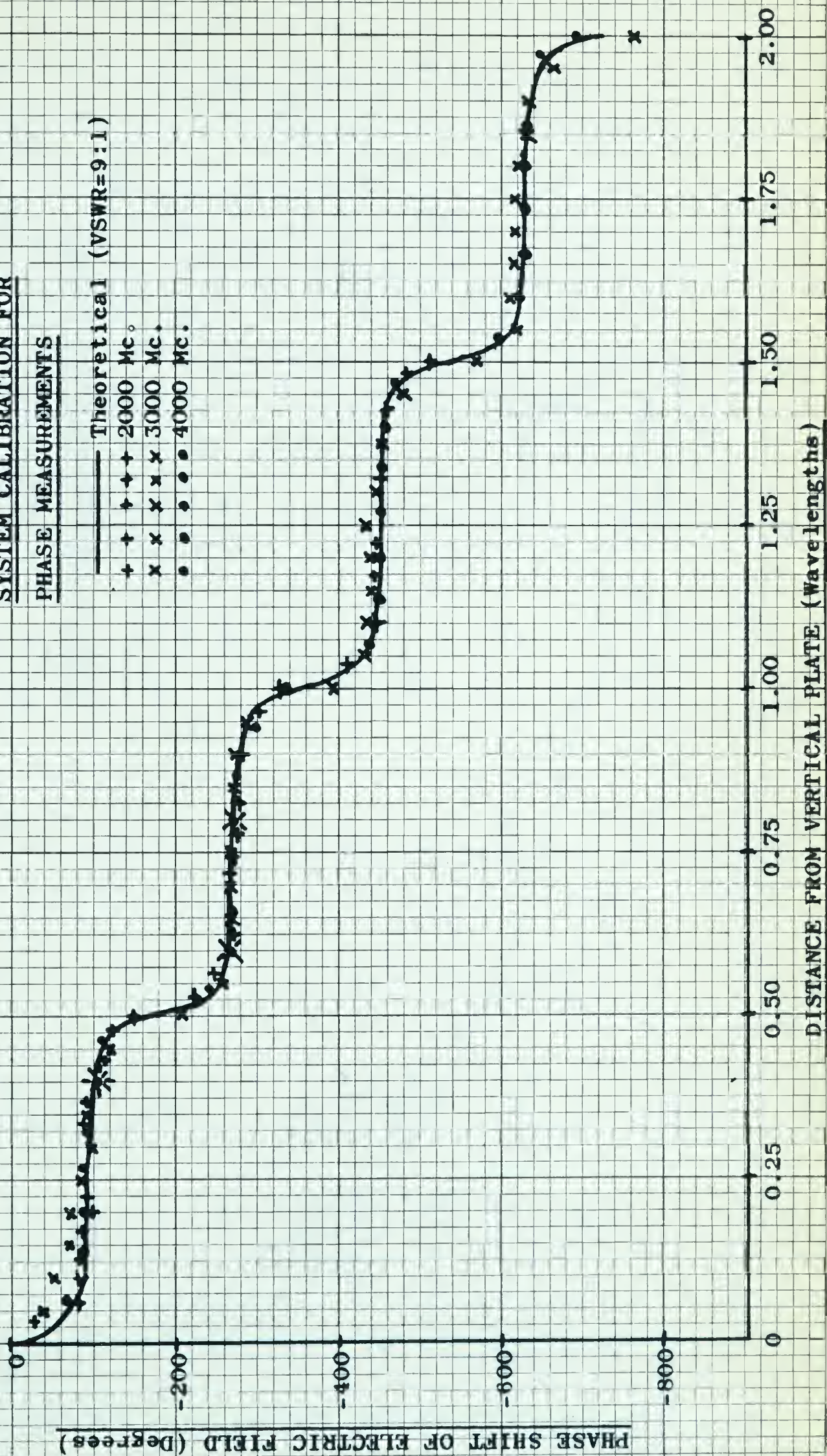


**FIGURE 20**

**SYSTEM CALIBRATION FOR**

**PHASE MEASUREMENTS**

- Theoretical (VSWR=9:1)
- + + + 2000 Mc.
- x x x 3000 Mc.
- • • 4000 Mc.

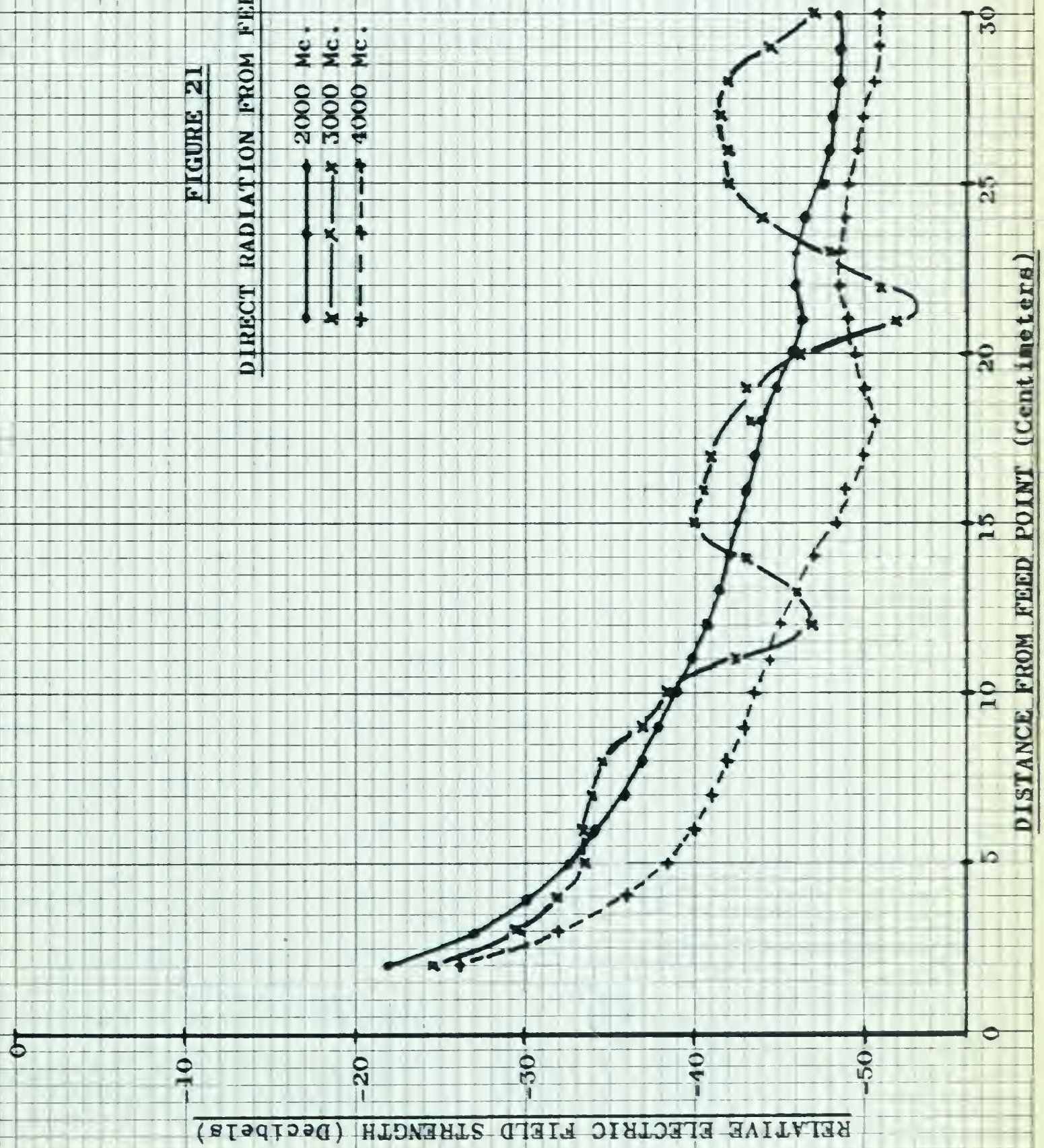








**FIGURE 21**  
**DIRECT RADIATION FROM FEED POINT**







direct radiation from the feed system with the antenna removed. The zero decibel level was taken to be the maximum signal measured with an antenna half structure soldered in place. As can be seen, the direct radiation is quite low and was found to have little effect on the measurements, with the possible exception of the area to the rear of the active region where the transmission line field is highly attenuated. Fortunately, the measurements made in this region are of no importance in this thesis.

#### 4.4 Radiation Pattern Measurements

In order to evaluate the theory, it is necessary to compare the shape of the calculated radiation patterns to those obtained experimentally.

The equipment for an antenna range was purchased during this semester, and a temporary installation set up for use in conjunction with the undergraduate laboratories. It was found to give very good results for measurements made in the azimuth plane.

A photograph of the receiving site is shown in Figure 22. A wooden structure with absorbing material tied to it was built around the turntable in order to minimize reflections from the walls of the room. The turntable is a Scientific-Atlanta Inc. Medium Duty Azimuth Positioner, Model PMA-5R and the recorder is a Scientific-Atlanta Inc. Series APR-20 Rectangular Recorder.

A Hewlett Packard Model 420A crystal detector was used with the recorder. The calibration of the crystal detector-recorder combination was checked using the receiver as a standard. Based on







FIGURE 22  
TEMPORARY ANTENNA RANGE RECEIVING SITE





a linear scale, it was found that the variations between the two were insignificant over the frequency range required for the pattern measurements.

The transmitting site was located at the other end of the room, a distance of about thirty feet. It consisted simply of an S band horn and the generator used for the measurements of the transmission line characteristics.









## 5. RESULTS

### 5.1 Presentation of Measured Data for Transmission Line Field

Measurements were made on the log periodic dipole antenna to determine the characteristics of the transmission line field for the parameters  $\tau = 0.95, 0.89, \text{ and } 0.81$ . The apex angle,  $\theta$ , for each of the above geometric ratios,  $\tau$ , was varied in steps of five degrees over the useable range of the antenna (measured VSWR of input impedance less than  $2:1^{15}$ ).

The log periodic dipole antenna was chosen since the apex angle for a particular geometric ratio could be varied simply by cutting the elements to the proper lengths. Therefore, it was necessary to build only three structures which represented a considerable saving in time and materials. This antenna does represent a simplification of the other types of log periodic antennas. However, the pattern of the complete log periodic antenna can be calculated using array theory once the pattern of each half structure has been determined using the leaky wave theory.

A length of brass welding rod 0.241 cms. in diameter was used for the boom. The half structure was mounted one centimeter above the ground plane as shown in Figure 14 on Page 52. Therefore, the characteristic impedance of the equivalent twin lead transmission line for the complete antenna,  $Z_0$ , is 340 ohms.<sup>28</sup> Number 14 copper wire was used for the elements which were soldered to the boom. Thus, the length to diameter ratio of the half wavelength





element varied from 46.1 at 2000 mc. to 26.4 at 3500 mc.

The measured results for the transmission line field are presented graphically in Appendix B. The position along the boom is measured from the apex of the antenna in wavelengths. The advantage of using this method is that the position of the half wave element for each graph remains fixed for all frequencies. Therefore, if there is no variation of the fields as the frequency is varied over a period, all the curves for a particular  $\zeta$  and  $\theta$  will be superimposed.

As previously stated, Bell, Elfving and Franks found that the VSWR of the transmission line wave was equal to the VSWR of the input impedance.<sup>5</sup> Isbell has measured the input impedance of log periodic antennas and found the VSWR to be less than 2:1 over the range of parameters for which the curves in Appendix B were made.<sup>15</sup> While the VSWR for various antenna parameters follows the trend of Isbell's impedance measurements, it is much greater than 2:1. This apparent discrepancy can be explained as follows: The measurements presented in Appendix B were made on a single boom mounted over a ground plane, which causes a mirror rather than complimentary image. The reflections from this mirror image are in phase with those from the real boom and so a large VSWR is obtained. For the actual antenna, the booms are complimentary, so that the reflections from them are  $180^\circ$  out of phase and tend to cancel. This results in the lower VSWR measured for the actual antenna. DuHamel and Ore have reported input impedance VSWR's of the

simply taking from 80.1 to 80.2 at 1000 m.

The amount of water for the experiment was 100 ml.

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is measured from the time of the onset of contraction.

At the end of the experiment the heart is removed and the

heart is placed in a solution of 10% formalin.

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order 5:1 to 8:1 if the elements are located less than a half wavelength from the ground plane.<sup>14</sup> A series of input measurements was made for  $\tau = 0.89$  and various apex angles. The VSWR's for the input impedance measurements agreed with those of the transmission line wave.

It was noticed that the position of the standing wave of the curves repeated with a period of  $2 \ln(1/\tau)$ . Thus, if the frequency is changed from  $f$  to  $\tau^{1/2} f$ , the position of the standing waves for the two frequencies will be  $90^\circ$  apart in terms of the transmission line field wavelength. This procedure was found to greatly aid in measuring the slope of the curves. The measurements for  $\tau = 0.95$  are at random frequencies because they were made before the advantages of this property were realized. A ground plane could not be used for  $\tau$ 's much less than 0.81 since the VSWR of the transmission line wave is nearly as large as the drop in the signal level in the active region. Fortunately, log periodic dipole antennas with  $\tau$ 's less than 0.8 break up excessively as the frequency is varied over a period.<sup>3</sup> Therefore, this value represents the lower limit of  $\tau$ .

A vertical line was drawn on each graph at the position of a hypothetical half wavelength element in order to give a reference point.

The results of the measurements presented in Appendix B for the transmission line field are given in Figures 23 and 24. They follow general trends which could have been predicted intuitively.



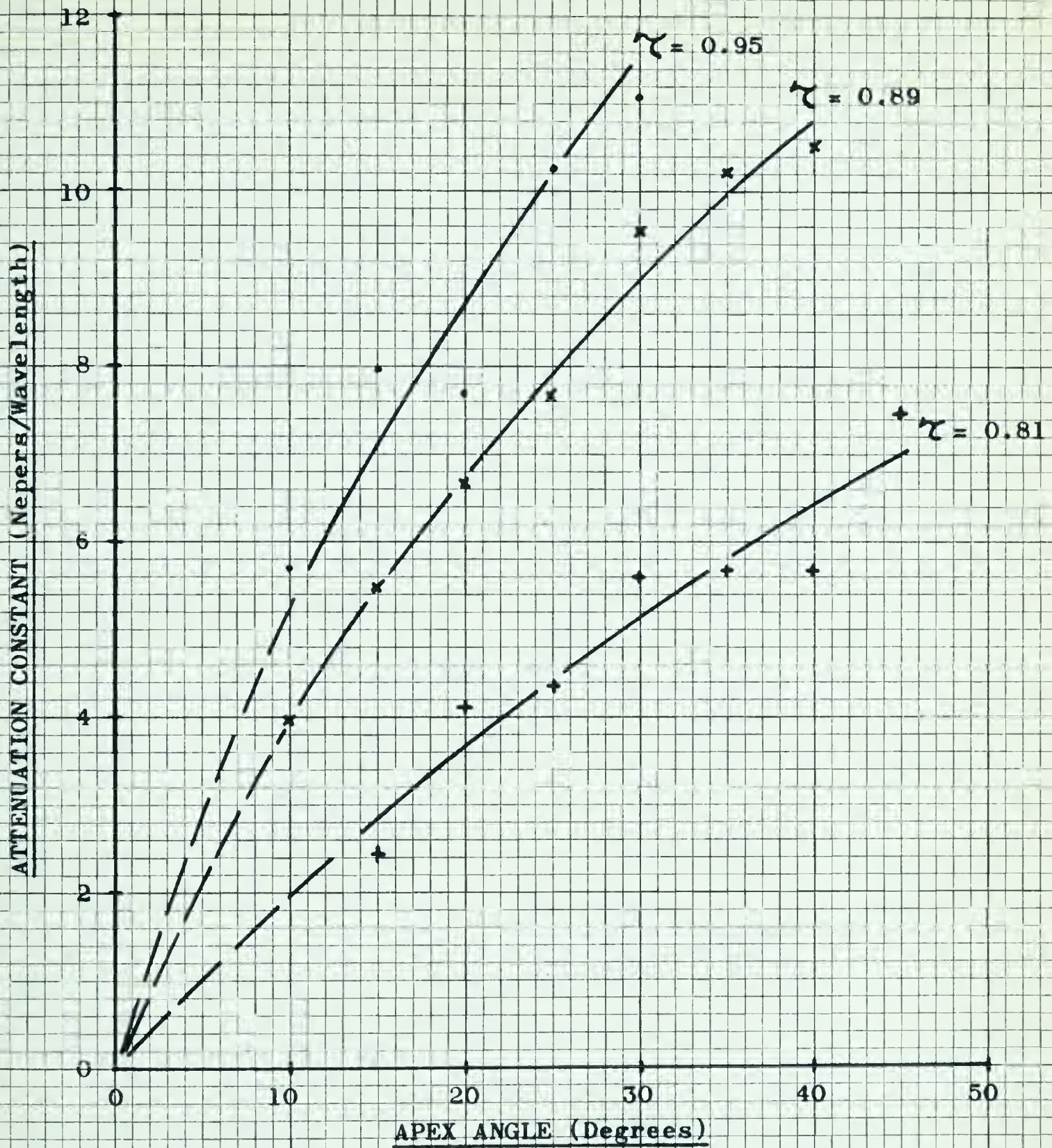
order 0.1 to 0.3 in the elements are located less than 0.1 cm from the ground plane. A series of input measurements are made for  $\Sigma = 0.95$  and various other values. The results for the input impedance measurements are shown in the transmission line theory.

It was noticed that the position of the standing wave of the curve repeated with a period of  $\frac{1}{2} \lambda$ . Thus, if the frequency is changed from  $f$  to  $f'$ , the position of the standing wave for the two frequencies will be  $90^\circ$  apart in terms of the transmission line fluid wavelength. This procedure was found to greatly aid in measuring the sign of the curves. The measurements for  $\Sigma = 0.95$  are at random frequencies because they were made before the advantages of this property were realized. A ground plane could not be used for  $\Sigma$  less than 0.95 since the wave of the transmission line wave is nearly as large as the wave in the signal level in the active region. Furthermore, for optimum signal response with  $\Sigma$  less than 0.95 some of the elements are in the ground plane over a period. Therefore, this value represents the lower limit of  $\Sigma$ .

A vertical line was drawn on each graph at the position of a hypothetical half wavelength element in order to give a reference point.

The results of the measurements presented in Figures 5 to 7 are the transmission line fluid are given in Figures 5 to 7. These values represent results which would have been obtained if the





**FIGURE 23**  
**ATTENUATION CHARACTERISTICS**  
**OF TRANSMISSION LINE FIELD**  
**FOR LOG PERIODIC DIPOLE ANTENNA**







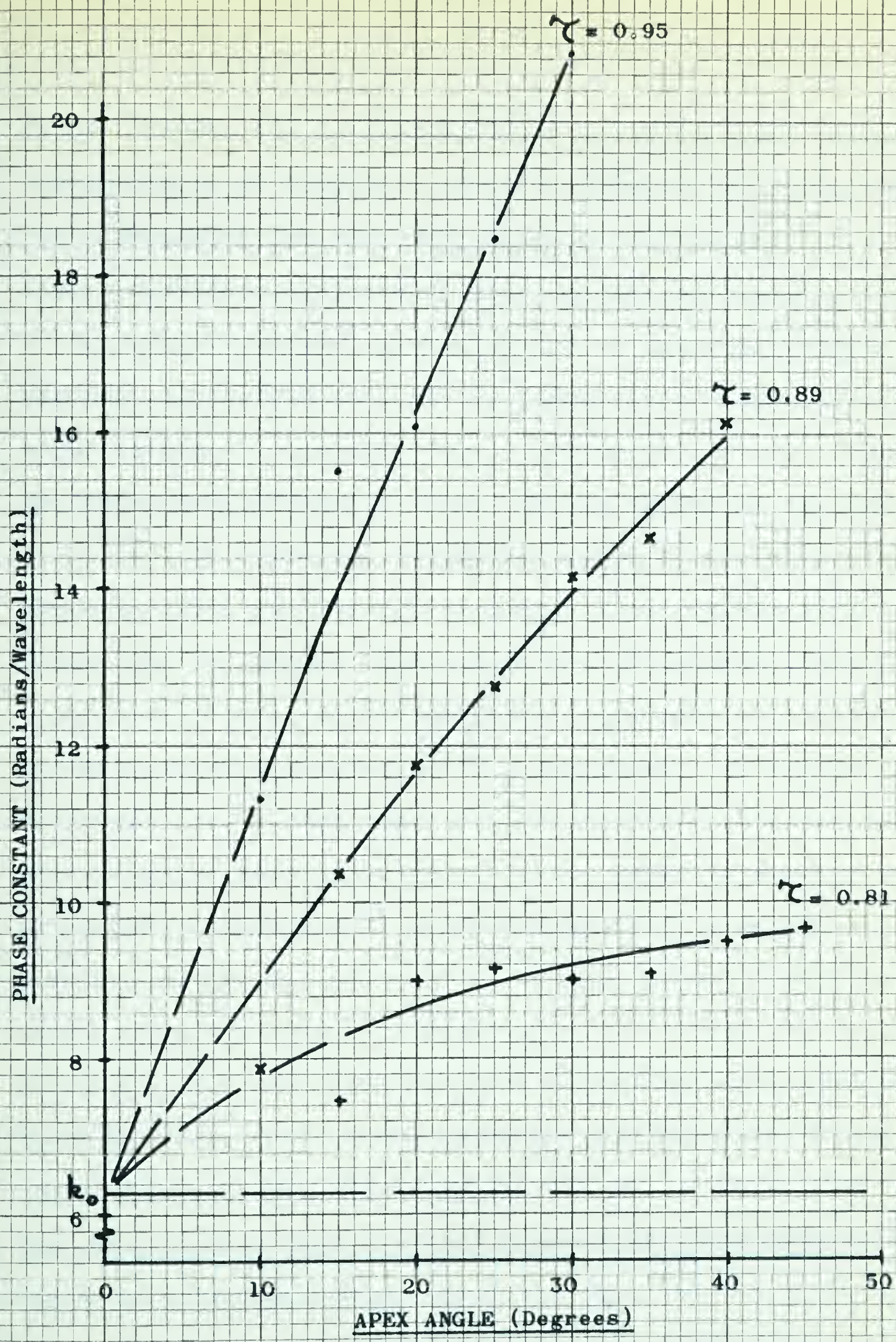


FIGURE 24

PHASE CHARACTERISTICS OF TRANSMISSION  
LINE FIELD FOR LOG PERIODIC DIPOLE ANTENNA





In the limiting case as the apex angle approaches zero, the antenna degenerates to an infinite parallel transmission line along which the transmission line wave will propagate with the velocity of light and without attenuation. Thus, as the apex angle is decreased, it is expected that the attenuation and the phase constants will decrease, approaching zero and  $k_0$  respectively in the limiting case as the apex angle approaches zero. This trend was observed and so the curves were extended to these limiting cases. As the geometric ratio,  $\tau$ , is increased, more elements will be closer to a half a wavelength long and they will be spaced closer together with the result that the antenna will radiate more efficiently. The resulting higher rate of decay of the transmission line wave will yield a higher attenuation constant as  $\tau$  is increased. For a given length of element, it has been found that the velocity of the wave passing over parasitic elements is reduced as the number of elements per wavelength is increased.<sup>22</sup> A similar result is to be expected for directly excited elements so that the phase constant should increase as the geometric ratio is increased for a particular apex angle. The same points could also be argued by considering the limiting cases as the geometric ratio approached zero for which  $\alpha_\tau$  and  $\beta_\tau$  approach zero and  $k_0$  respectively.

## 5.2 Comparison of Calculated and Measured Radiation Patterns for Log Periodic Dipole Antenna

The shapes of the far field radiation patterns were calculated using the theory of Section 3 and Figures 23 and 24. In order to



In the following table is given the results of the analysis

of the samples of the various types of material.

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table. It is assumed that the material is of the same quality as

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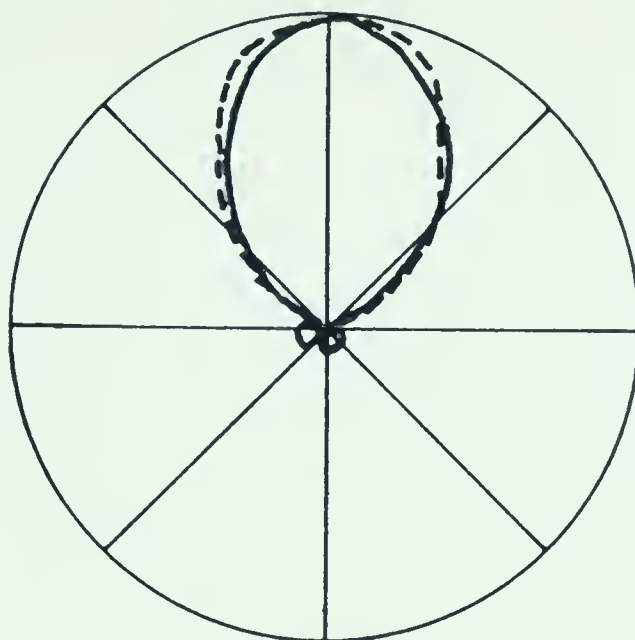
obtain the most meaningful comparison, the elements were made from the same size wire as those used to measure the attenuation and phase constants. The same frequencies were also used in an attempt to minimize the effect of changing the height to diameter ratio of the elements. Unfortunately, the RG-142/U rigid coaxial cable which is most suitable for constructing the boom carrying the feed cable has a diameter of 0.355 cms. Further, it is not practical to separate the booms two centimeters because this separation is not small compared to the wavelengths used and would have a noticeable effect on the H plane radiation pattern. Therefore, a separation of 0.6 cms. was used. This resulted in a  $Z_0$  of 140 ohms for the antennas used to make the pattern measurements compared to 340 ohms for the structure used to measure the characteristics of the transmission line field.

It has been found that increasing  $Z_0$  tends to broaden the beamwidth very slightly.<sup>3</sup> However, this effect is negligible compared to the degree of accuracy obtained using the theory developed in this paper.

Calculated and measured far field patterns for several parameters are shown in Figures 25 to 33. A comparison of the calculated and measured half power beamwidths is shown in Figures 34 and 35 for the E and H planes respectively. Since the calculated patterns do not account for any variations, they can be thought of as representing the average radiation pattern. Therefore, the far field patterns were measured at several frequencies over a period

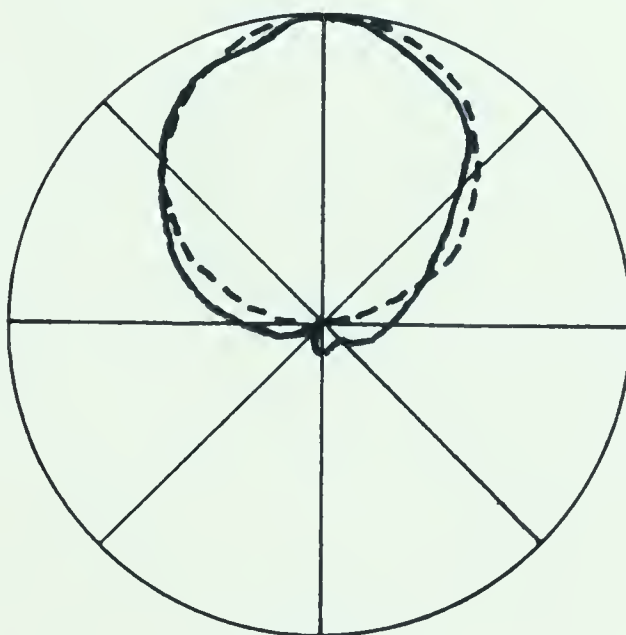






E Plane

——— Measured  
 - - - - - Calculated

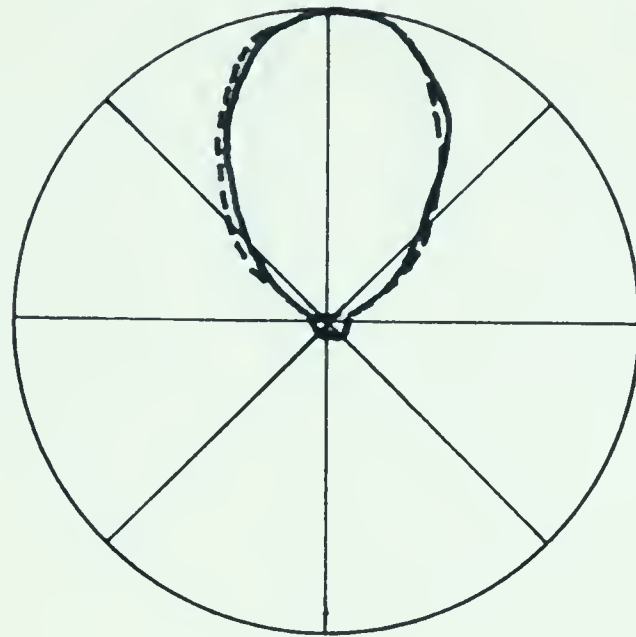


H Plane

FIGURE 25

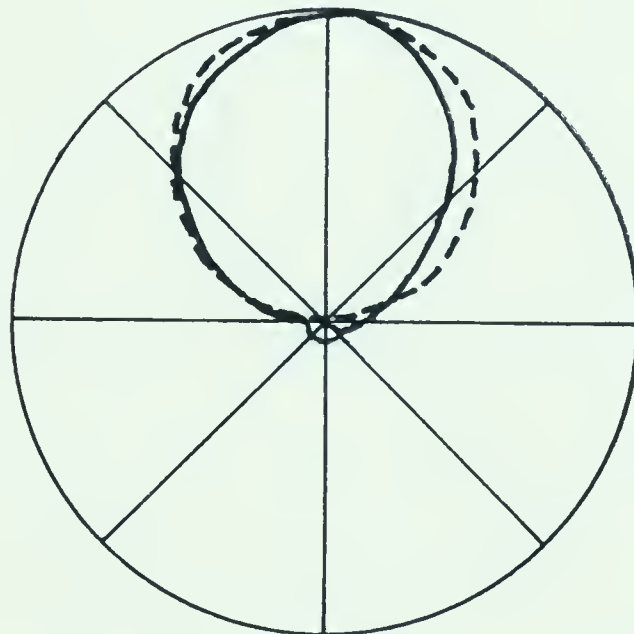
RADIATION PATTERNS OF LOG PERIODIC  
 DIPOLE ANTENNA FOR  $\gamma = 0.95$  AND  $\phi = 30^\circ$





E Plane

———— Measured  
 ----- Calculated



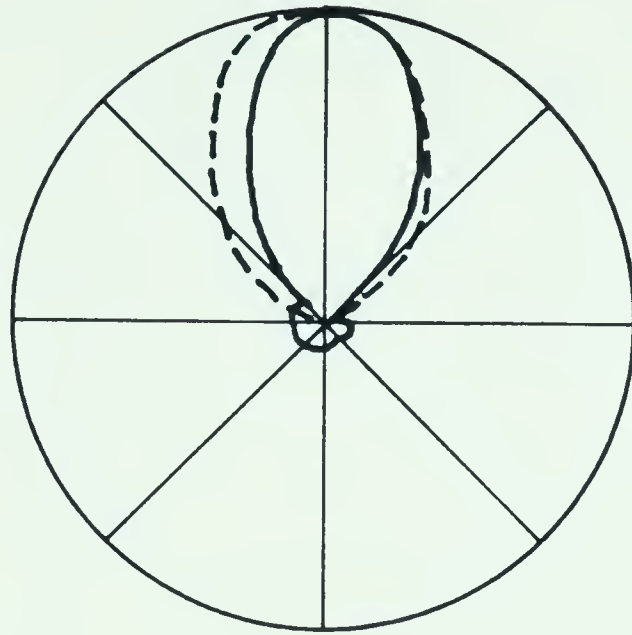
H Plane

FIGURE 26

RADIATION PATTERNS OF LOG PERIODIC  
 DIPOLE ANTENNA FOR  $\gamma = 0.95$  AND  $\theta = 20^\circ$

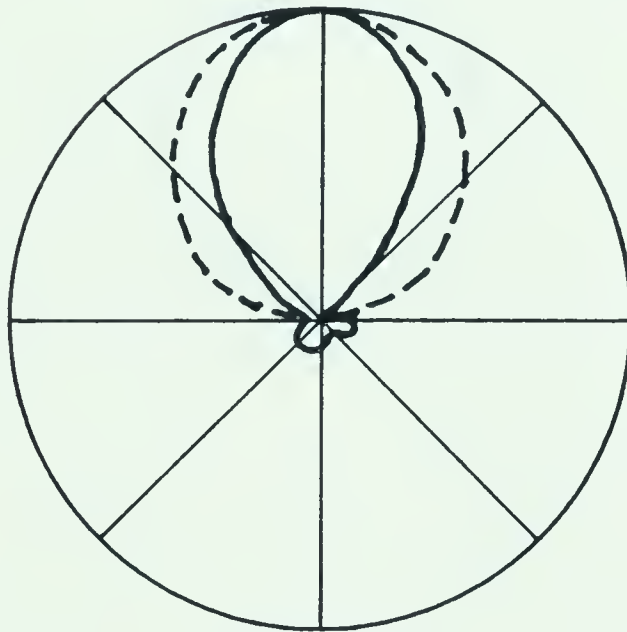






E Plane

———— Measured  
 - - - - - Calculated



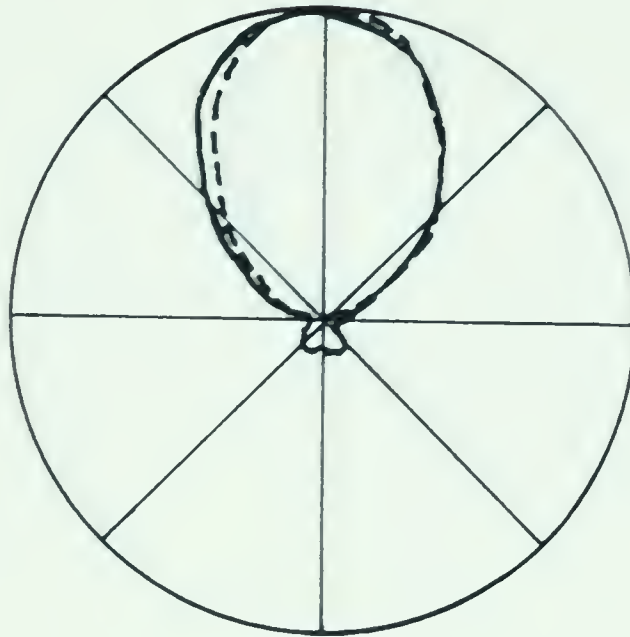
H Plane

FIGURE 27

RADIATION PATTERNS OF LOG PERIODIC  
 DIPOLE ANTENNA FOR  $\gamma = 0.95$  AND  $\theta = 10^\circ$

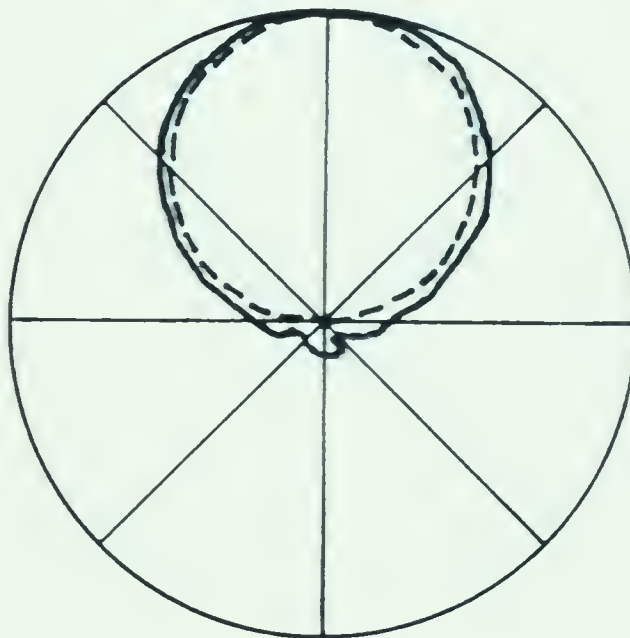






E Plane

————— Measured  
 - - - - - Calculated

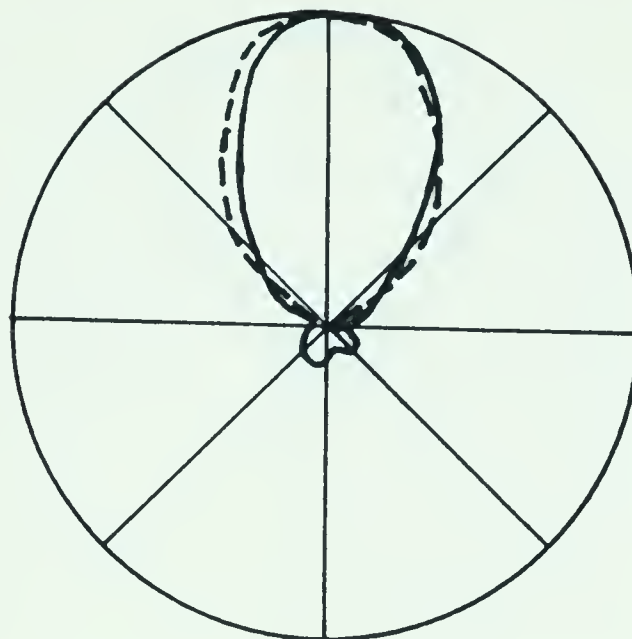


H Plane

FIGURE 28

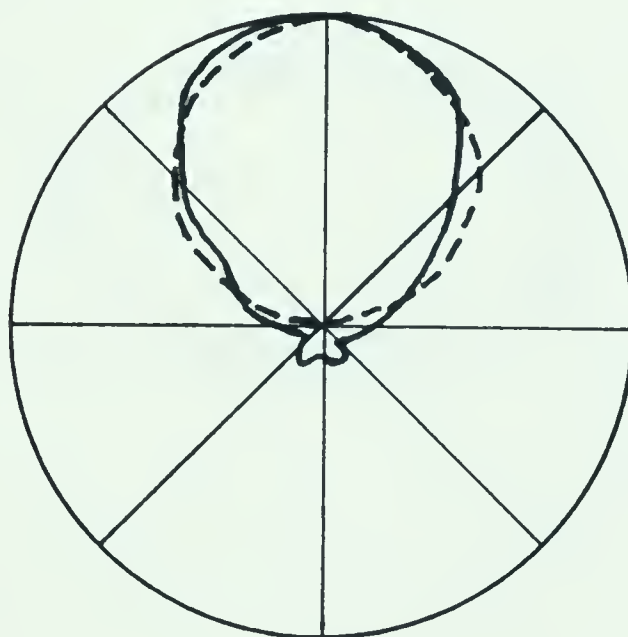
RADIATION PATTERNS OF LOG PERIODIC  
 DIPOLE ANTENNA FOR  $\tau = 0.89$  AND  $\theta = 40^\circ$





E Plane

————— Measured  
 ----- Calculated



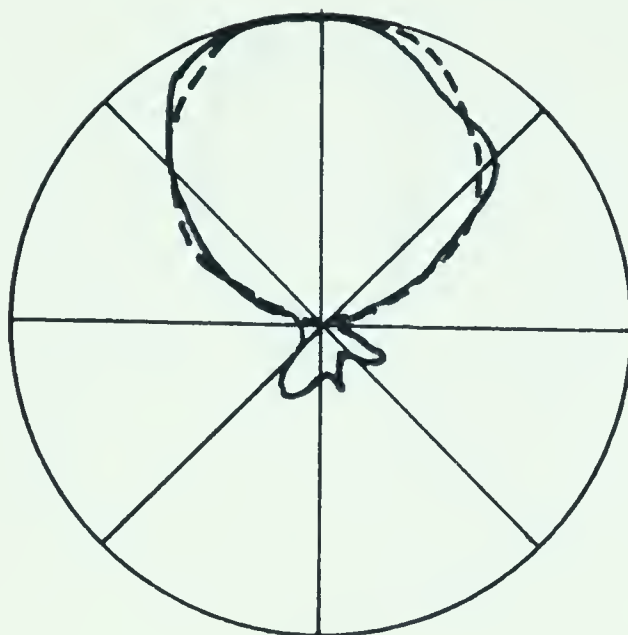
H Plane

FIGURE 29

RADIATION PATTERNS OF LOG PERIODIC  
 DIPOLE ANTENNA FOR  $\gamma = 0.89$  AND  $\theta = 25^\circ$

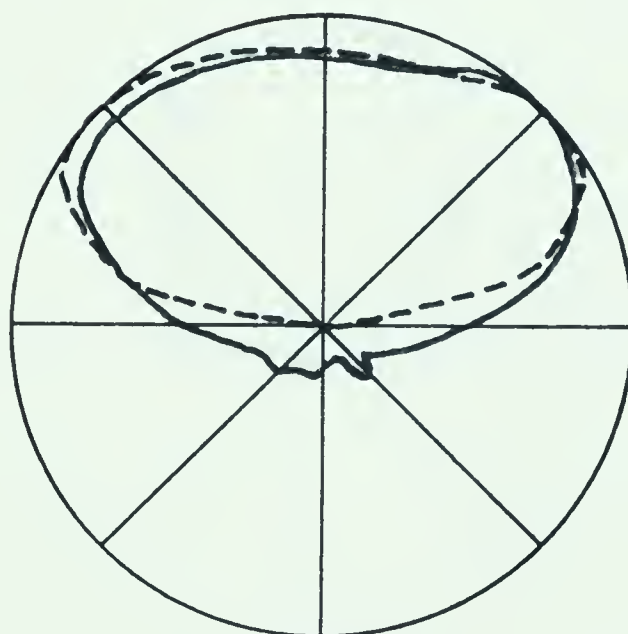






E Plane

————— Measured  
 - - - - - Calculated



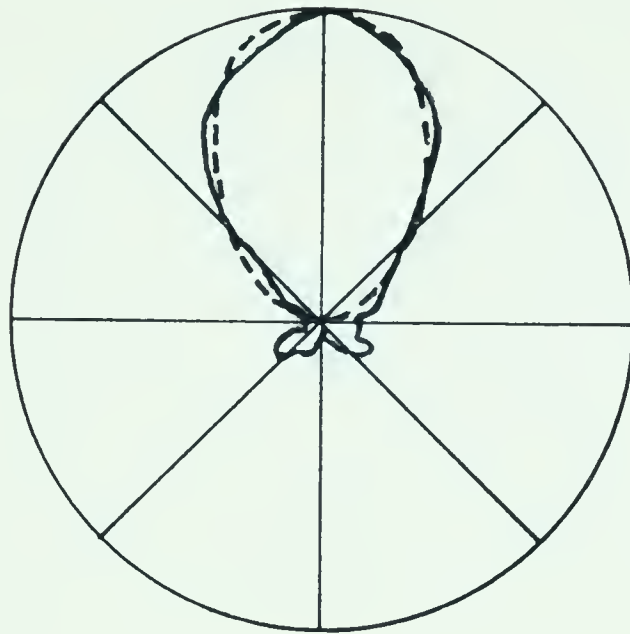
H Plane

FIGURE 30

RADIATION PATTERNS OF LOG PERIODIC  
 DIPOLE ANTENNA FOR  $\gamma = 0.89$  AND  $\theta = 10^\circ$

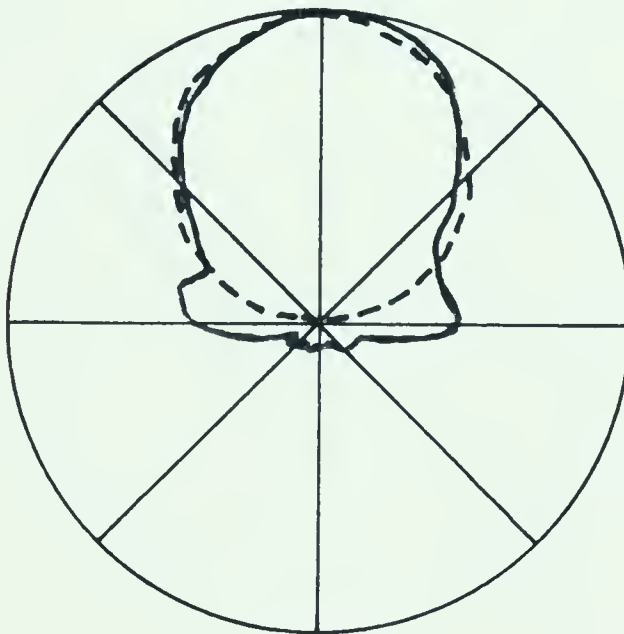






E Plane

————— Measured  
 - - - - - Calculated

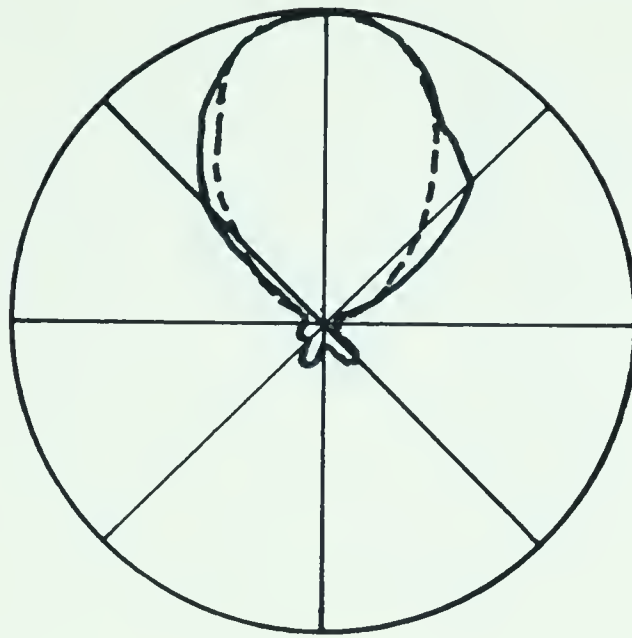


H Plane

FIGURE 31

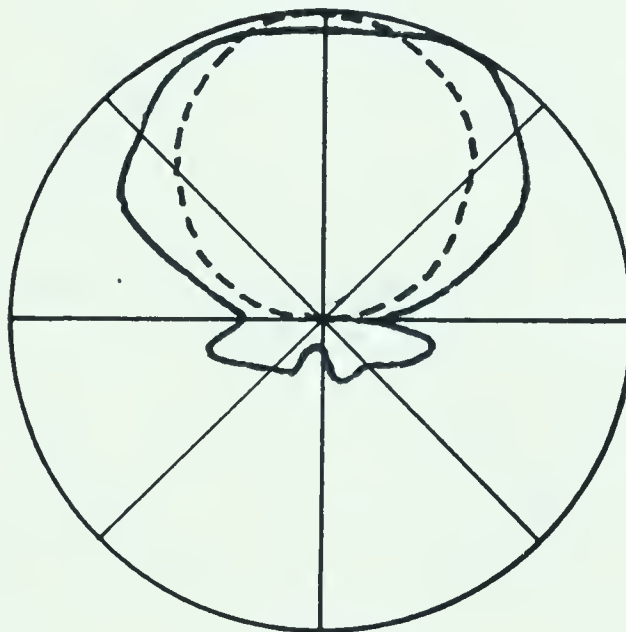
RADIATION PATTERNS OF LOG PERIODIC  
DIPOLE ANTENNA FOR  $\tau = 0.81$  AND  $\theta = 40^\circ$





E Plane

————— Measured  
 - - - - - Calculated



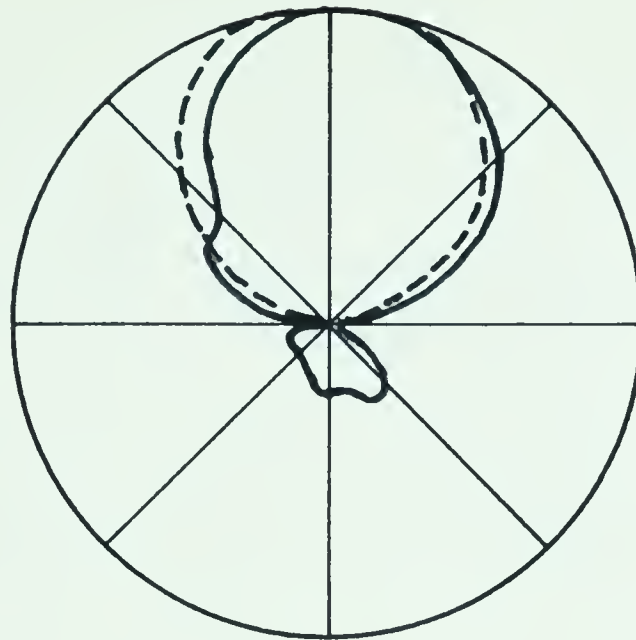
H Plane

FIGURE 32

RADIATION PATTERNS OF LOG PERIODIC  
DIPOLE ANTENNA FOR  $\tau = 0.81$  AND  $\theta = 30^\circ$

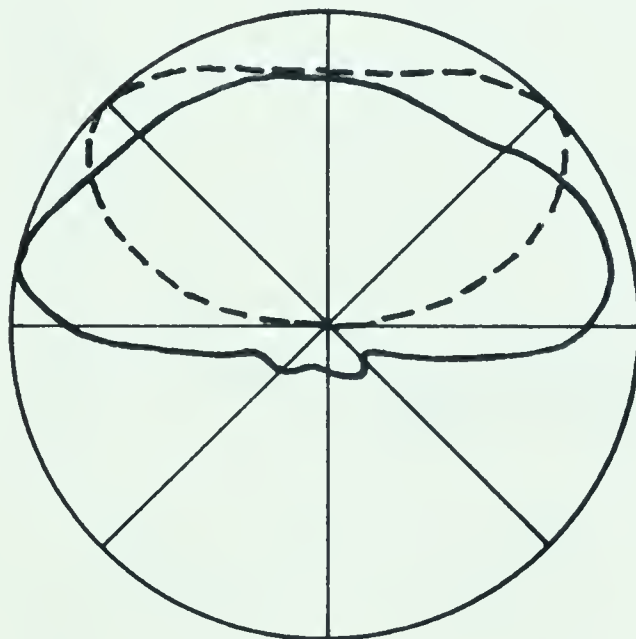






E Plane

————— Measured  
 - - - - - Calculated



H Plane

FIGURE 33

RADIATION PATTERNS OF LOG PERIODIC  
 DIPOLE ANTENNA FOR  $\gamma = 0.81$  AND  $\theta = 20^\circ$





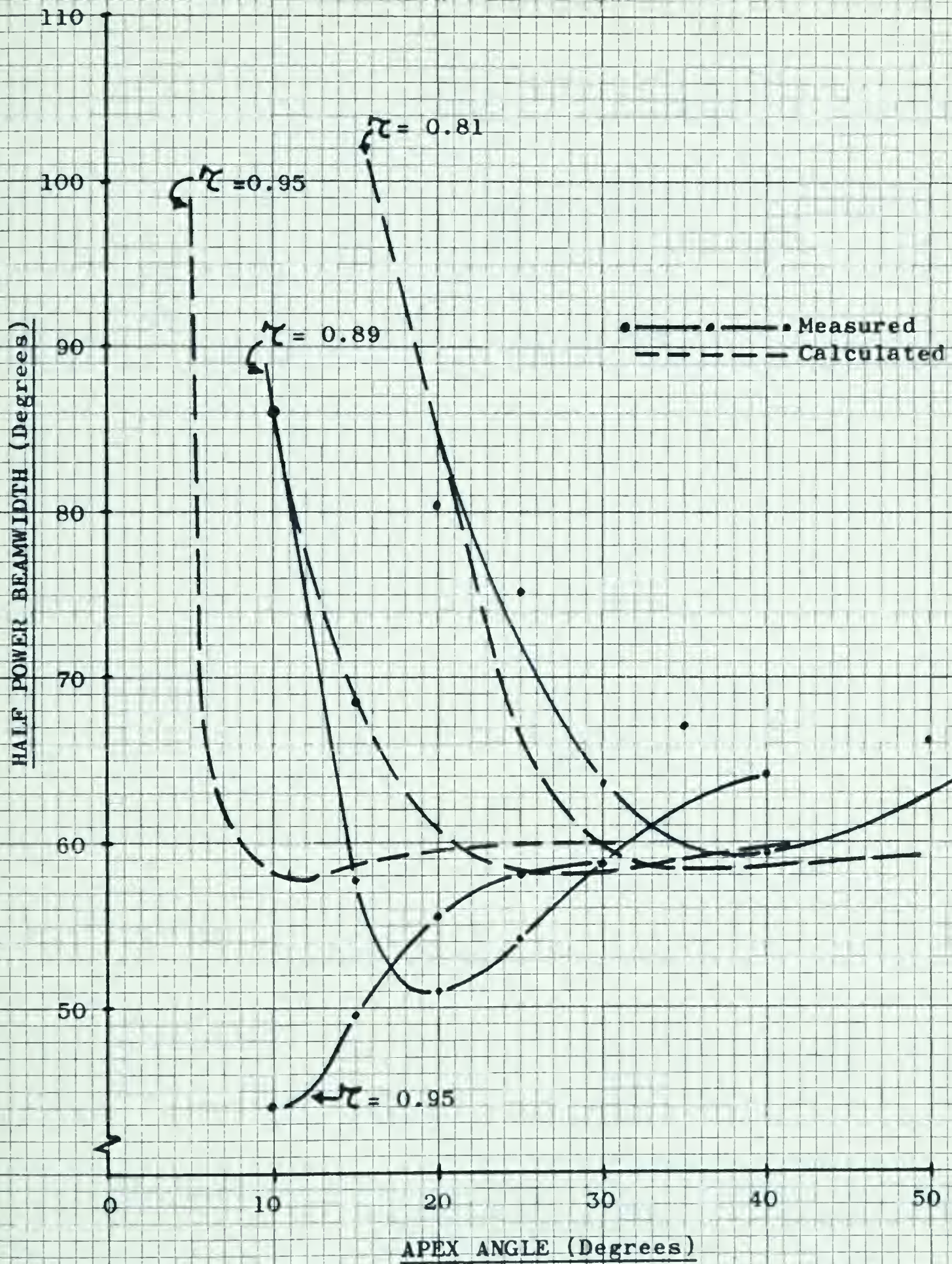


FIGURE 34

E PLANE HALF POWER BEAMWIDTH  
FOR LOG PERIODIC DIPOLE ANTENNA







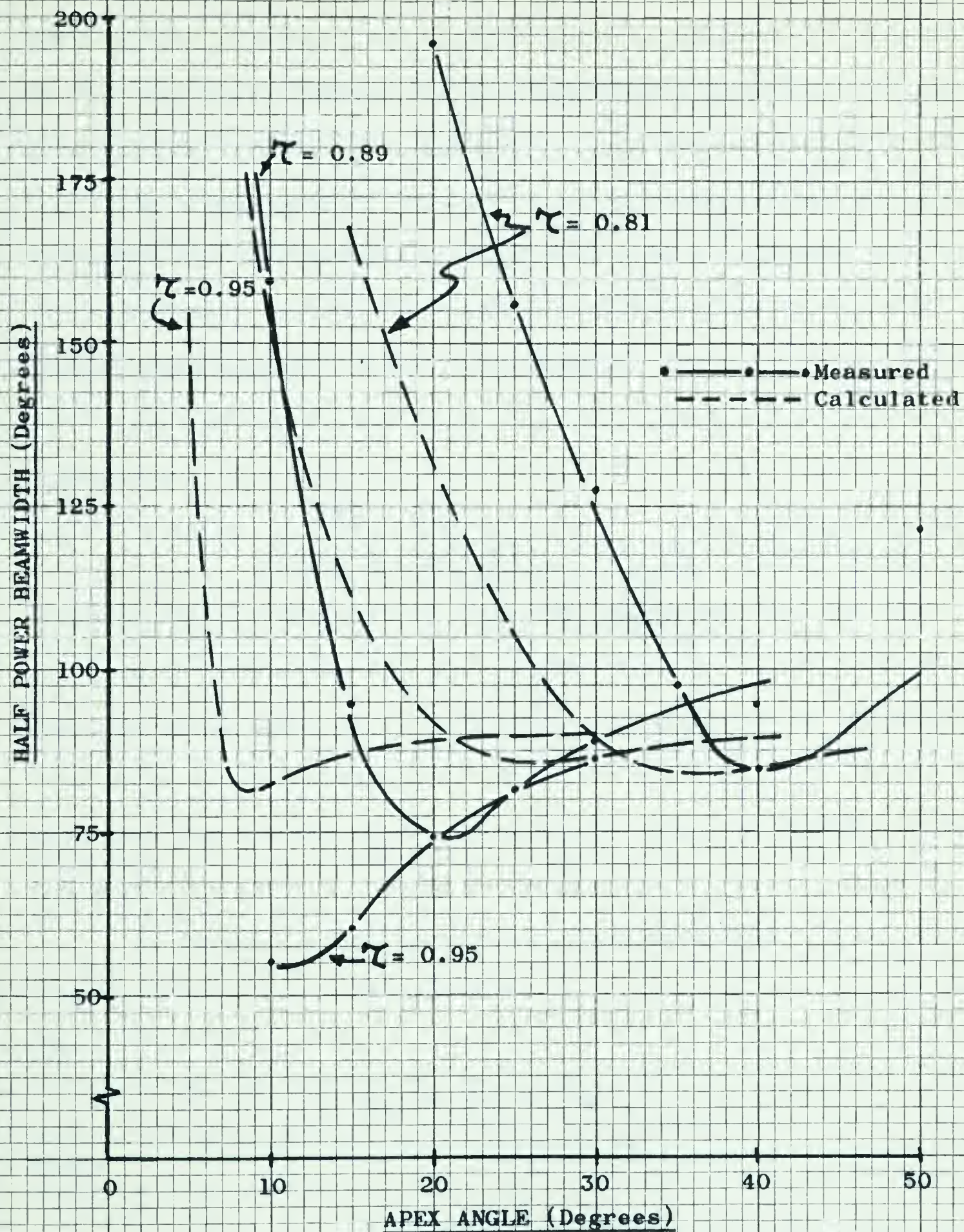


FIGURE 35

H PLANE HALF POWER BEAMWIDTH  
FOR LOG PERIODIC DIPOLE ANTENNA







and averaged to obtain the measured results presented in Figures 25 to 35. The variations of the patterns over a period were found to be quite small within the useable range of antenna parameters.

An examination of Figures 34 and 35 reveals that the minimum beamwidth for each geometric ratio occurs at a different apex angle. For larger apex angles, the beamwidth increases slowly and for smaller apex angles, it increases rapidly.

The effects of the leaky wave pole can be seen from the expression

$$f_i(\theta) = \frac{-\frac{1}{2}k_o^2 \cos \theta}{\left(\cos^2 \theta - \frac{\beta_r^2 - \alpha_r^2}{k_o^2}\right) - j \frac{2\beta_r \alpha_r}{k_o^2}} = \frac{-\frac{1}{2}k_o^2 \cos \theta}{(\cos^2 \theta - A) - jB} \quad (51)$$

which is the H plane pattern function. Equation (51) can be readily obtained by combining (25) and (40) on Pages 29 and 34 respectively.

When the denominator is at a minimum, the magnitude of  $f_i(\theta)$  will tend to be a maximum depending on the value of  $\cos \theta$ . It can be seen that the minimum of the denominator occurs at  $\cos \theta = \pm A^{1/2}$  and the sharpness of this minimum is determined by the magnitude of B.

In Section 3, the assumptions were made that

$$\alpha_R = \alpha_T \quad (47)$$

and

$$\beta_R = \beta_T - \pi/\sigma \quad (45)$$

where

$$\sigma = \frac{1 - \tau}{4 \tan \frac{\theta_1}{2}} \quad (46)$$

Using these equations and Figures (23) and (24), it is readily seen that the magnitudes of both  $\alpha_R$  and  $\beta_R$  increase as the apex angle

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THESE RESULTS WERE OBTAINED BY A STUDY OF THE EFFECTS OF VARIOUS FACTORS ON THE GROWTH OF THE BACTERIA.



is widened. The magnitude of  $\beta_R$  varies faster than the magnitude of  $\alpha_R$  and for large apex angles, the magnitude of  $\beta_R$  is much larger than that of  $\alpha_R$ . Therefore, the magnitudes of A and B are proportional to the apex angle, being large for wide apex angles and small for narrow apex angles.

For wide apex angles, both A and B are large so the denominator of (51) is nearly constant for all azimuth angles. As a result, the theory predicts that the H plane pattern function will asymptotically approach a cosine distribution for very large apex angles.

This distribution can also be predicted intuitively as a first order approximation to the actual antenna having a large apex angle. In the limit as the apex approaches  $180^\circ$ , all the elements of the antenna will be located vanishingly close to the apex with the result that the spacings between the elements will be very small. Therefore, the net effect of the elements in the active region will be essentially that of a single half wavelength dipole. The elements to the rear of the active region, which are longer, will not be excited (providing they are of a non-resonant length). Thus, the structure can be simplified to that of a driven half wavelength dipole located immediately in front of a longer non-resonant element which acts as a reflector. For the idealized case, the reflector will create a perfect image dipole located vanishingly close to the driven element. In the H plane, these two idealized dipoles form a Hertzian dipole pair whose pattern function is given by a cosine



is shown. The magnitude of  $\mathcal{E}_2$  varies linearly with the magnitude of  $\mathcal{E}_1$  and the phase angle, the magnitude of  $\mathcal{E}_2$  is more than that of  $\mathcal{E}_1$ . Therefore, the magnitude of  $\mathcal{E}_2$  and  $\mathcal{E}_1$  are proportional to the open angle. Being linear for this angle and small for small angle.

For wide open angle, with a small angle and the magnitude of  $\mathcal{E}_2$  is nearly constant for all angles. As a result, the theory predicts that the linear pattern formed will be approximately constant a cosine distribution for very large angles.

This distribution can also be examined initially as a first order approximation to the actual pattern. Since the angle in the limit is the open angle  $\theta_0$ , all the elements of the antenna will be located approximately along the open angle. The result that the elements are located will be very small. Therefore, the net effect of the elements in the active region will be essentially that of a single unit with a cosine distribution. The elements in the rest of the active region, when the angle, will not be excited inwards from the open-angle region. Thus, the waveform can be modified to that of a cosine only waveform. Hence focusing immediately is found in a focused non-waveform signal which acts as a reflector. For the focused case, the waveform will create a focused beam. Since the waveform is given in the active region, from the focusing region, the waveform will be given as a cosine.

distribution.<sup>27</sup>

As the apex angle is reduced, both A and B in (51) decrease. In this region A is still greater than unity so that the minimum of the denominator occurs at the azimuth angle  $\Theta = 0^\circ$ . Since the magnitude of B is proportional to the apex angle, the beamwidth gradually decreases as the apex angle is reduced. To a first order approximation, the minimum beamwidth for a particular geometric ratio,  $\zeta$ , occurs at the azimuth angle for which  $A = 1$ , or

$$\beta_R^2 - \alpha_R^2 = k_o^2 \quad (52)$$

It is easily seen that B will be large for this apex angle unless  $\alpha_R$  is very small. Since the active region for successful log periodic structures is fairly short,  $\alpha_R$  must be reasonably large in order to have the energy in the transmission line wave efficiently transferred to the radiation wave. Thus, as often happens in engineering problems, the two requirements are contradictory and a compromise must be made.

The variations of A and B with respect to the apex angle were found to have secondary effects on the patterns, and it was found that the minimum beamwidth for a particular geometric ratio occurred at a slightly wider apex angle than that given by (52).

With a further reduction in the apex angle, both A and B continue to decrease in magnitude. In this region A is less than unity and the azimuth angle for which the denominator of (51) is a minimum begins to scan toward broadside. B becomes rather small and this minimum can become quite sharp with the result that the







maximum value of  $f_1(\theta)$  also scans towards broadside. Good examples of this behavior are shown in Figures 30 and 33.

Although Figures 25 to 35 show that the calculated patterns agree with those obtained experimentally with a reasonably good degree of accuracy, the theory does have some failings. First, the measured patterns do not asymptotically approach a cosine distribution for large apex angles. Figures 34 and 35 show that the beamwidth in this region is inversely proportional to the geometric ratio and does not asymptotically approach a single beamwidth for all geometric ratios as the theory predicts. Figures 25 and 28 are examples of the patterns in this region and show that the theory is valid to a first order approximation. Surprisingly, both the measured and calculated patterns reveal that this region begins for apex angles much smaller than the limiting case of  $180^\circ$ . As might be expected from the qualitative analysis given above, the apex angle for which this region begins is inversely proportional to the geometric ratio. Structures built in this region of the parameters are not too practical since the VSWR of the input impedance is fairly high.<sup>15</sup> The curves presented in Figures 34 and 35 extend approximately over the range for which the measured VSWR of the input impedance is less than 2:1 and reasonably good agreement between the measured and calculated beamwidths is obtained. The impedance measurements indicate that the VSWR will increase rapidly for larger apex angles outside the range shown in Figures 34 and 35.<sup>15</sup>

For small apex angles, the pattern maximum scans toward

maximum value of  $\sqrt{f(\omega)}$  also shows towards resonance. Good examples

of this behavior are shown in Figures 26 and 27.

Although Figure 27 is 20 times that the calculated values

agree with those obtained experimentally with a reasonably good

degree of accuracy, the theory does not seem to follow. First, the

measured values do not satisfactorily approach a curve with

frequency for large  $\omega$  values. Figures 26 and 27 show that the

behavior in this region is inversely proportional to the frequency

ratio and does not necessarily approach a single behavior for

all frequencies ratios as the theory predicts. Figures 26 and 27 are

examples of the behavior in this region and show that the theory is

valid to a first order approximation. Consequently, both the

measured and calculated values reveal that this region gives the

best approximation to the limiting case of  $\omega \rightarrow 0$ . As might

be expected from the qualitative analysis given above, the theory

for which this region begins is inversely proportional to the frequency

ratio. Distortionless half in this region of the parameter set 26

the practical case the VSWR at the input impedance is fairly high.

The curves presented in Figures 26 and 27 show approximately equal

the ratio for which the measured VSWR of the input impedance is less

than 2:1 and reasonably good agreement between the measured and

calculated behavior is obtained. The agreement at resonance

indicates that the VSWR will increase rapidly for larger  $\omega$  values

outside the range shown in Figures 26 and 27.

For small  $\omega$  values, the behavior becomes more complex



broadside as previously described. This can also be predicted intuitively from the relationship

$$\beta_R = \beta_T - \frac{\pi}{\sigma} \quad (45)$$

For small apex angles,  $\sigma$  becomes large so that  $\beta_R$  can become zero and might even have the same sign as  $\beta_T$  for certain parameters. When  $\beta_R$  is zero, all the elements are excited in phase. This is the condition for broadside radiation from other types of antennas. Figure 33 reveals that the measured patterns tend to support this reasoning. Because of the cosine term in the numerator of (51), the theory cannot predict broadside radiation although it certainly does predict the scanning of the maximum toward broadside. However, the measured far field patterns break up rather badly for these small apex angles as the frequency is varied over a period, and so the antennas are not of any practical use in this region.

Figure 36, for  $\gamma = 0.81$  and  $\theta = 15^\circ$ , shows the break up of the patterns in this region of the parameters. The patterns were measured over a period and found to fall within the crosshatched area between the two curves. It was observed that this break up of the patterns occurred rather suddenly. The measured patterns for  $\gamma = 0.81$  and  $\theta = 20^\circ$  showed a much smaller variation as the frequency was changed. The variations of the patterns for most of the other parameters would have only caused a thickening of the line if the patterns were drawn to the same scale used in Figure 36.

As might be expected, the shape of the measured E plane



provided is previously described. This can also be modified

intuitively from the relationship

$$\Theta_s = \Theta_T - \frac{r}{R}$$

for small spot angles,  $\Theta$  becomes large so that  $\Theta_s$  can become zero

and might even take the same sign as  $\Theta$  for certain geometries.

From  $\Theta_s$  it now, with the previous are added in figure. This is the

condition for providing radiation from a spot type of antenna.

Figure 20 shows that the measured radiation from the antenna is

reasonable. Because of the cosine term in the denominator of (11),

the theory cannot predict maximum radiation although it correctly

does predict the maximum at the antenna focus distance. However,

the measured far field pattern does not exhibit nulls for spot

small spot angles as the theory is valid over a period, and as

the antennas are not of any particular size in this region.

Figure 20, for  $f = 0.51$  and  $\theta = 15^\circ$ , shows the beam up of

the pattern in this region of the diagram. The pattern was

measured over a period and found to fall within the theoretical

area between the two curves. It was observed that the beam is at

the pattern occurred at the antenna. The measured pattern for

$f = 0.51$  and  $\theta = 15^\circ$  showed a main lobe pattern as the

frequency was changed. The nulls at the pattern for most of

the other frequencies were also found to be at the same angle as the

if the pattern were drawn to the same scale as in figure 20.

As with the previous, the same at the measured  $\theta$  angle



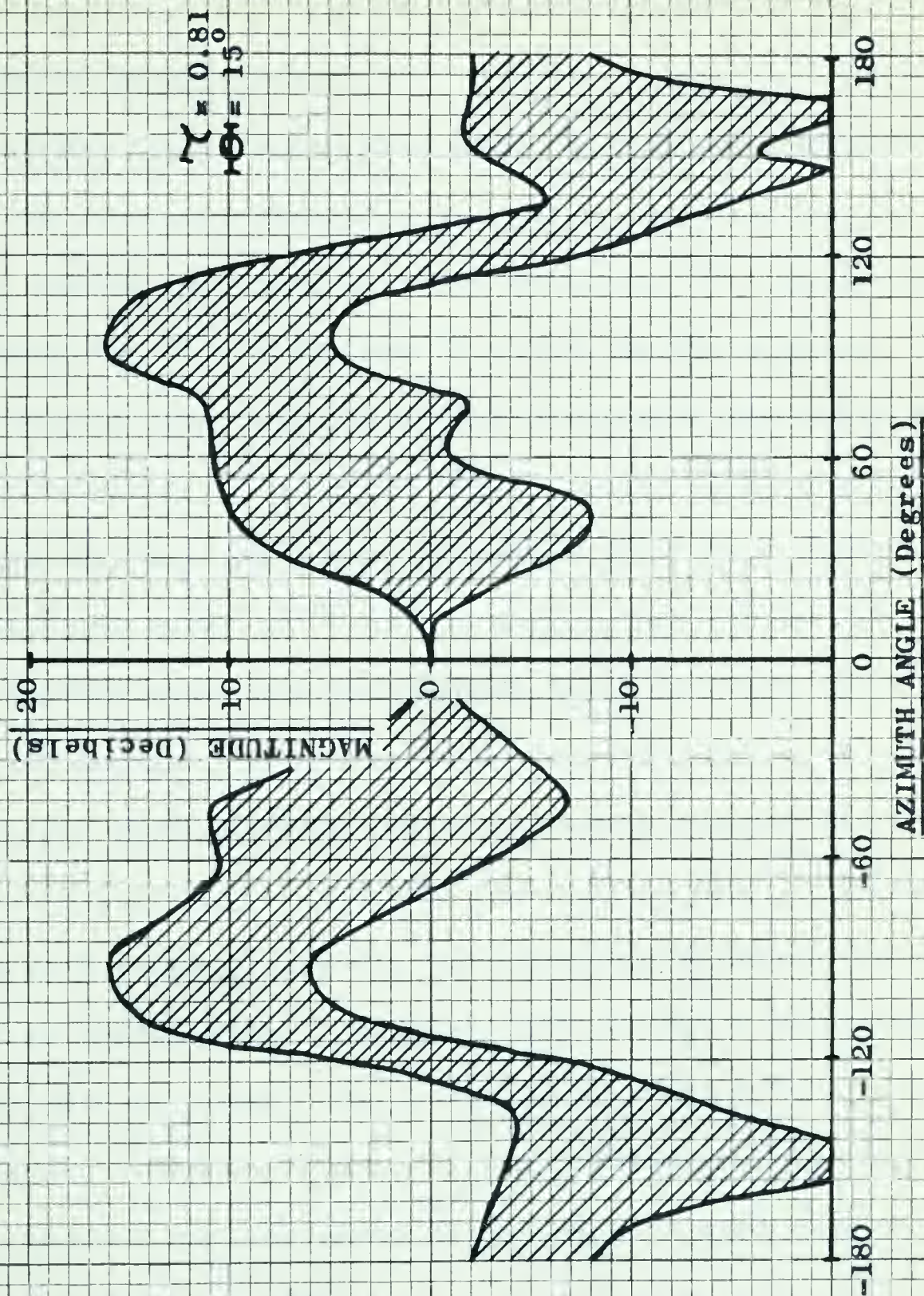


FIGURE 36

VARIATION OF H PLANE PATTERN  
 FOR A LOG PERIODIC DIPOLE ANTENNA







patterns show better agreement with the calculated patterns than the H plane patterns do. This is because the unit pattern of the individual elements does not permit radiation in the direction broadside to the array.

The experimental data shown in Figures 34 and 35 reveals that the minimum obtainable beamwidth, for a particular choice of  $\tau$ , varies inversely with the parameter  $\tau$ , being about  $25^\circ$  narrower for  $\tau = 0.95$  than  $\tau = 0.81$  in the H plane. The calculated curves show that the minimum obtainable beamwidth for the three values of  $\tau$  used are nearly equal. This represents a rather serious failure since the conclusion to be drawn from these curves is that any value of  $\tau$  in the above range can be used to obtain the minimum beamwidth (or maximum gain) for this class of antennas as long as the proper apex angle is used. This, of course, is an incorrect conclusion as the measurements clearly show. It is believed that these erroneous results are due to the assumptions made in Section 3 relating the characteristics of the radiation wave to transmission line wave, and are not a basic deficiency of the leaky wave approach. While the assumptions made relating the two waves are valid as first order approximations, the actual relationship between the two waves is undoubtedly more complicated than that given by (45) and (47).

Since the transform used is valid only in the half plane to the front of the antenna, no information is given regarding the pattern shape in the rear quadrants. However, Figures (25) to (33)

Further, when better agreement with the calculated values than the 10% error is obtained, this is because the unit values of the individual elements have not been adjusted in the direction favorable to the error.

The experimental data shown in Figures 34 and 35 reveals that the minimum admissible homogeneity, for a particular choice of  $T$ , varies inversely with the parameter  $T$ , being about  $10^{-6}$  percent for  $T = 0.01$  and  $T = 0.02$  in the 10% error. The calculated curves show that the minimum admissible homogeneity for the

three values of  $T$  used are nearly equal. This represents a minimum value for the admissible homogeneity in the case of the three regions shown. The value of  $T$  in the above cases can be used for the minimum homogeneity (the minimum value) for this class of homogeneity. The proper way to use this is shown. This, in turn, is the

inverted condition on the homogeneity of the error. It is believed that these regions contain the minimum homogeneity

made in Section 2. The characteristic of the region have no transmission loss, and are not a wave velocity of the body wave approach. With the transmission loss being the same as the loss in the other regions, the same result is obtained between the two cases in which the wave velocity is the same.

given by (37) and (47).

Since the transmission loss is not only in the direction of the front of the pulse, no information is given regarding the pattern shown in the test results. However, since the test

show little energy is radiated to the rear of this antenna in the useable region of its parameter, so that this deficiency is not too important. If it is desired to predict the front to back ratio of these antennas, a more sophisticated theory will be required.











## 6. CONCLUSIONS

The object of this thesis was to provide an analytic solution of the far field radiation patterns for the log periodic class of antennas. Within certain limitations, this goal was obtained by considering them to behave as a leaky wave device.

It has been shown by DuHamel and Berry that each half structure radiates by itself with the result that the pattern of the complete antenna can be obtained using array theory if the pattern functions of each boom are known. This approach neglects the effects of mutual coupling, but it has been shown that these effects have only a small influence on the shapes of the radiation patterns.<sup>18</sup>

The radiation pattern for each half structure was obtained by hypothesising the existence of a radiating aperture at the front of the active region in the plane perpendicular to the half structure. The approximate aperture distribution was obtained by considering the radiation wave to behave as a leaky wave in the active region. The far field pattern was found from this aperture distribution using a transform technique. This procedure did not consider the half wave dipoles from which the radiation wave originated so that this pattern represented the group pattern. Since dipoles are isotropic in the H plane, this pattern gave the H plane pattern function for the half structure. The assumption was made that the unit pattern for the elements is that of a half wave dipole. The product of the group pattern and this unit pattern gave the E plane

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considering them as a body of work.

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pattern function. The pattern function of the complete structure was then obtained using array theory.

Assumptions were made relating the radiation wave to the transmission line wave so that the far field pattern functions of these antennas can be calculated if the characteristics of the transmission line wave are known.

These characteristics were obtained experimentally for a log periodic dipole antenna and used to calculate the far field patterns. It was found that each value of the geometric ratio,  $\tau$ , had a corresponding apex angle,  $\Theta$ , for which minimum beamwidth was obtained. It was found that this minimum beamwidth occurs approximately at the apex angle for which the relation

$$\beta_R^2 - \alpha_R^2 = k_0^2 \quad (52)$$

is satisfied. The beamwidth for this apex angle was found to be directly proportional to the magnitude of  $\alpha_R$ . It was observed that this property effectively limits the minimum beamwidth, or maximum gain, which can be obtained from this antenna since the magnitude of  $\alpha_R$  must be fairly large for frequency independent operation to take place.

The calculated far field patterns were found to show reasonably good agreement with the main beam of patterns obtained experimentally. This agreement demonstrates that the actual radiation patterns for these antennas can be approximated by the pattern obtained using aperture theory where the aperture field is computed



particular function. The various functions of the various elements  
was then obtained under other theory.

Assumptions were made relating the radiation wave to the  
transmission line wave as well as the far field pattern function of  
these antennas can be calculated if the characteristics of the  
transmission line wave are known.

These characteristics were obtained experimentally for a  
for periodic dipole antenna and used to calculate the far field  
pattern. It was found that each value of the impedance ratio,  $Z/Z_0$ ,  
had a corresponding open angle,  $\theta$ , and each antenna characteristic  
was obtained. It was found that this system described antenna  
characteristics at the open angle for which the relation

$$\frac{Z}{Z_0} = \frac{1}{\sin^2 \theta}$$

is satisfied. The characteristic for this open angle was found to be  
directly proportional to the resistance of  $Z_0$ . It was assumed  
that this property of the antenna is valid for all antennas, and  
assumed that, with the assumption that the antenna is a  
resistor of  $Z_0$  ohms, the far field pattern for the antenna is  
obtained in this manner.

The calculated far field pattern was found to agree with  
only good agreement with the data of antenna characteristics  
experimentally. This agreement demonstrates that the far field pattern  
pattern for these antennas can be approximated by the far field  
obtained under the assumption that the antenna is a resistor.

using leaky wave theory. While this approach gives only the approximate aperture distribution, the far field patterns thus calculated have been found to show exceedingly close agreement with the actual patterns if the magnitude of the attenuation constant is small.<sup>23,24</sup> For the log periodic class of antennas, reasonably good agreement was also obtained using this simplified aperture distribution, although the correlation was not as good as that obtained for structures with small attenuation constants. In the case of log periodic antennas, where the magnitude of the attenuation constant is fairly large, the influence of other factors on the shape of the radiation patterns becomes significant. This occurs because the effect of the leaky wave pole is inversely proportional to the magnitude of the attenuation constant. However, the generally good agreement obtained demonstrates that the leaky wave mode plays the dominant role in determining the far field radiation patterns of these antennas.

One main failure of the theory was noted. This was that the minimum obtainable beamwidths for various values of  $\tau$  were approximately equal and did not vary inversely with  $\tau$  as was experimentally observed. This failure of the theory was attributed to the assumptions made relating the characteristics of the transmission line and radiation waves.

It was also noted that if the front to back ratio of these antennas is required, a more sophisticated theory would have to be used since the transform is valid only for the half plane in





front of the antenna.

The conclusion was made that the radiating patterns for the log periodic antenna can be obtained analytically from the properties of the transmission line field by considering this antenna to behave as a leaky wave device.









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## APPENDIX A: THE SADDLE-POINT METHOD OF APPROXIMATE INTEGRATION

The saddle-point method of approximate integration allows integrals of the form given by (33) in Section 3 to be evaluated for large arguments using an asymptotic expansion. The following analysis of this method is a condensed version of that given in Reference (26) and is put in a convenient form to be used in Section 3.

Equation (33), the integral to be solved using this approximate method of integration, is

$$E_y(x, z) = \frac{E_a \delta_z}{j \pi} \int_C \frac{e^{-\delta'_z(x-a) - \delta'_z z}}{\delta_z^2 - \delta'^2_z} d\delta'_z \quad (33)$$

where the path of integration lies along the imaginary axis

$-j\infty \leq \delta'_z \leq j\infty$  in the complex  $\delta'_z$  plane. This integral could be solved by closing the contour at infinity and using Cauchy's residue theorem. However, the branch points given by  $\delta'_z = j(k_o^2 + \delta_z^2)^{1/2} = 0$  which occur at  $\delta'_z = \pm j k_o$  must not be enclosed by the contour.

Therefore, the contour must come back in from infinity on one side of the branch line, encircle the branch point,  $\delta'_z = j k_o$ , and recede out to infinity along the opposite side of the branch line. This contour is shown in Figure 37, assuming  $k_o$  has a small loss term (i.e.  $k_o$  is complex), for an arbitrary position of the branch line. Although the orientation of the branch line is arbitrary, it must be chosen carefully if the integral is to vanish along the semi-circle closing the contour at infinity and so a rigorous solution based on this approach may be nearly impossible to obtain.

# APPENDIX 2: THE STATION-TO-STATION ON OBSERVATION INFORMATION

The station-to-station method of observation information is

illustrated of the form given in Table 2. In the following  
large response, some are indicated in parentheses. The following  
analysis of this method is a condensed version of that given in  
reference (30) and is not to be considered; but it is not to

be used as

illustration (31). The formula for the station-to-station

approximation method of information is

$$(32) \quad E_2(x) = \frac{E_1(x)}{1 - \frac{E_1(x)}{E_2(x)}} = \frac{E_1(x)}{1 - \frac{E_1(x)}{E_2(x)}}$$

where the ratio of information from the two methods is

$\frac{E_2(x)}{E_1(x)} \geq \frac{E_1(x)}{E_2(x)}$  in the complex  $E_1(x)$  case. The following

be used as a basis for the method of station-to-station

analysis. However, the method is not to be used as a basis

for the method of station-to-station information.

Therefore, the method is not to be used as a basis

of the method of station-to-station information.

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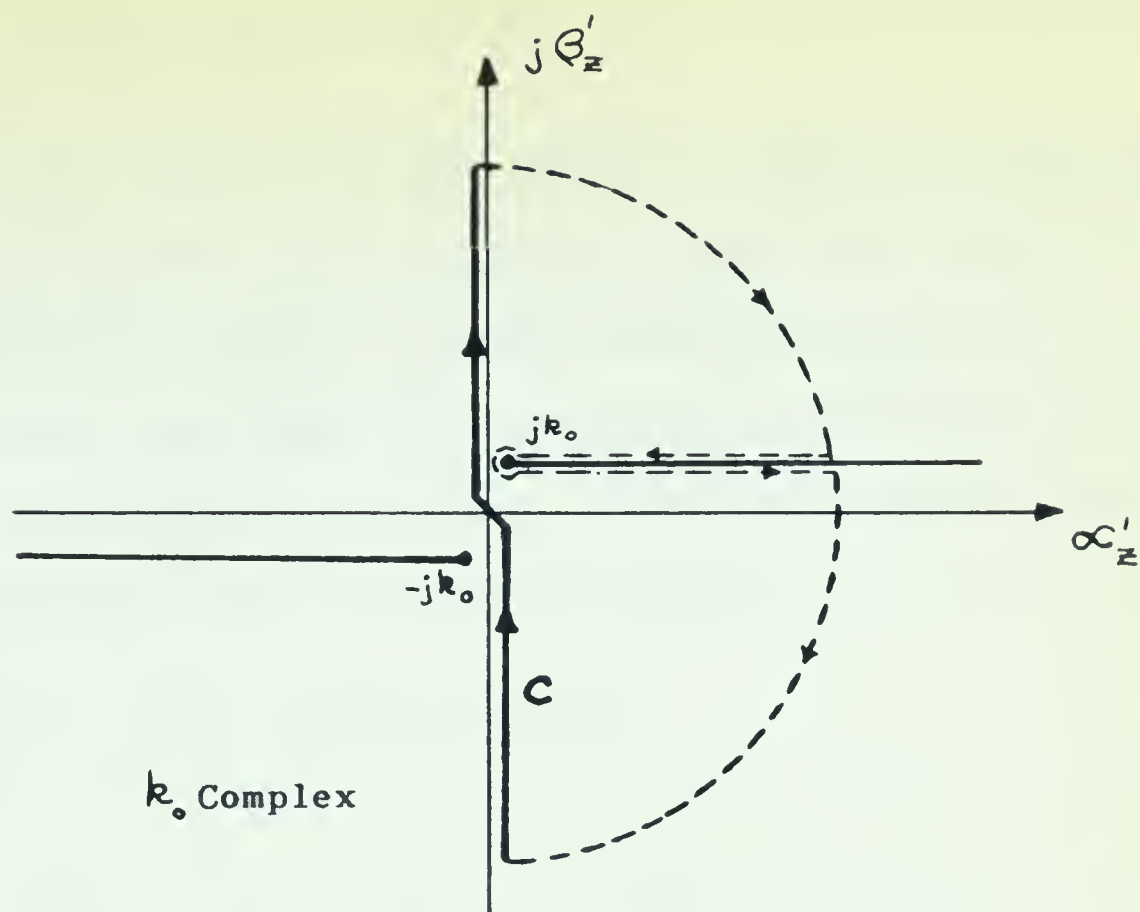


FIGURE 37

CONTOUR C FOR EQUATION 33

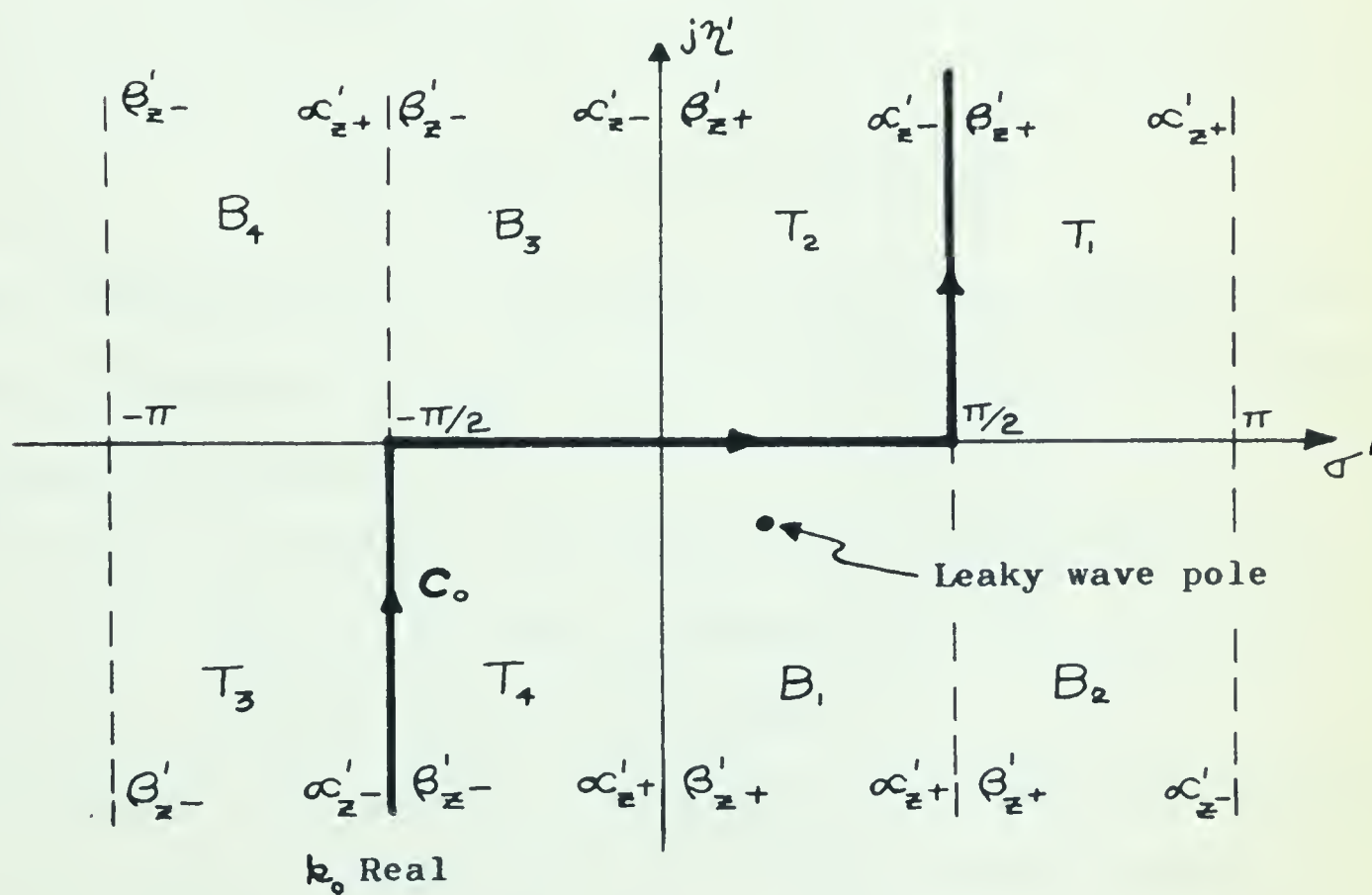


FIGURE 38

MAPPING OF TWO SHEETS OF  $\gamma'_z$  PLANE  
ON TO A STRIP OF THE  $\phi'$  PLANE





For large values of  $h = [(x-a)^2 + z^2]^{1/2}$ , integrals of this type may be solved asymptotically using the saddle-point method (or method of steepest descent) of approximate integration.

Since only one square root is involved (with two corresponding Riemann surfaces) it is possible to eliminate the multivaluedness of the integrand in (33) using the transformation

$$\gamma'_z = j k_0 \sin \phi' \quad (34)$$

where

$$\gamma'_z = \alpha'_z + j\beta'_z \quad \phi' = \sigma' + j\eta' \quad (35)$$

Therefore

$$d\gamma'_z = j k_0 \cos \phi' d\phi' \quad (53)$$

and

$$(\gamma'^2_z + k_0^2)^{1/2} = + k_0 \cos \phi' \quad (54)$$

since  $\cos^2 \phi' + \sin^2 \phi' = 1$ . Equation 34 represents a mapping of the complex  $\gamma'_z$  plane into a strip on the complex  $\phi'$  plane. The two sheets of the Riemann surface map into a connected strip of width  $2\pi$  along the  $\sigma'$  axis. From (34) and (35) the result

$$\alpha'_z = -k_0 \cos \sigma' \sinh \eta' \quad (55)$$

$$\beta'_z = k_0 \sin \sigma' \cosh \eta' \quad (56)$$

is obtained. This mapping is illustrated in Figure 38. The quadrants in which  $\beta'_z$  and  $\alpha'_z$  are positive or negative are found from (55) and (56) and are specified in the Figure. The proper (top) Riemann surface is the one for which the integral (33) converges on the

For these values of  $\lambda = \left[ \frac{1}{2}(\alpha - \alpha_0) + \pi \right]^{1/2}$ , solutions of form (18)

may be solved analytically using the usual-point method for

method of steepest descent, or approximately (19)

Since only one square root is involved (18) and (19) are

the same (20) (21) it is possible to eliminate the coefficients

of the integrand in (21) using the transformation

$$(18) \quad \delta'_z = \lambda \sin \phi$$

where

$$(19) \quad \delta'_z = \alpha_0 \cos \phi \quad \delta'_z = \lambda \sin \phi$$

Therefore

$$(20) \quad \delta'_z = \lambda \cos \phi \quad \delta'_z = \lambda \sin \phi$$

and

$$(21) \quad (\delta'_z + \lambda)^2 = \lambda^2 \cos^2 \phi$$

Since  $\delta'_z = \lambda \cos \phi$ , (21) becomes a square of the

complex  $\delta'_z$  into a circle in the complex  $\delta'_z$  plane. The two

branches of the Riemann surface are then a connected sheet of which

It is along the  $\delta'_z$  axis. From (21) and (22) the results

$$(22) \quad \delta'_z = \lambda \cos \phi \quad \delta'_z = \lambda \sin \phi$$

$$(23) \quad \delta'_z = \lambda \cos \phi \quad \delta'_z = \lambda \sin \phi$$

is obtained. This mapping is illustrated in Figure 2B. The mapping

in which  $\delta'_z$  and  $\delta'_z$  are positive or negative and from (21)

and (22) and are specified in the figure. The branch (21) branch

surface is the one for which the integral (20) converges on the



infinite semicircle in the  $\gamma'_z$  plane and is given by  $\text{Im}(j\gamma'_z)$   
 $= \text{Im}(\gamma'^2_z + k_o^2)^{1/2} < 0$  (For the improper, or bottom, Riemann surface,  
 given by  $\text{Im}(j\gamma'_z) > 0$ , the integral does not converge at  
 infinity.) Therefore, from (54) the corresponding proper part of the  
 $\phi'$  plane is determined by the condition that  $\sin \sigma' \sinh \eta' > 0$ . The  
 The four quadrants of the top and bottom sheets of the  $\gamma'_z$  plane map  
 into the regions designated  $T_i$  and  $B_i$  ( $i = 1, 2, 3, 4$ ) respectively,  
 in the Figure. The positive sign in (54) was chosen so that the  
 quadrants  $T_1, T_2$  and  $T_3, T_4$  would map into adjacent strips. There  
 are no branch lines in the  $\phi'$  plane since (34) maps both sheets of  
 the Riemann surface into a connected strip.

As a further convenience, the rectangular coordinates,  $x, z$ ,  
 are changed to the cylindrical coordinates,  $r, \theta$ , where  $\theta$  is  
 measured from the  $x$  axis.

$$x - a = r \cos \theta \quad (57)$$

$$z = r \sin \theta \quad (58)$$

Using Equations (34), (53), (54), (57) and (58) the integral  
 (33) becomes

$$E_y(r, \theta) = \frac{E_o \gamma_z k_o}{\pi} \int_{C_o} \frac{e^{-jk_o r \cos(\phi' - \theta)} \cos \phi'}{k_o^2 \sin^2 \phi' + \gamma_z^2} d\phi' \quad (36)$$

where  $C_o$ , the contour of integration is shown in Figure 38.

This integral may be put in the form

$$I = \int_{C_o} R(\phi) e^{f(\phi)} d\phi' \quad (59)$$

where

$$R(\phi) = \frac{E_o \gamma_z k_o}{\pi} \cdot \frac{\cos \phi'}{k_o^2 \sin^2 \phi' + \gamma_z^2} \quad (60)$$

Let  $\mathcal{H}$  be a Hilbert space and let  $\mathcal{H}^*$  be its dual space.

Let  $\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$  be a direct sum decomposition of  $\mathcal{H}$ .

Let  $\mathcal{H}^* = \mathcal{H}_1^* \oplus \mathcal{H}_2^*$  be a direct sum decomposition of  $\mathcal{H}^*$ .

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Let  $\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$  be a direct sum decomposition of  $\mathcal{H}$ .

$$\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$$

101

$$\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$$

102

Let  $\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$  be a direct sum decomposition of  $\mathcal{H}$ .

Let  $\mathcal{H}^* = \mathcal{H}_1^* \oplus \mathcal{H}_2^*$  be a direct sum decomposition of  $\mathcal{H}^*$ .

$$\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$$

103

Let  $\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$  be a direct sum decomposition of  $\mathcal{H}$ .

Let  $\mathcal{H}^* = \mathcal{H}_1^* \oplus \mathcal{H}_2^*$  be a direct sum decomposition of  $\mathcal{H}^*$ .

$$\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$$

104

$$\mathcal{H} = \mathcal{H}_1 \oplus \mathcal{H}_2$$

105

and

$$f(\phi') = -jk_0 r \cos(\phi' - \theta) \quad (61)$$

Since there are no branch lines in the  $\phi'$  plane the contour of integration may be deformed into any other convenient contour without changing the integral (59) provided no poles of  $R(\phi')$  are swept across.

In the integrand for (36)

$$f(\phi') = -jk_0 r \cos(\phi' - \theta) \quad (61)$$

and the derivative of  $f(\phi')$  with respect to  $\phi'$ ,

$$\frac{df}{d\phi'} = jk_0 r \sin(\phi' - \theta) \quad (62)$$

is equal to zero at  $\phi' = \theta$  (or  $\sigma' = \theta$ ,  $\eta' = 0$ ). Since a complex function cannot have a maximum or a minimum, these stationary points are saddle points. In the vicinity of the saddle point, a Taylor's expansion gives

$$\begin{aligned} f(\phi') &= f(\theta) + \left. \frac{df}{d\phi'} \right|_{\theta} (\phi' - \theta) + \frac{1}{2} \left. \frac{d^2 f}{d\phi'^2} \right|_{\theta} (\phi' - \theta)^2 + \dots \\ f(\phi') &= -jk_0 r + j \frac{k_0 r}{2} (\phi' - \theta)^2 + \dots \end{aligned} \quad (63)$$

since the first derivative vanishes and the second derivative equals  $jk_0 r$  at  $\theta$ . Let  $\rho$  and  $S$  be the radial and angular coordinates with their origin at  $\theta$  in the  $\phi$  plane. Therefore,

$$(\phi' - \theta) = (\sigma' - \theta + j\eta') = \rho e^{jS} \quad (64)$$

Let

$$f_1 + jf_2 = f(\phi') - f(\theta) \quad (65)$$





Therefore, in the vicinity of the saddle point at  $\phi' = \theta$

$$f_1 + j f_2 = \left[ -j k_0 r + \frac{j k_0 r}{2} (\phi' - \theta)^2 + \dots \right] - [-j k_0 r]$$

$$f_1 + j f_2 = j \frac{k_0 r}{2} \rho^2 e^{j 2s} + \dots$$

and in the limiting case as  $\rho \rightarrow 0$

$$f_1 = -\frac{k_0 r}{2} \rho^2 \sin 2s \quad (66)$$

$$f_2 = \frac{k_0 r}{2} \rho^2 \cos 2s \quad (67)$$

The contours  $f_1 = \text{constant}$  are orthogonal to the contours

$f_2 = \text{constant}$ . These contours are sketched in Figure 39. The maximum rate of change of  $f_1$  with  $\rho$  occurs for  $s = \frac{\pi}{4}$  and  $\frac{\pi}{4} + \pi$  and also for  $s = \frac{3\pi}{4}$  and  $-\frac{\pi}{4}$ . Along these paths of steepest descent  $f_2 = 0$ .

In view of the properties of the functions  $f_1$  and  $f_2$ , it is clear that it is desirable to deform the contour  $C_0$  in (59) into a steepest descent contour (i.e. along  $f_2 = 0$ ) passing through the saddle point,  $\phi' = \theta$ , at an angle  $s = \frac{\pi}{4}$  since, along this path  $f_1$  is negative and hence  $e^{f(\phi)}$  will decrease rapidly away from the saddle point. Therefore, the major contribution to the integral comes from a small region  $0 < \rho < \rho_1$ , along the steepest descent contour. For a general value of  $\phi'$ , (61) and (65) give

$$f_1 = -k_0 r \sin(\phi' - \theta) \sinh \eta' \quad (68)$$

Therefore, in the vicinity of the saddle point at  $\tilde{q} = 0$

$$V(\tilde{q}) = \left[ -\frac{1}{2}(\tilde{q} - \tilde{q}_0)^2 + \frac{1}{4}(\tilde{q} - \tilde{q}_0)^4 + \dots \right] = \tilde{q}_1 + \tilde{q}_2$$

$$\dots = \frac{1}{2} \tilde{q}^2 + \frac{1}{4} \tilde{q}^4 + \dots = \tilde{q}_1 + \tilde{q}_2$$

and in the limit case as  $\tilde{q} \rightarrow 0$

(66)

$$\tilde{q}_1 = -\frac{1}{2} \tilde{q}^2 + \frac{1}{4} \tilde{q}^4 + \dots = \tilde{q}_2$$

(67)

$$\tilde{q}_1 = \frac{1}{2} \tilde{q}^2 + \frac{1}{4} \tilde{q}^4 + \dots = \tilde{q}_2$$

The constant  $\tilde{q}_1 = \text{constant}$  is orthogonal to the contour  $\tilde{q}_2 = \text{constant}$ . These contours are shown in Figure 1. The minimum value of  $\tilde{q}_1$  is  $\tilde{q}_1 = 0$  when  $\tilde{q} = 0$  and  $\tilde{q}_2 = 0$  and also for  $\tilde{q} = \pm \sqrt{2}$  and  $\tilde{q}_2 = 0$ . Along these lines of constant energy  $\tilde{q}_1 = 0$ .

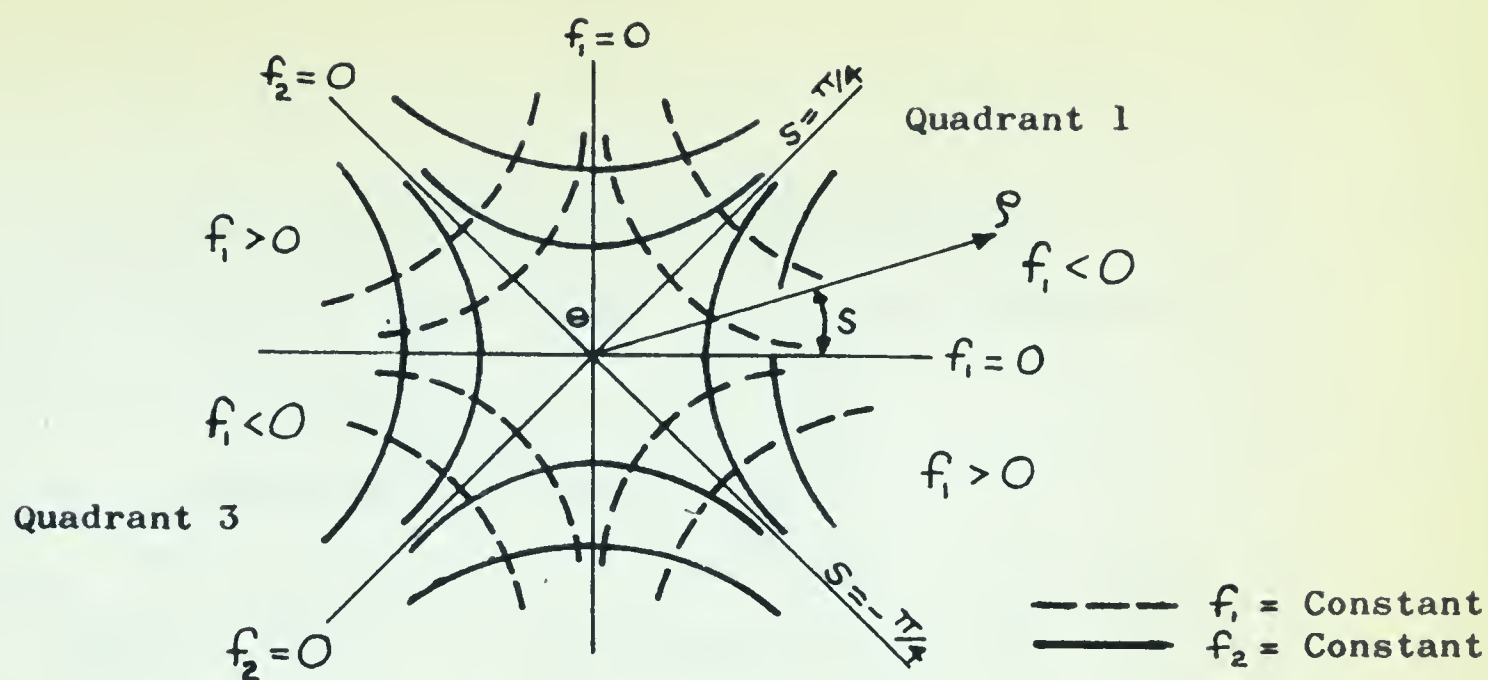
In view of the properties of the functions  $\tilde{q}_1$  and  $\tilde{q}_2$ , it is clear that in the vicinity of the minimum  $\tilde{q}_1$  in (66) is a second degree function (i.e. when  $\tilde{q}_1 = 0$ ) and, although the saddle point,  $\tilde{q}_1 = 0$ , is not a minimum, along this path  $\tilde{q}_1$  is stationary and hence  $\tilde{q}_2$  is stationary rapidly away from the saddle point. Therefore, the wave function in the integral comes from a small region  $\tilde{q}_1 \approx 0$ , where the second degree constant. For a small value of  $\tilde{q}_1$ , (66) and

(67) give

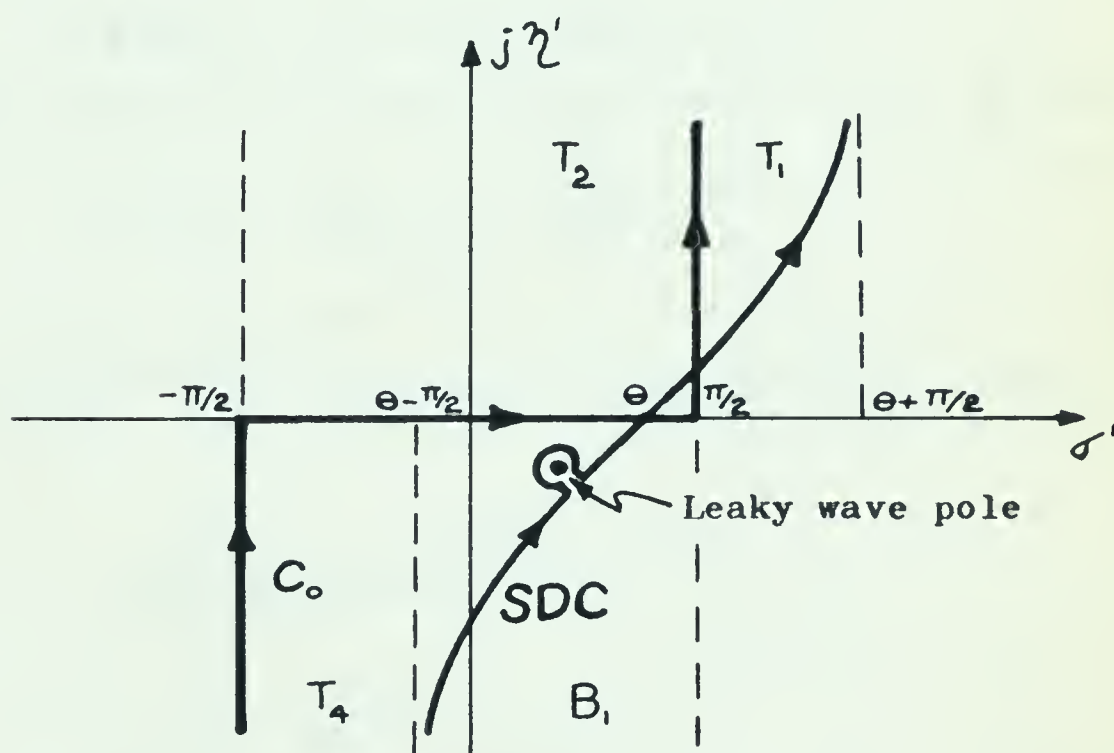
(68)

$$\tilde{q}_1 = -\frac{1}{2} \tilde{q}^2 + \frac{1}{4} \tilde{q}^4 + \dots = \tilde{q}_2$$





**FIGURE 39**  
**CONTOURS OF CONSTANT  $f_1$  AND  $f_2$**



**FIGURE 40**  
**DEFORMATION OF CONTOUR  $C_0$  INTO**  
**STEEPEST DESCENT CONTOUR  $SDC$**



$$f_2 = k_0 \rho [1 - \cos(\sigma' - \theta) \cosh \eta'] \quad (69)$$

The steepest descent is given by  $f_2 = 0$  and hence the steepest descent contour is given by

$$\cos(\sigma' - \theta) \cosh \eta' = 1 \quad (70)$$

This contour is illustrated as the contour SDC in Figure 40. Any poles of  $R(\phi')$  encountered while deforming  $C_0$  into SDC must not be swept over. The contour passes through the saddle point at an angle  $\frac{\pi}{4}$  in accordance with (67) which is valid near the saddle point.

In the vicinity of the saddle point, (63) and (64) give

$$f(\phi') \approx -jk_0 \rho + j \frac{k_0 \rho}{2} \rho^2 e^{2js}$$

which becomes, along the steepest descent contour  $S = \frac{\pi}{4}$  and  $\frac{\pi}{4} + \pi$

$$f(\phi') \approx -jk_0 \rho - \frac{k_0 \rho}{2} \rho^2 \quad (71)$$

Along this path  $\phi' = \theta + \rho e^{j\pi/4}$  in the first quadrant ( $S = \frac{\pi}{4}$ ) and  $\phi' = \theta + \rho e^{j(\pi + \pi/4)}$  in the third quadrant ( $S = \pi + \frac{\pi}{4}$ ).

Therefore,

$$\phi' - \theta = \rho e^{j\pi/4} \quad (72)$$

and

$$d\phi' = e^{j\pi/4} d\rho \quad (73)$$

in the first quadrant and

$$\phi' - \theta = -\rho e^{j\pi/4} \quad (74)$$





and

$$d\phi' = -e^{j\pi/4} d\rho \quad (75)$$

in the third quadrant. Therefore, the integral, (59),

$$I = \int_{C_0} R(\phi) e^{f(\phi)} d\phi \quad (59)$$

becomes

$$I = \int_{SDC} R(\phi) e^{f(\phi)} d\phi - j2\pi \sum_p F(\phi'_p) e^{f(\phi'_p)} \quad (76)$$

where  $F(\phi'_p)$  is the residue of  $R(\phi)$  at a pole,  $\phi'_p$ , which the contour encounters when it is deformed from  $C_0$  to SDC. The series arises from integrating about the circles surrounding these poles.

If  $R(\phi)$  does not have a pole in the vicinity of the saddle point, the integral along the steepest descent contour can be evaluated by expanding  $R(\phi)$  in a Taylor series about the saddle point,  $\phi' = \theta$ . Thus

$$R(\phi) = R(\theta) + \sum_{n=1}^{\infty} R^n(\theta) \frac{(\phi' - \theta)^n}{n!} \quad (77)$$

where

$$R^n(\theta) = \left. \frac{d^n R(\phi')}{d\phi'^n} \right|_{\phi' = \theta}$$

The steepest descent contour integral may be solved by substituting (71), (72), (73), (74), (75) and (77) in the steepest descent contour integral part of (76). Making the above substitutions gives





$$I_{SDC} = \sum_{n=0}^{\infty} \left\{ \int_{\rho_1}^0 \left[ R^n(\theta) \frac{(-\rho e^{j\pi/4})^n}{n!} \right] \left[ e^{-jk_0 \rho} - \frac{k_0 \rho}{2} \rho^2 \right] \left[ -e^{j\pi/4} d\rho \right] \right. \\ \left. + \int_0^{\rho_1} \left[ R^n(\theta) \frac{(\rho e^{j\pi/4})^n}{n!} \right] \left[ e^{-jk_0 \rho} - \frac{k_0 \rho}{2} \rho^2 \right] \left[ e^{j\pi/4} d\rho \right] \right\} \\ I_{SDC} = e^{-j(k_0 \rho - \pi/4)} \sum_{n=0}^{\infty} \frac{R^n(\theta)}{n!} e^{j\frac{n\pi}{4}} [1 + (-1)^n] \int_0^{\rho_1} \rho^n e^{-\frac{k_0 \rho}{2} \rho^2} d\rho \quad (78)$$

The integral over  $\rho$  does not change much if  $\rho_1$  is extended to infinity. The result is

$$\int_0^{\infty} \rho^n e^{-\frac{k_0 \rho}{2} \rho^2} d\rho = \frac{\Gamma(\frac{n+1}{2})}{2(\frac{k_0 \rho}{2})^{\frac{n+1}{2}}} \quad n > -1 \quad (79)$$

where  $\Gamma$  is the gamma function. Therefore,

$$I_{SDC} = e^{-j(k_0 \rho - \pi/4)} \sum_{n=0,2,\dots} \frac{R^n(\theta)}{n!} e^{j\frac{n\pi}{4}} \left(\frac{2}{k_0 \rho}\right)^{\frac{n+1}{2}} \Gamma\left(\frac{n+1}{2}\right) \quad (80)$$

and the total solution of (76), and hence (59), is

$$I = e^{-j(k_0 \rho - \pi/4)} \sum_{n=0,2,\dots} \frac{R^n(\theta)}{n!} e^{j\frac{n\pi}{4}} \left(\frac{2}{k_0 \rho}\right)^{\frac{n+1}{2}} \Gamma\left(\frac{n+1}{2}\right)$$

$$-j2\pi \sum_{\rho} F(\phi'_\rho) e^{-jk_0 \rho \cos \alpha (\phi'_\rho - \theta)} \quad (81)$$

where  $F(\phi'_\rho)$  is the residue of  $R(\phi)$  at the pole,  $\phi'_\rho$ , encountered when the contour is deformed from  $C_0$  to SDC and

$$R^n(\theta) = \left. \frac{d^n R(\phi)}{d\phi^n} \right|_{\phi'=\theta}$$

The saddle point method of approximate integration is valid only if  $\frac{k_0 \rho}{2}$  is large for small values of  $\rho$ . This means that  $k_0 \rho$  must be very large.

For the far field pattern  $\rho$  approaches infinity and (81)



becomes

$$I = \left(\frac{2\pi}{k_0 \rho}\right)^{1/2} e^{-j(k_0 \rho - \frac{\pi}{4})} R(\theta) \quad (82)$$

since  $\Gamma(1/2) = \sqrt{\pi}$ . For large  $\rho$  the contribution of the residues at the leaky wave poles of  $R(\phi)$  are neglected because they damp out exponentially in any radial direction from the origin within their wedges of contribution.<sup>29</sup>

Therefore, the solution of  $E_y(\rho, \theta)$  in (36) is, using (59),

$$E_y(\rho, \theta) = E_a \delta_z k_0 \left(\frac{2}{\pi k_0 \rho}\right)^{1/2} e^{-j(k_0 \rho - \pi/4)} f_1(\theta) \quad (37)$$

where the pattern function  $f_1(\theta)$  is given by

$$f_1(\theta) = \frac{\cos \theta}{k_0^2 \sin^2 \theta + \delta_z^2} \quad (38)$$

where  $\rho$  and  $\theta$  are the components of a cylindrical coordinate system in the  $x, z$  plane and  $\theta$  is measured from the  $x$  axis.







APPENDIX B





FIGURE 41

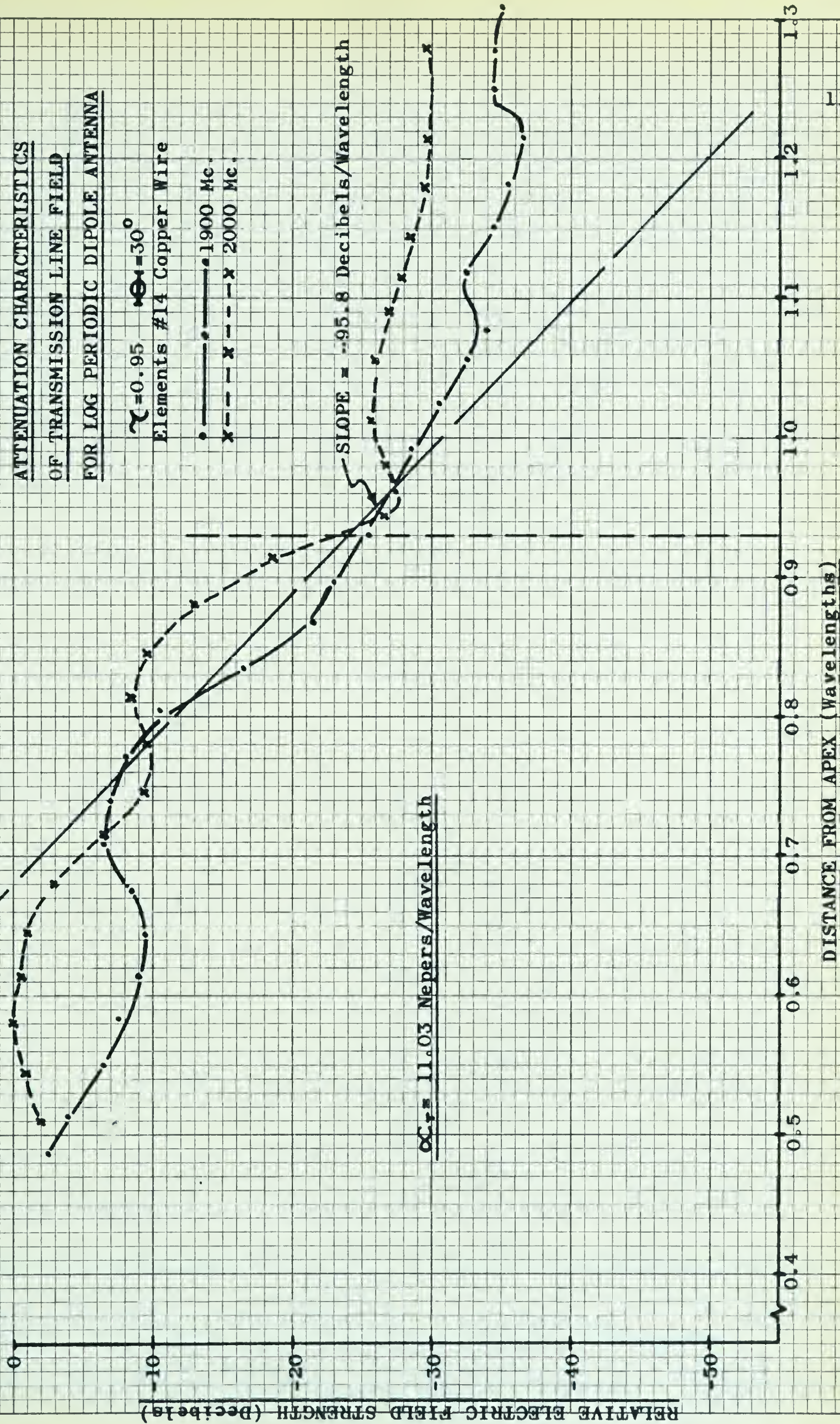
ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.95$   $\Theta = 30^\circ$   
Elements #14 Copper Wire

—•— 1900 Mc.  
-x- 2000 Mc.

SLOPE = -95.8 Decibels/Wavelength

$\alpha \gamma = 11.03$  Nepers/Wavelength



DISTANCE FROM APEX (Wavelengths)





FIGURE 42

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.95$   $\theta = 30^\circ$   
Elements #14 Copper Wire

---x--- 1900 Mc.  
—•— 2000 Mc.

PHASE SHIFT OF ELECTRIC FIELD (Degrees)

$\beta_r = 20.85$  Radians/Wavelength

SLOPE = -1196 Degrees/Wavelength

DISTANCE FROM APEX (Wavelengths)

113

0.4

0.5

0.6

0.7

0.8

0.9

1.0

1.1

1.2

1.3

0

100

200

300

400

500

600







FIGURE 43

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD

FOR LOG PERIODIC DIPOLE ANTENNA

$$\gamma = 0.95 \quad \theta = 25^\circ$$

Elements #14 Copper Wire

- 1900 Mc.
- x— 2000 Mc.
- +--- 2100 Mc.

RELATIVE ELECTRIC FIELD STRENGTH (Decibels)

$$\propto r^{-10.21} \text{ Nepers/Wavelength}$$

SLOPE = -88.7 Decibels/Wavelength

DISTANCE FROM APEX (Wavelengths)







FIGURE 44

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

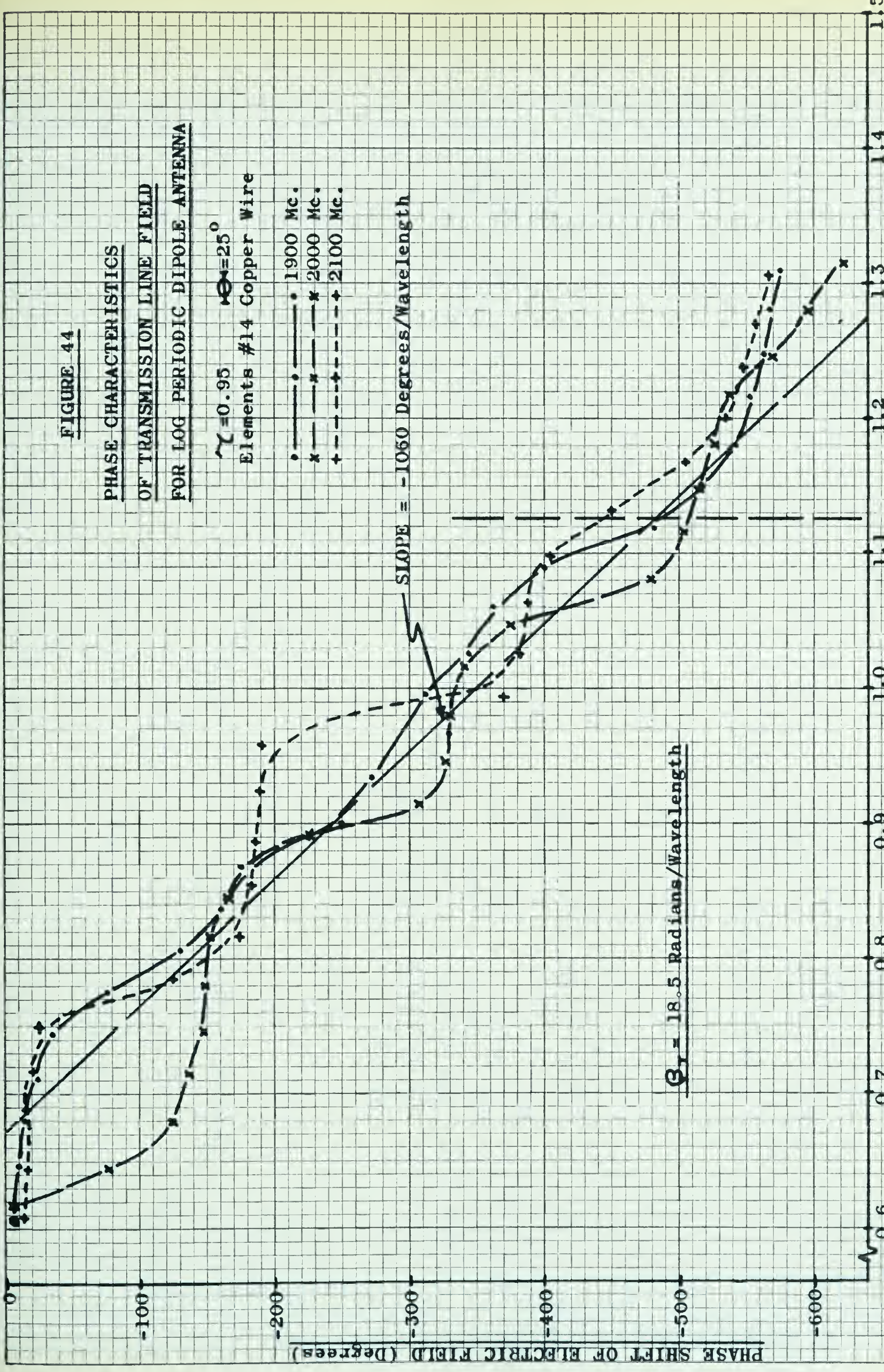
$\gamma = 0.95$   $\theta = 25^\circ$   
Elements #14 Copper Wire

- —•—• 1900 Mc.
- x —x—x 2000 Mc.
- + —+—+ 2100 Mc.

SLOPE = -1060 Degrees/Wavelength

$\beta_T = 18.5$  Radians/Wavelength

DISTANCE FROM APEX (Wavelengths)









SLOPE = -66.4 Decibels/Wavelength

FIGURE 45

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$$\gamma = 0.95 \quad \theta = 20^\circ$$

Elements #14 Copper Wire

• ——— 1900 Mc.  
x ——— 2100 Mc.

$$\alpha_T = 7.65 \text{ Nepers/Wavelength}$$

RELATIVE ELECTRIC FIELD STRENGTH (Decibels)

DISTANCE FROM APEX (Wavelengths)







FIGURE 46

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.95$   $\theta = 20^\circ$   
Elements #14 Copper Wire  
 - - - - - 1900 Mc.  
 - - - - - 2100 Mc.

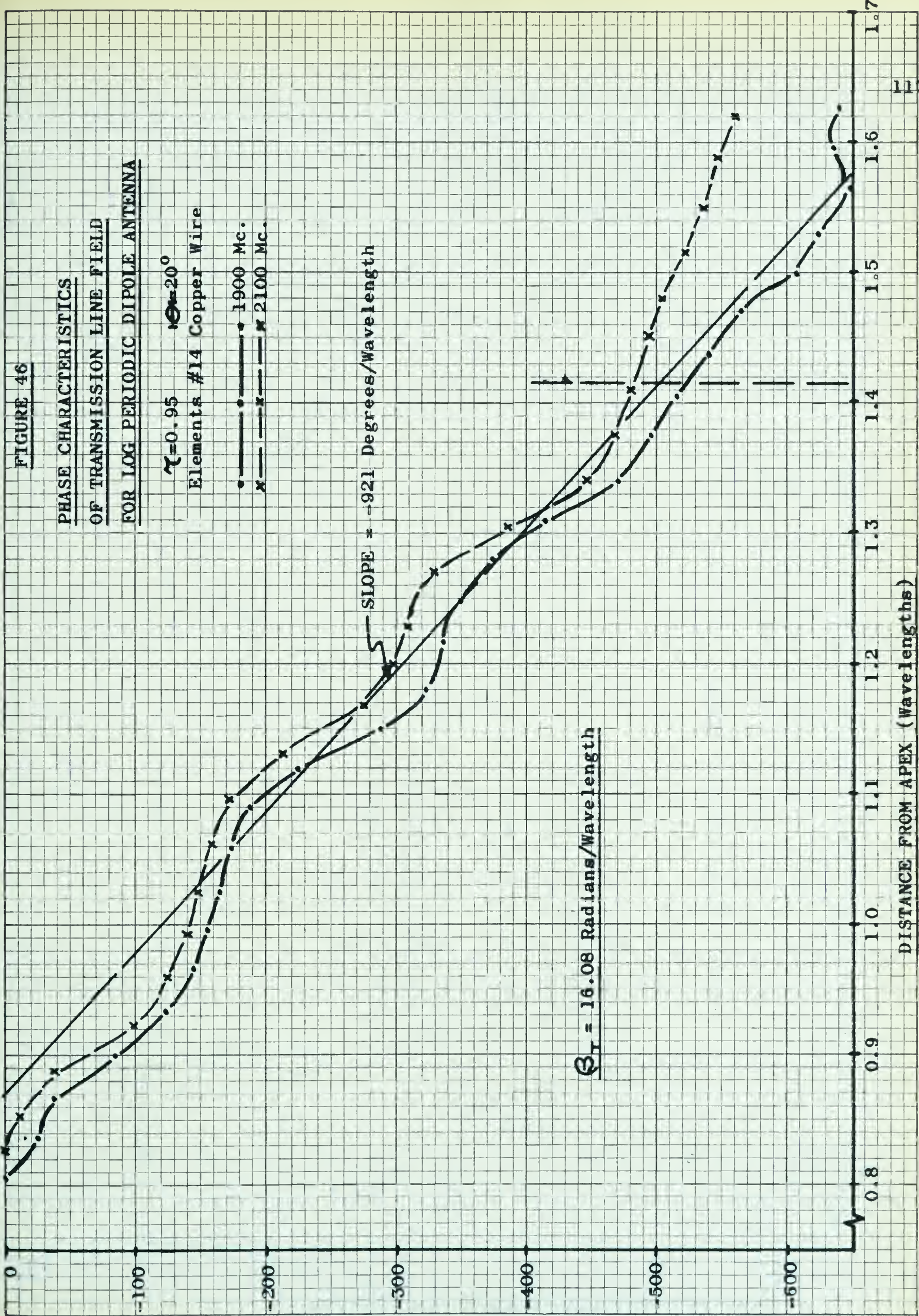
PHASE SHIFT OF ELECTRIC FIELD (Degrees)

$\beta_T = 16.08$  Radians/Wavelength

SLOPE = -921 Degrees/Wavelength

DISTANCE FROM APEX (Wavelengths)

117









SLOPE = -69.4 Decibels/Wavelength

FIGURE 47

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.95$   $\theta = 15^\circ$

Elements #14 Copper Wire

- 2500 Mc.
- x- 2750 Mc.
- +---+ 3000 Mc.

$\alpha_{\gamma} = 7.99$  Nepers/Wavelength

DISTANCE FROM APEX (Wavelengths)







FIGURE 48

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\tau = 0.95$   $\theta = 15^\circ$   
Elements #14 Copper Wire

- 2500 Mc.
- x— 2750 Mc.
- +— 3000 Mc.

SLOPE = -890 Degrees/Wavelength

$\beta_T = 15.52$  Radians/Wavelength

DISTANCE FROM APEX (Wavelengths)

PHASE SHIFT OF ELECTRIC FIELD (Degrees)







FIGURE 49

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.95$   $\theta = 10^\circ$   
Elements #14 Copper Wire

- 3300 Mc.
- x- 3400 Mc.
- +- 3500 Mc.

SLOPE = -49.1 Decibels/Wavelength

$\alpha_{\gamma} = 5.66$  Nepers/Wavelength

RELATIVE ELECTRIC FIELD STRENGTH (Decibels)

DISTANCE FROM APEX (Wavelengths)







FIGURE 50

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.95$   $\theta = 10^\circ$   
Elements #14 Copper Wire

- 3300 Mc.
- x— 3400 Mc.
- +— 3500 Mc.

PHASE SHIFT OF ELECTRIC FIELD (Degrees)

$\beta_r = 11.31$  Radians/Wavelength

SLOPE = -647 Degrees/Wavelength

DISTANCE FROM APEX (Wavelengths)







SLOPE = -91.5 Decibels/Wavelength

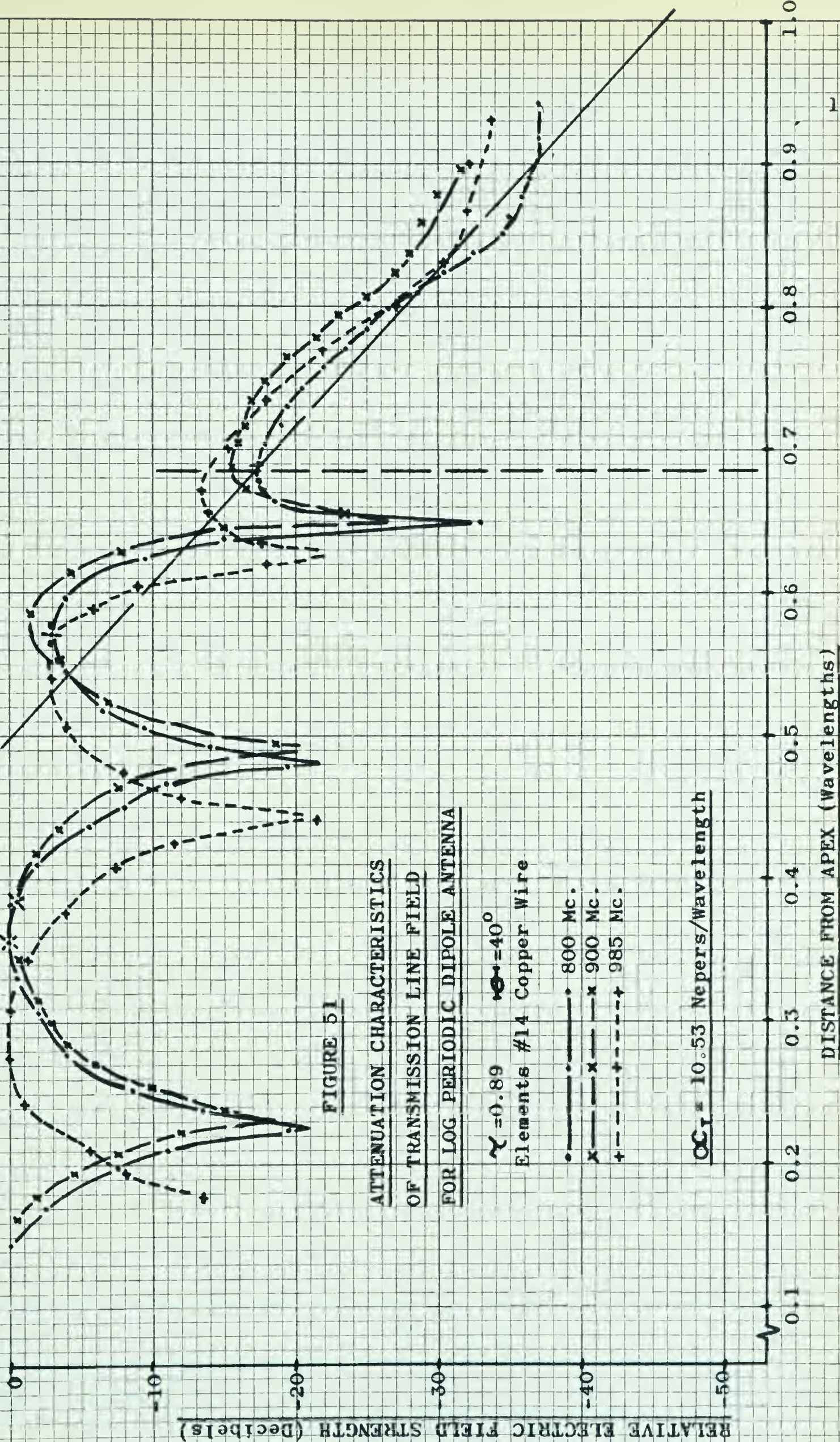


FIGURE 51

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.89$   $\theta = 40^\circ$   
Elements #14 Copper Wire

• — 800 Mc.  
x — 900 Mc.  
+ - - 985 Mc.

$\alpha_T = 10.53$  Nepers/Wavelength

DISTANCE FROM APEX (Wavelengths)







FIGURE 52

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC ANTENNA

$\gamma = 0.89$   $\theta = 40^\circ$   
Elements #14 Copper Wire

- 800 Mc.
- x— 900 Mc.
- - - + - 985 Mc.

SLOPE = -925 Degrees/Wavelength

$\theta_T = 16.16 \text{ Radians/Wavelength}$

PHASE SHIFT OF ELECTRIC FIELD (Degrees)

DISTANCE FROM APEX (Wavelengths)







SLOPE = -88.5 Decibels/Wavelength

RELATIVE ELECTRIC FIELD STRENGTH (Decibels)

FIGURE 53

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.89$   $\theta = 35^\circ$   
Elements #14 Copper Wire

800 Mc.  
1900 Mc.  
2100 Mc.

$\alpha C_T = 10.20$  Nepers/Wavelength

DISTANCE FROM APEX (Wavelengths)







FIGURE 54

PHASE CHARACTERISTICS

OF TRANSMISSION LINE FIELD

FOR LOG PERIODIC DIPOLE ANTENNA

$$\gamma = 0.89 \quad \Theta = 35^\circ$$

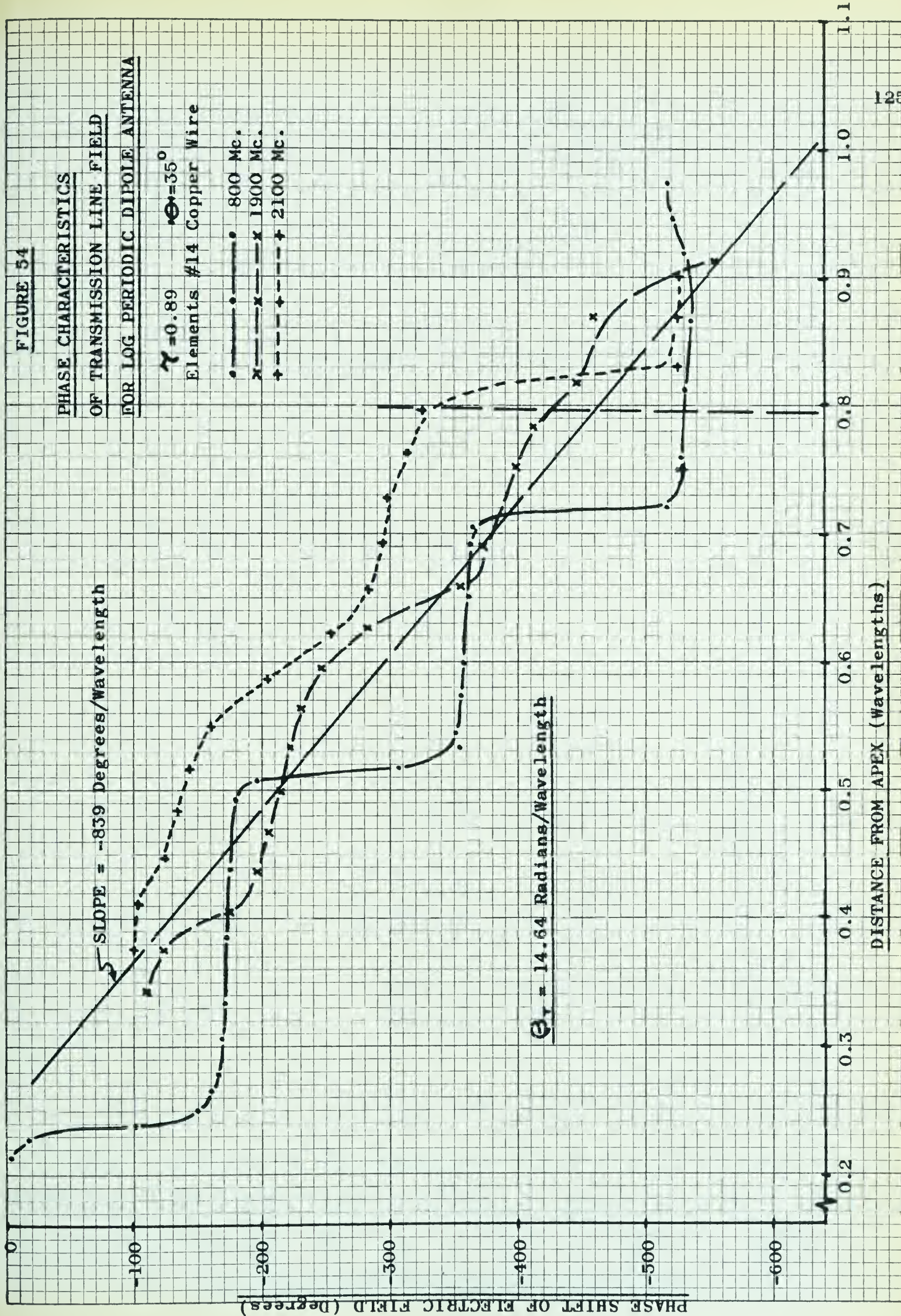
Elements #14 Copper Wire

- 800 Mc.
- x— 1900 Mc.
- +— 2100 Mc.

SLOPE = -839 Degrees/Wavelength

$$\Theta_T = 14.64 \text{ Radians/Wavelength}$$

DISTANCE FROM APEX (Wavelengths)









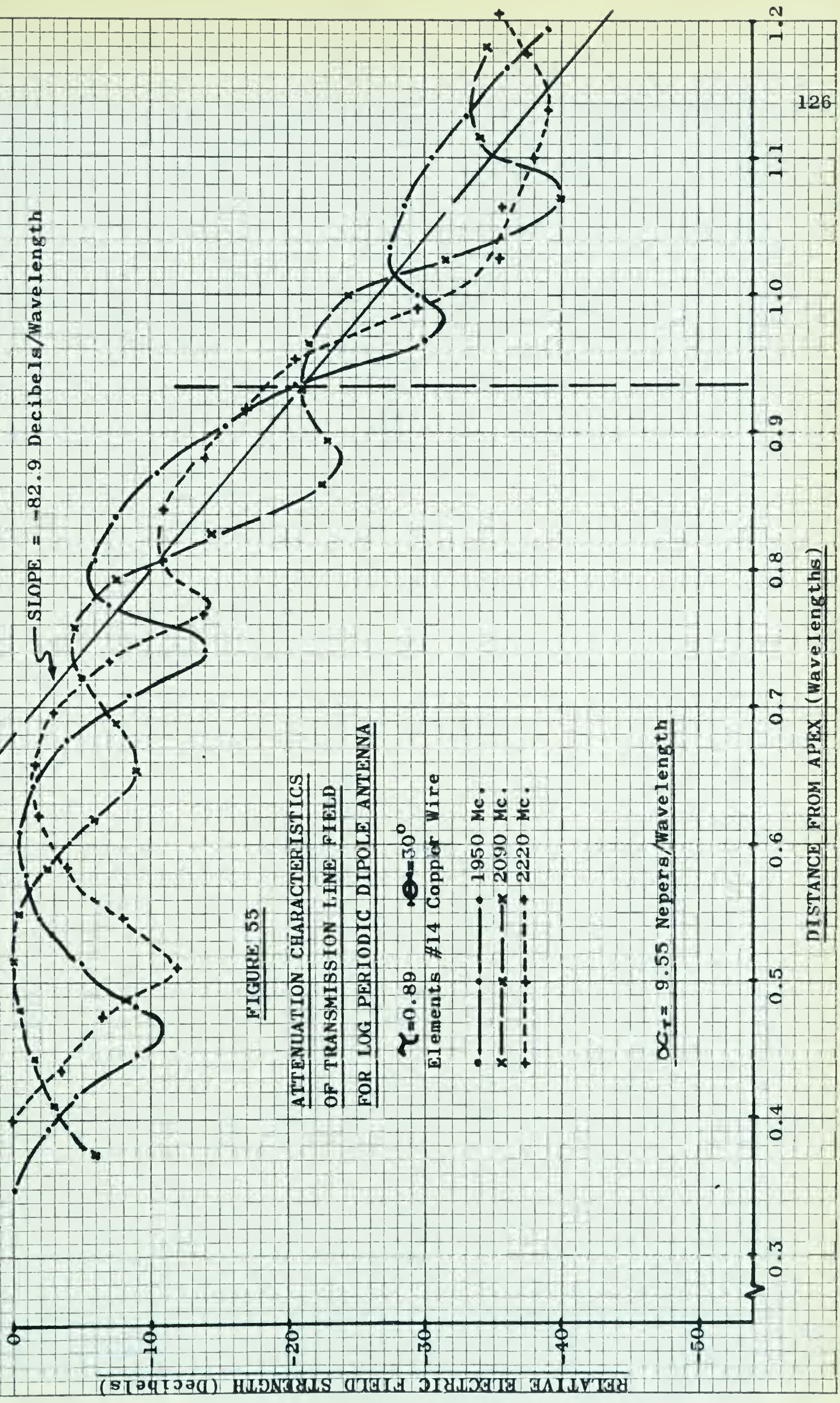


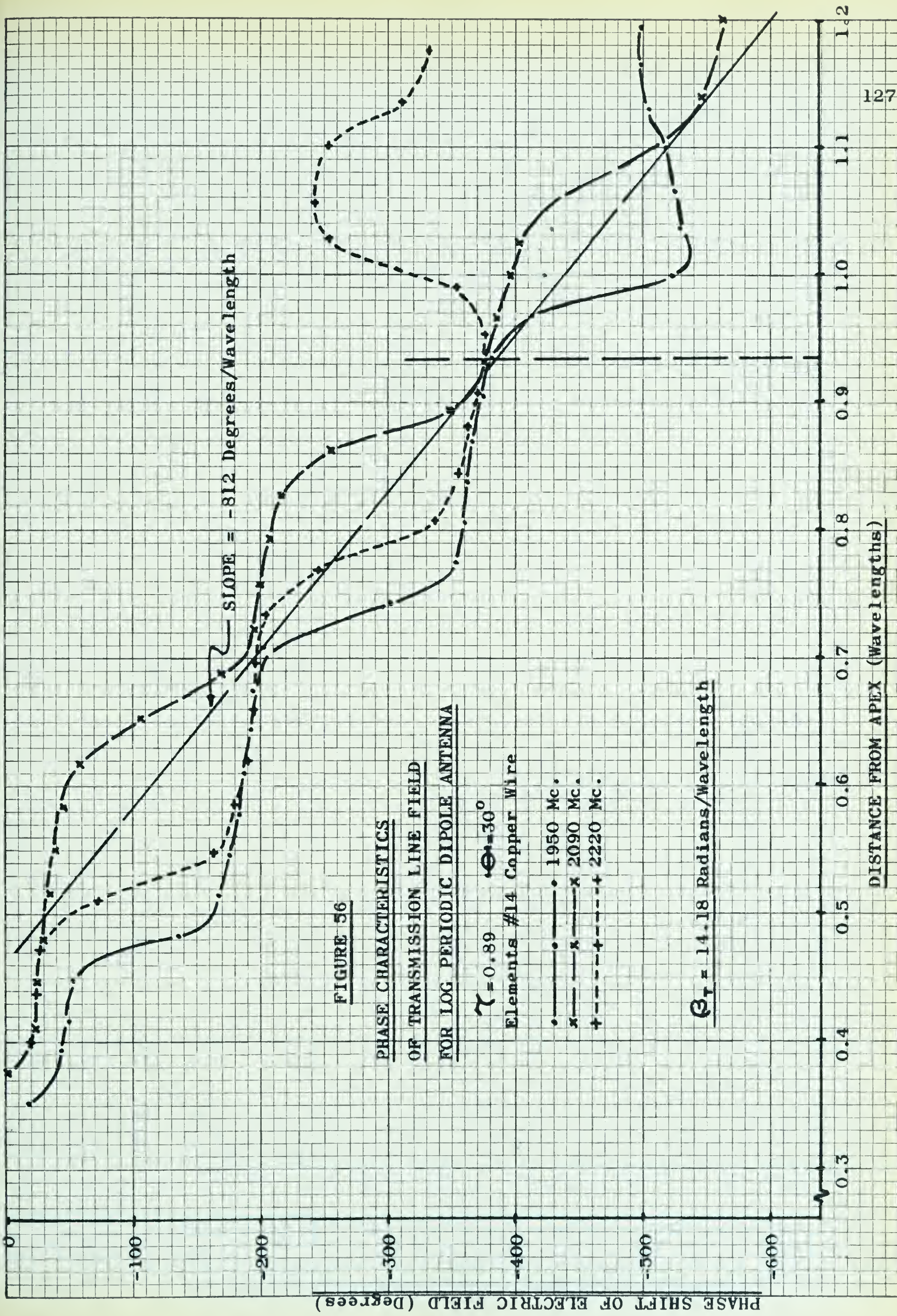
FIGURE 55

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA















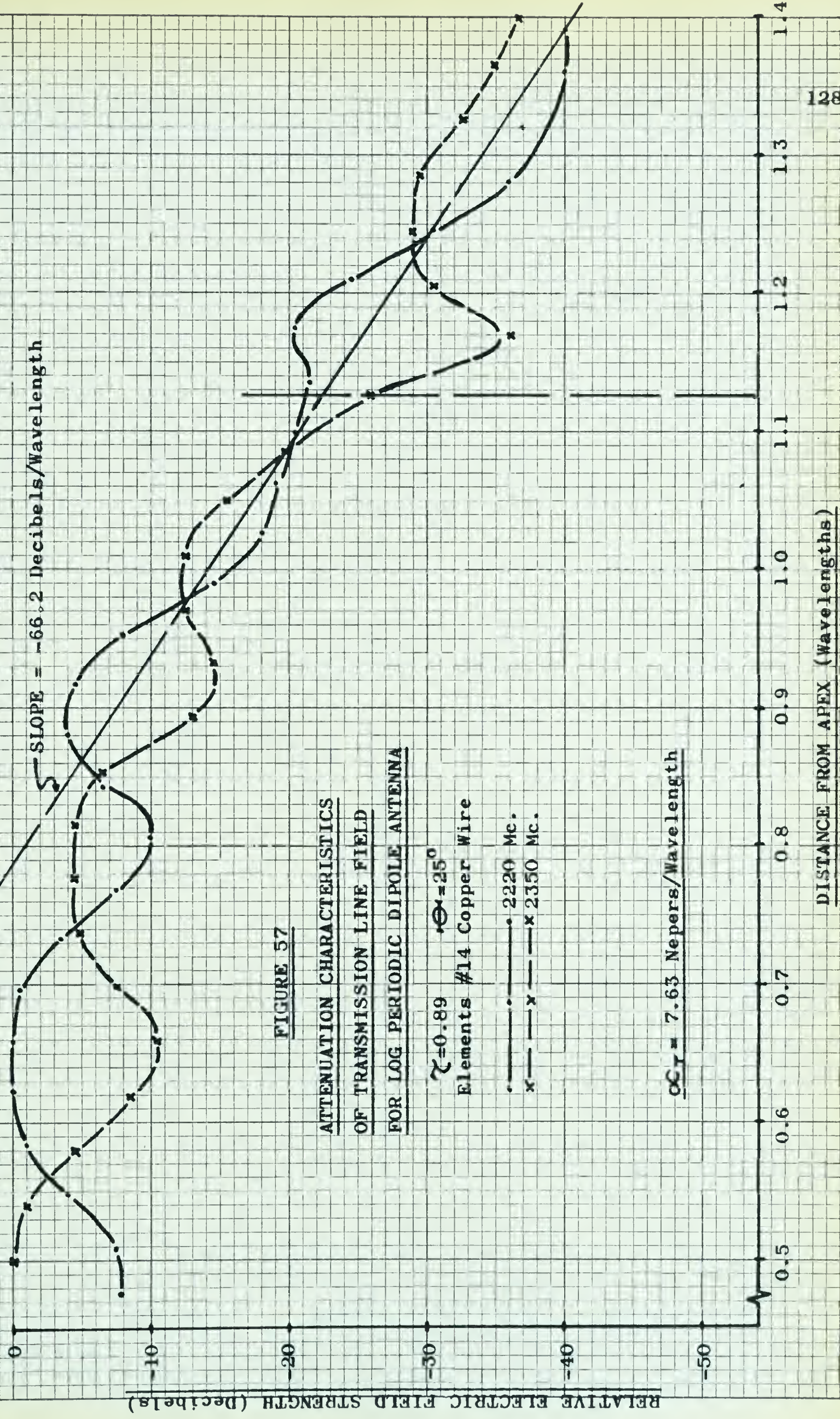
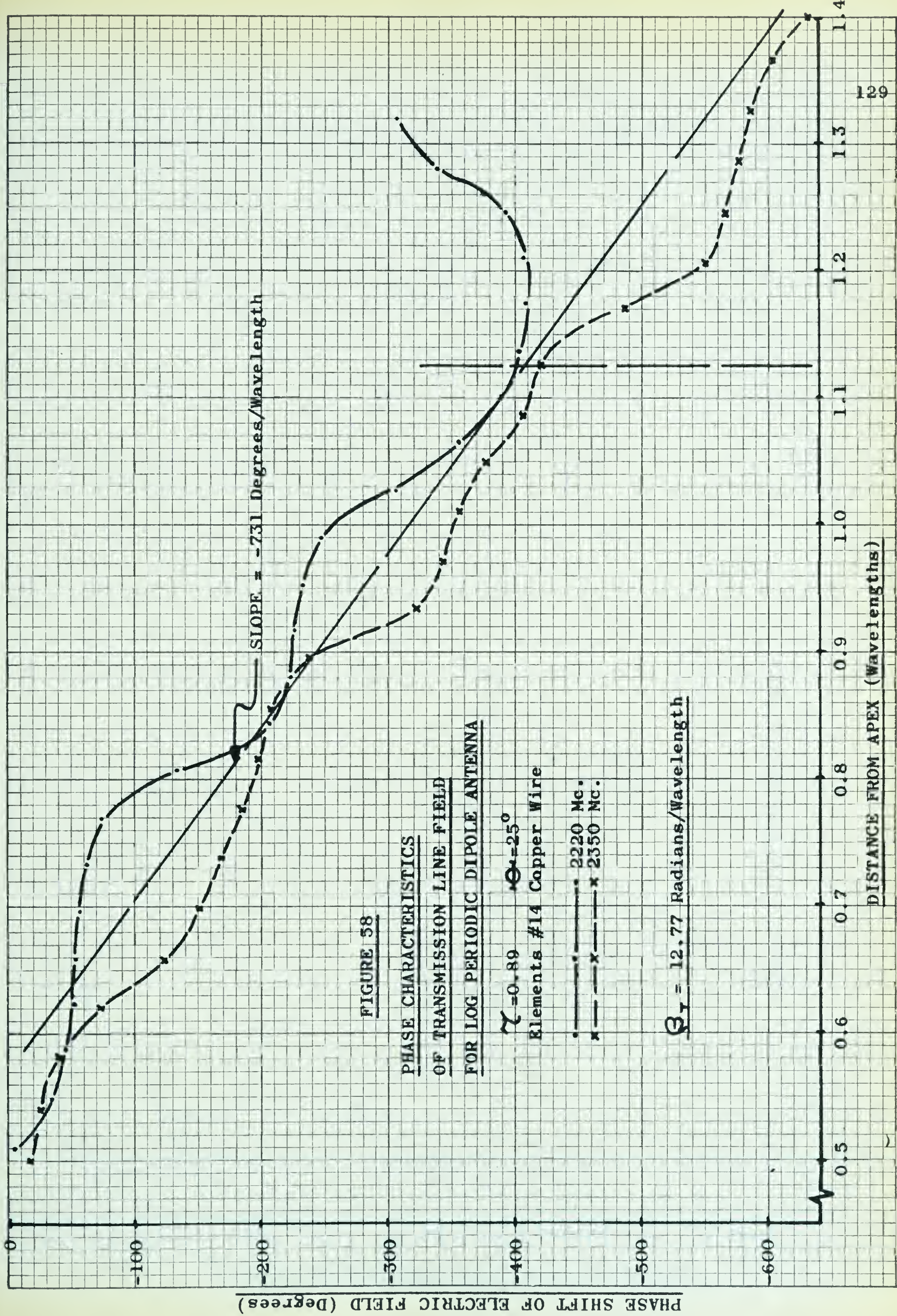


FIGURE 57





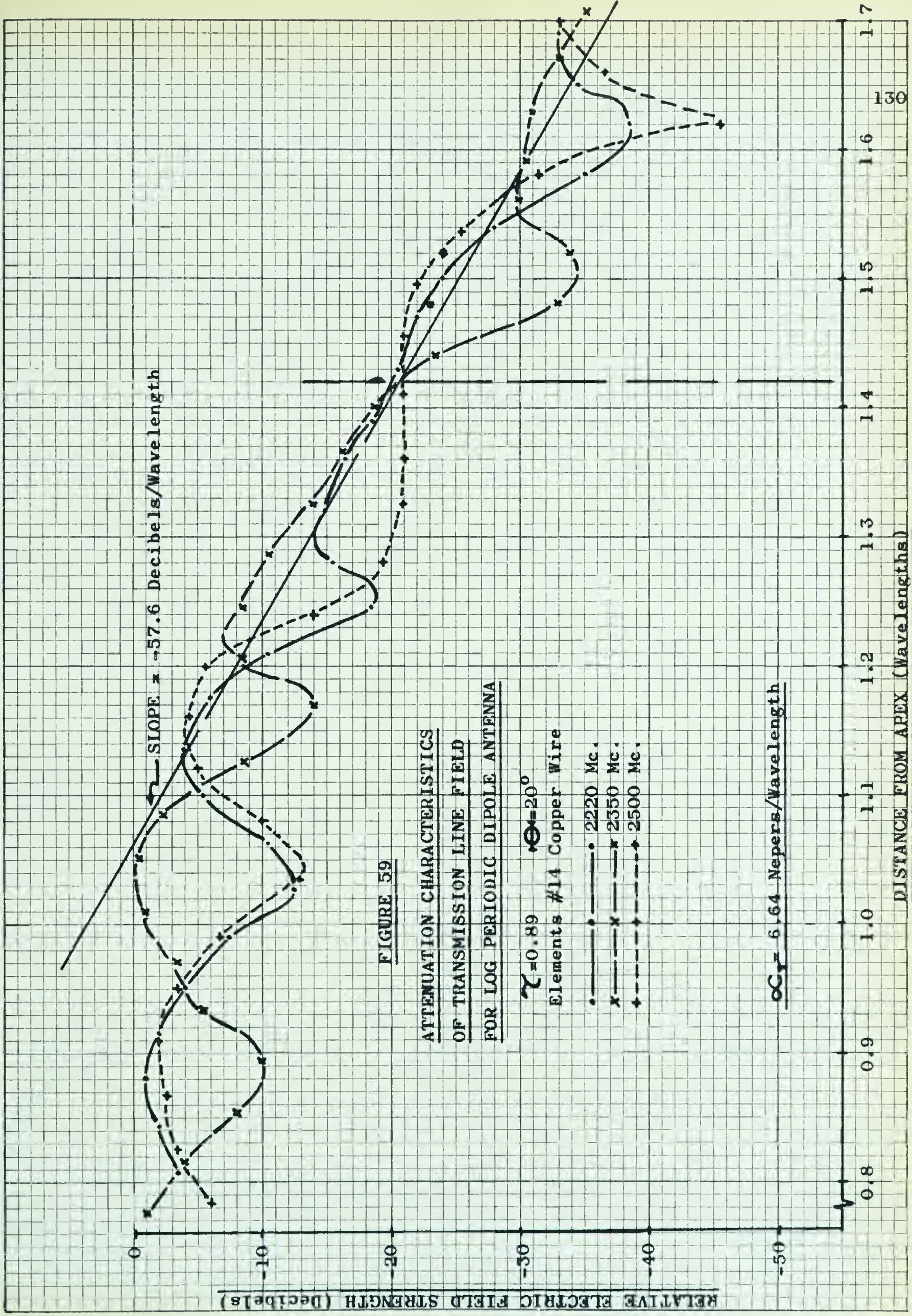








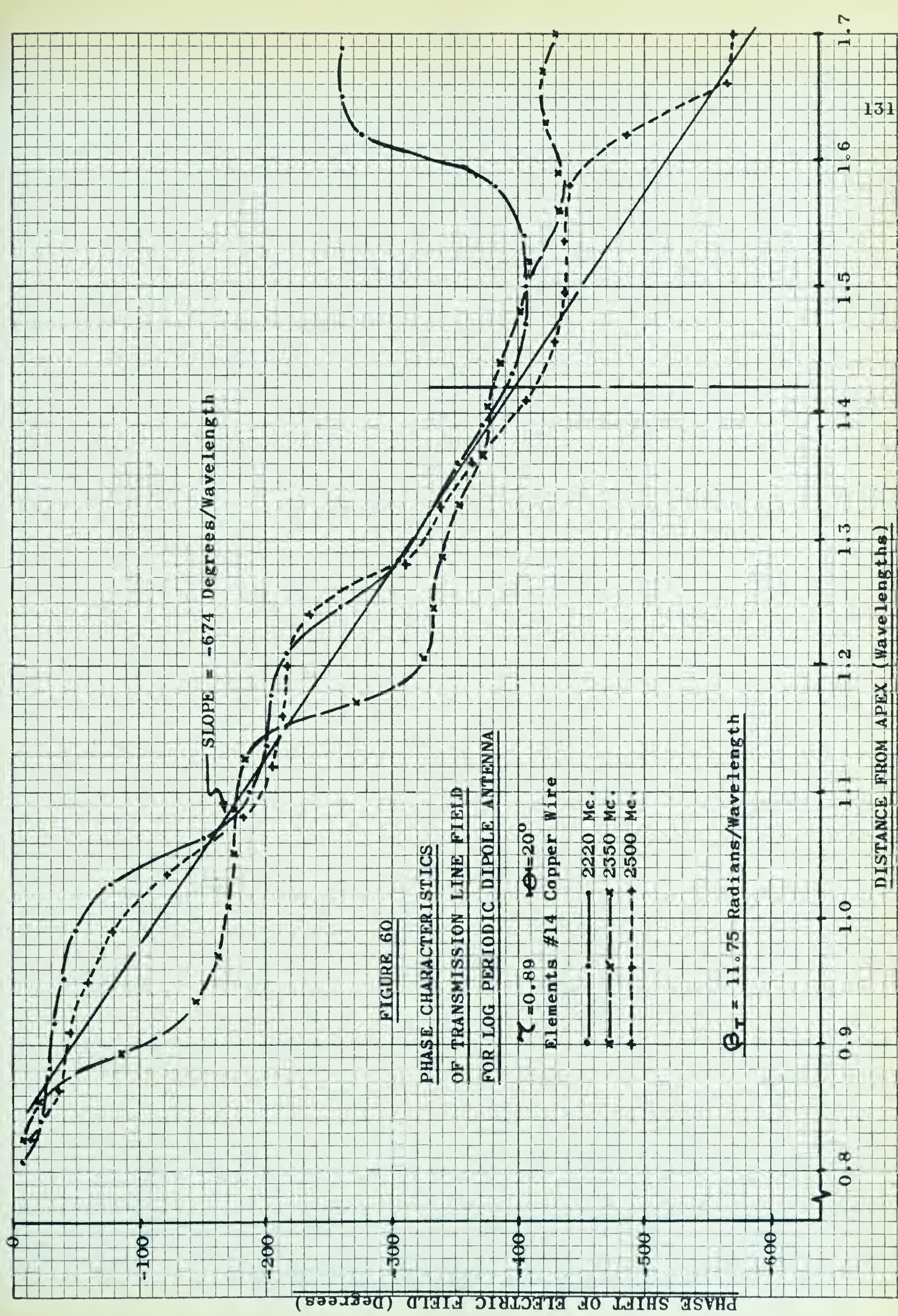


















SLOPE = -46.3 Decibels/Wavelength

FIGURE 61

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.99$   $\theta = 15^\circ$   
Elements #14 Copper Wire

—•— 2650 Mc.  
—x— 2810 Mc.

$OC_r = 5.46$  Nepers/Wavelength

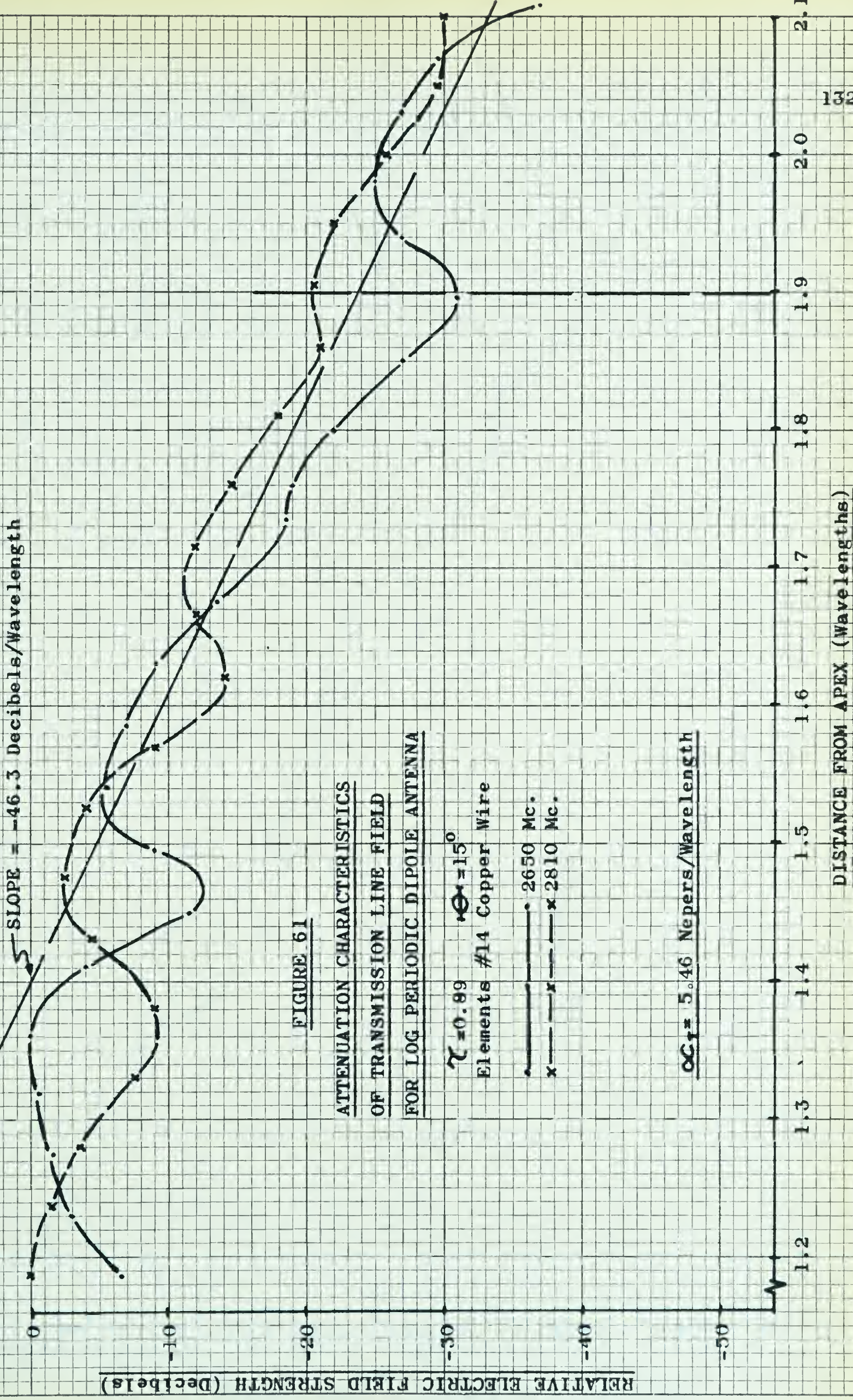








FIGURE 62

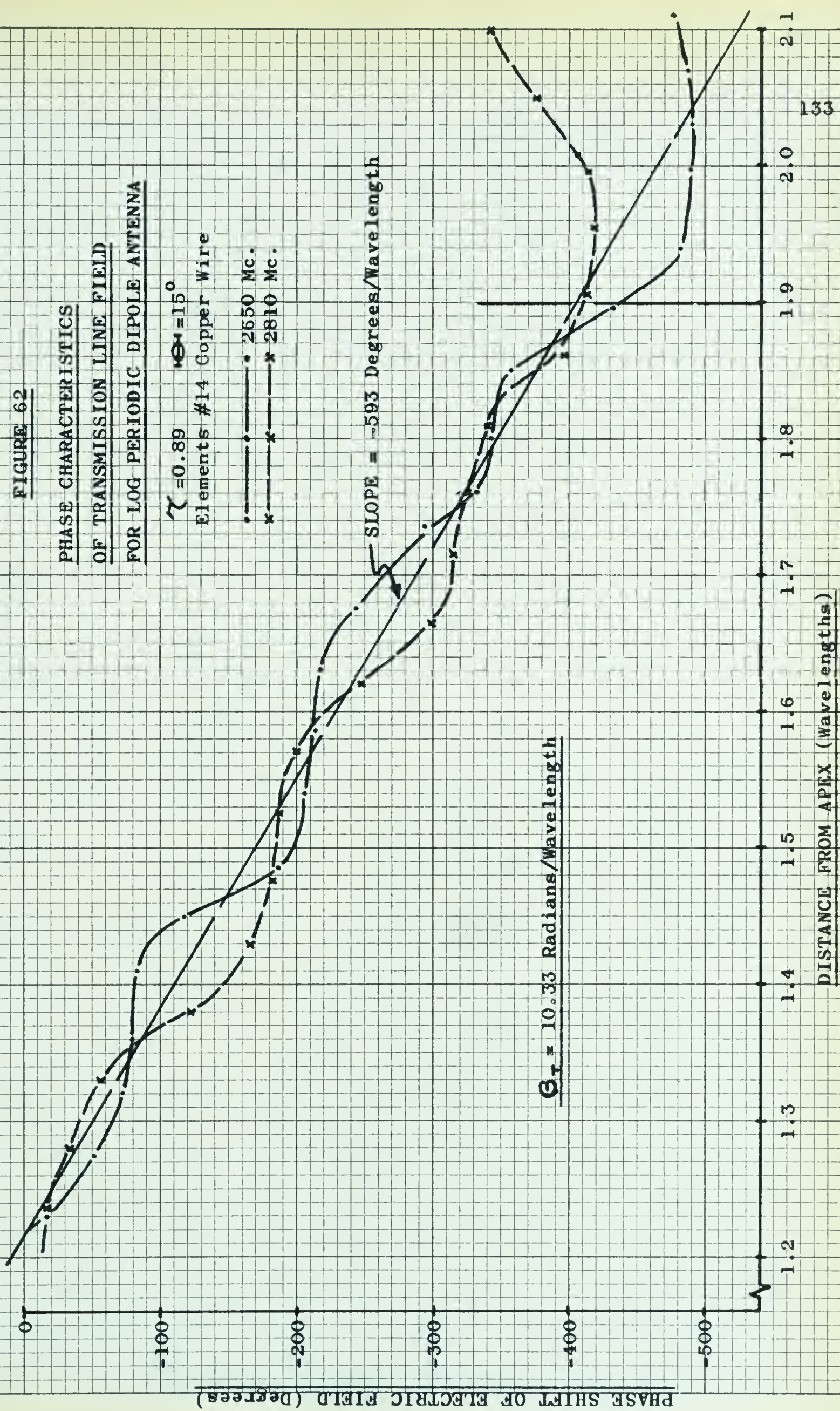
PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.89$   $\theta = 15^\circ$   
Elements #14 Copper Wire

—•— 2650 Mc.  
—x— 2810 Mc.

SLOPE = 593 Degrees/Wavelength

$\theta_T = 10.33$  Radians/Wavelength







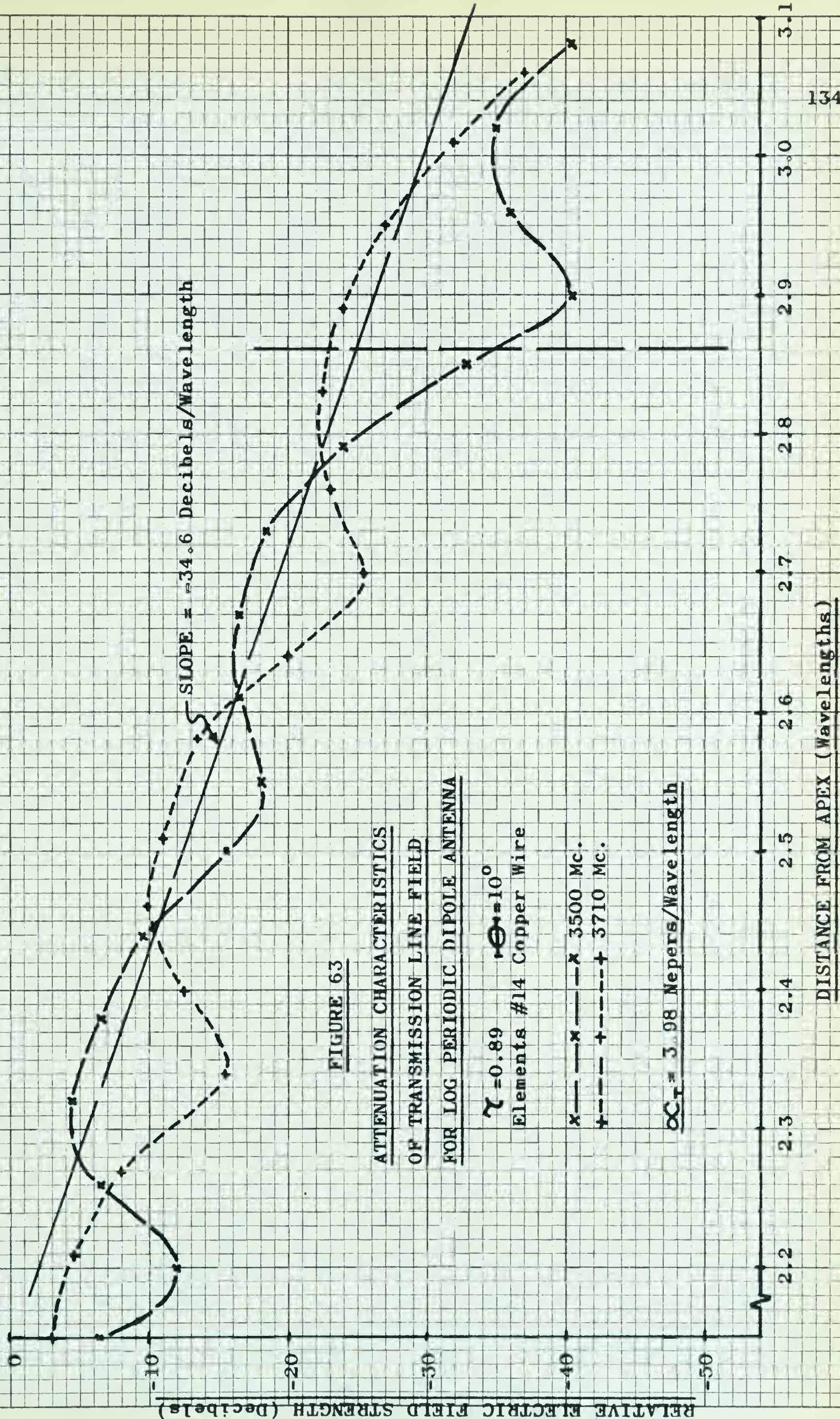


FIGURE 63

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.89$   $\theta = 10^\circ$   
Elements #14 Copper Wire

x — 3500 Mc.  
+ --- 3710 Mc.

$\alpha_T = 3.98$  Nepers/Wavelength



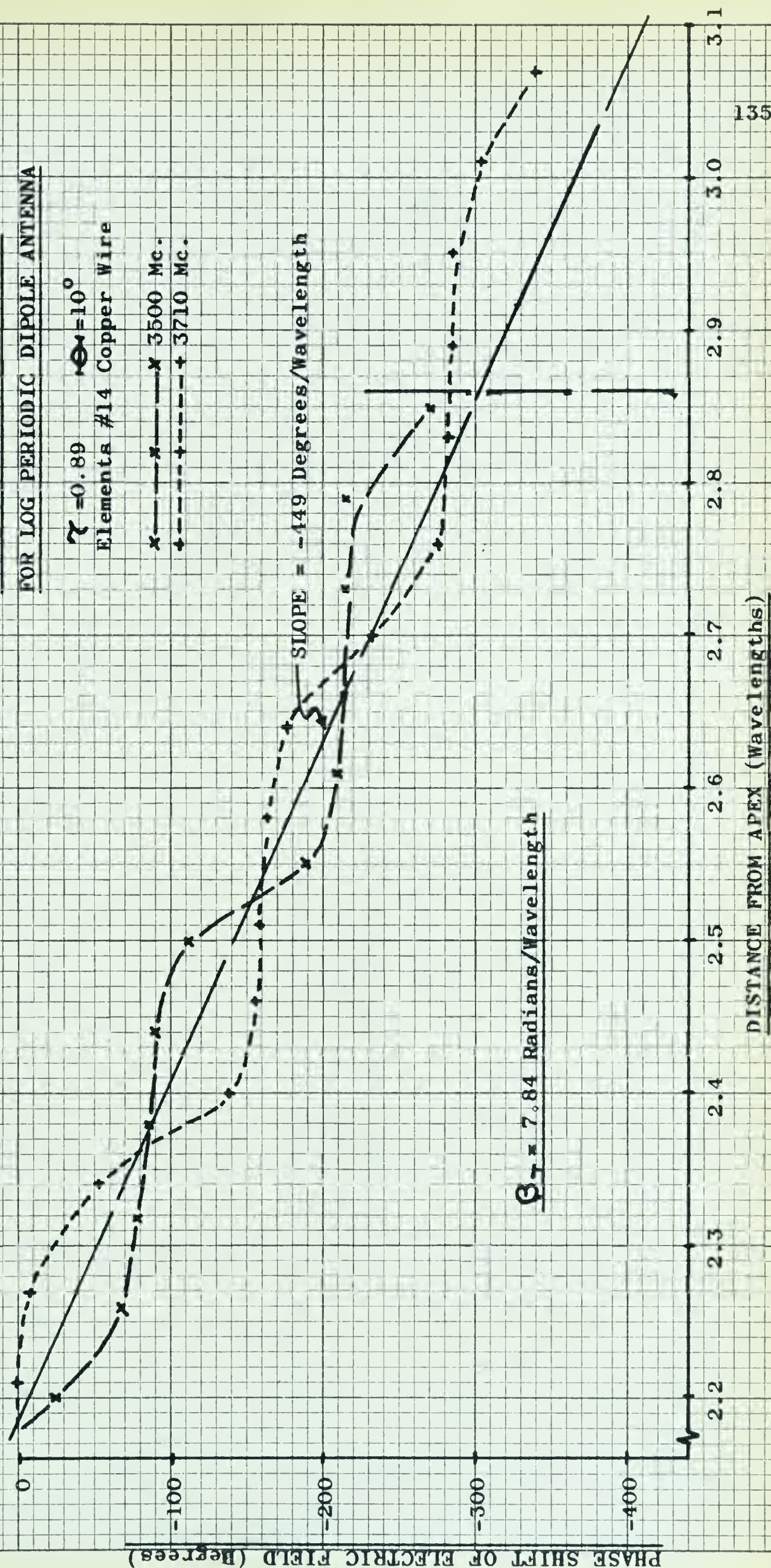




FIGURE 64

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.89$   $\theta = 10^\circ$   
Elements #14 Copper Wire  
 $\times$  ---  $\times$  3500 Mc.  
 $+$  ---  $+$  3710 Mc.









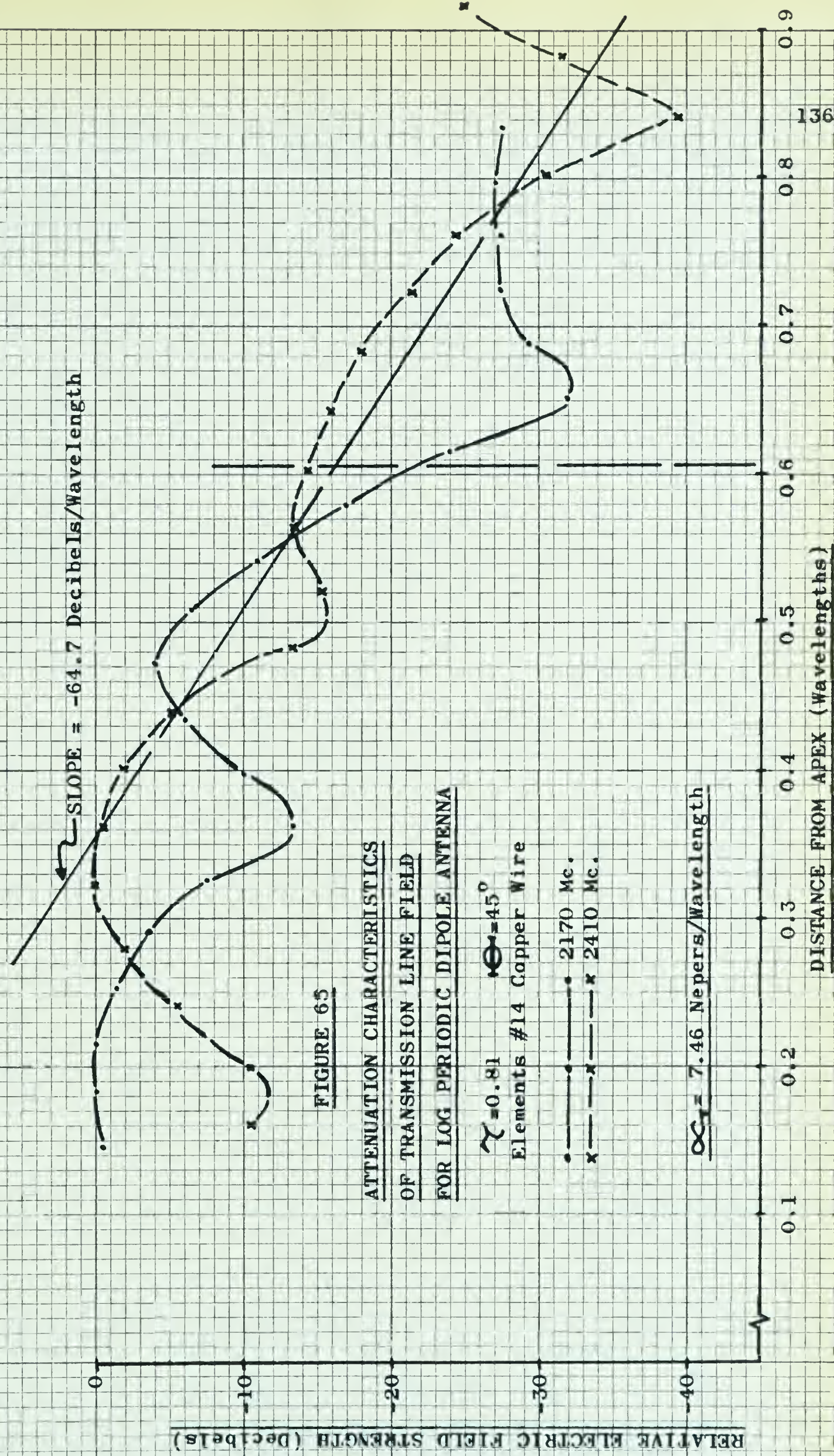




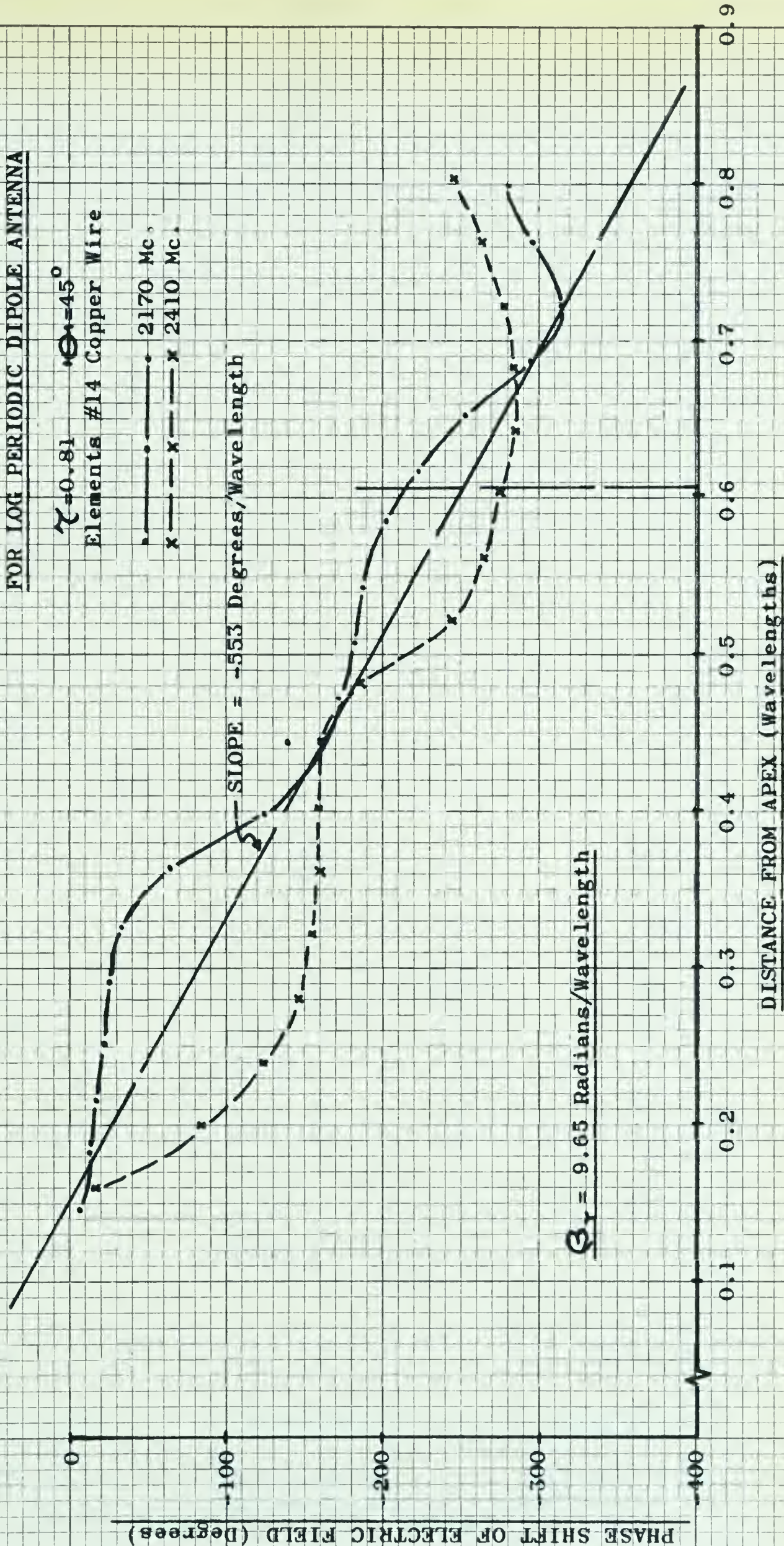




FIGURE 66

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.81$   $\theta = 45^\circ$   
Elements #14 Copper Wire  
 - - - - - 2170 Mc.  
 x x x x x 2410 Mc.









SLOPE = -49.0 Decibels/Wavelength

FIGURE 67

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.81$   $\theta = 40^\circ$   
Elements #14 Copper Wire

- 2170 Mc.
- x— 2300 Mc.
- +— 2410 Mc.

$\alpha_{\gamma} = 5.65$  Nepers/Wavelength

DISTANCE FROM APEX (Wavelengths)







FIGURE 68

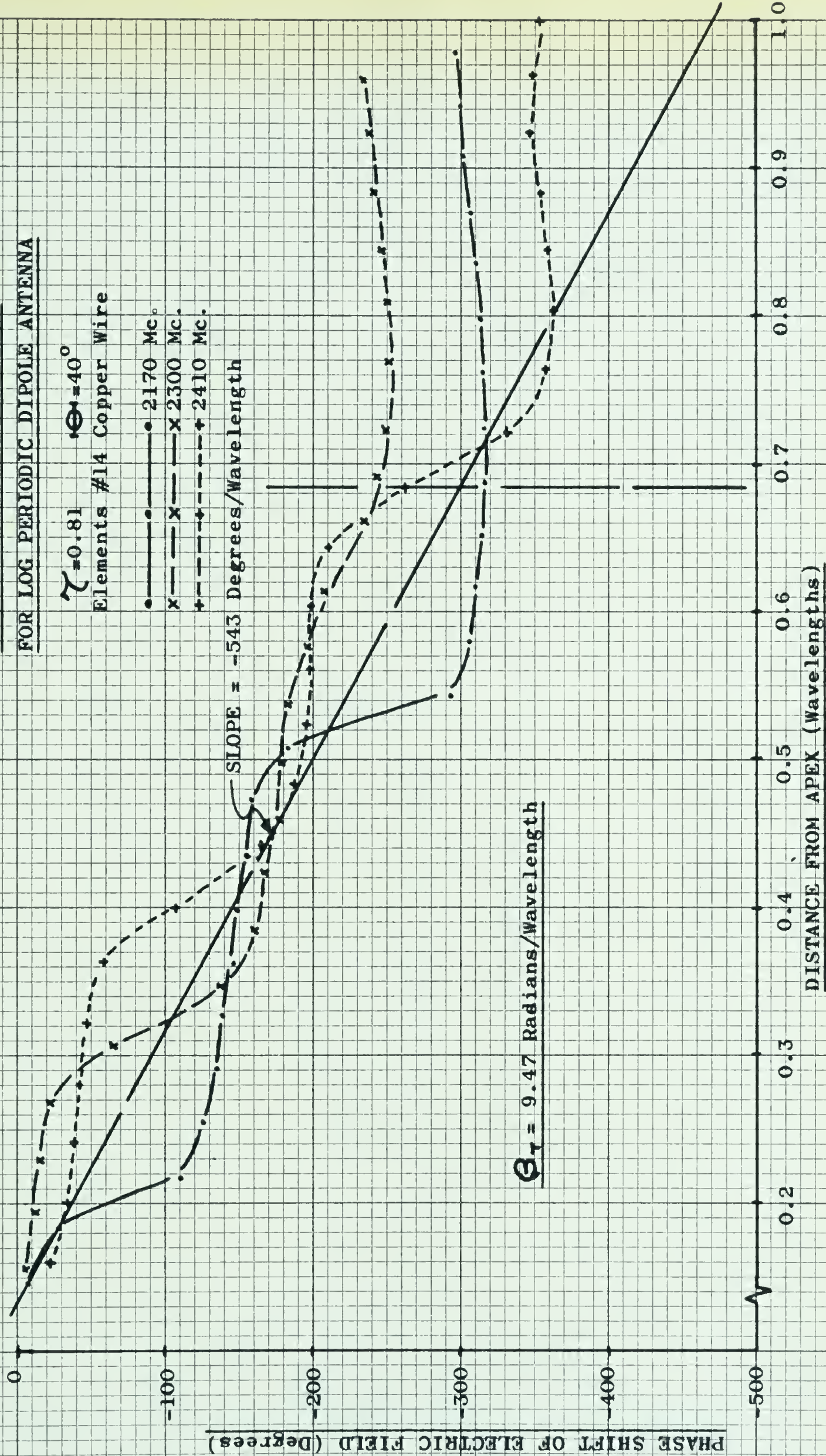
PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\tau = 0.81$   $\Theta = 40^\circ$   
Elements #14 Copper Wire

- 2170 Mc.
- x— 2300 Mc.
- +— 2410 Mc.

SLOPE = -543 Degrees/Wavelength

$\Theta_T = 9.47$  Radians/Wavelength







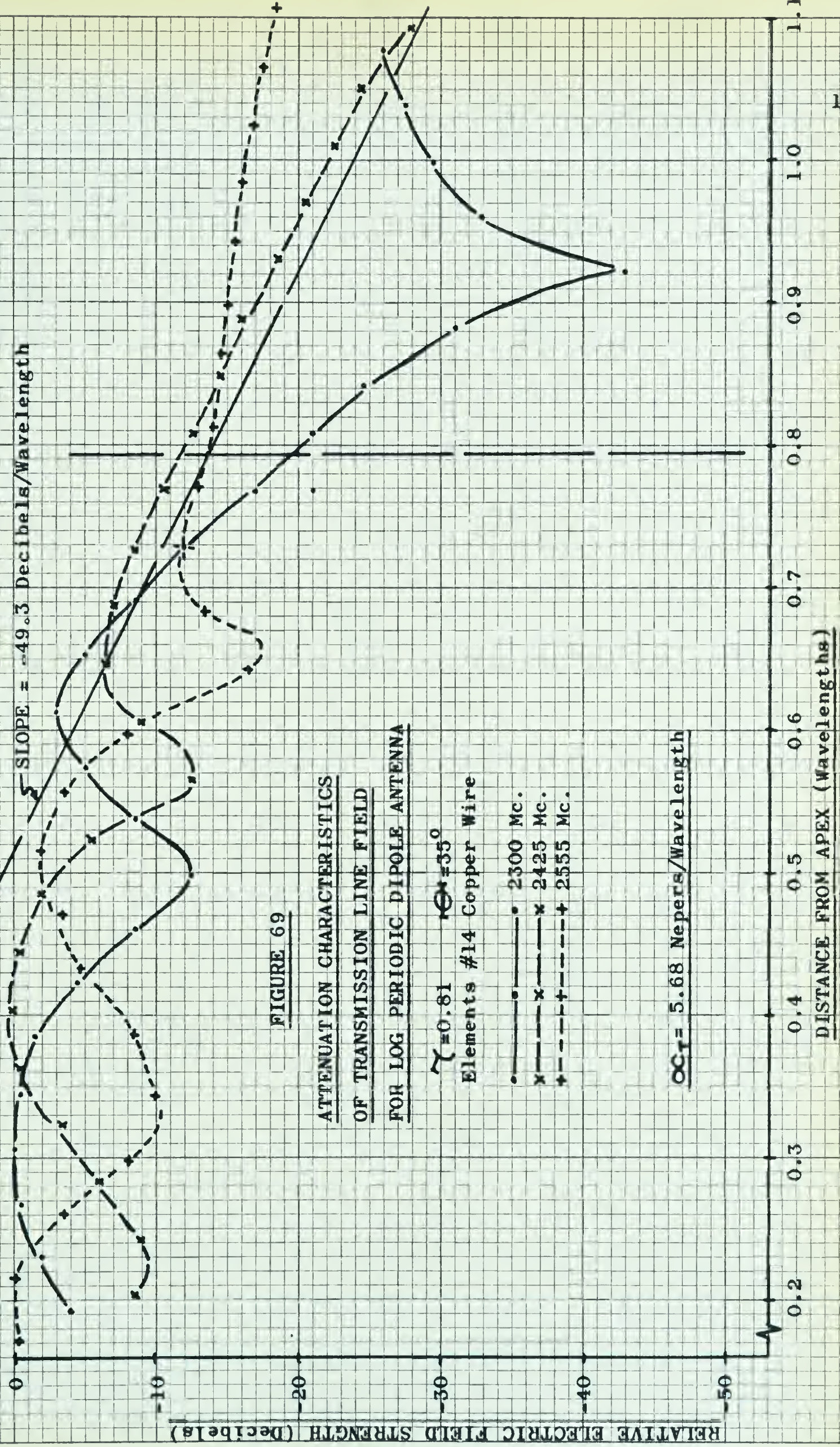


FIGURE 69

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.81$   $\theta = 35^\circ$   
Elements #14 Copper Wire

$OC_T = 5.68$  Nepers/Wavelength





FIGURE 70

PHASE CHARACTERISTICS

OF TRANSMISSION LINE FIELD

FOR LOG PERIODIC DIPOLE ANTENNA

$\tau = 0.81$   $\theta = 35^\circ$   
Elements #14 Copper Wire

- 2300 Mc.
- x— 2425 Mc.
- +— 2555 Mc.

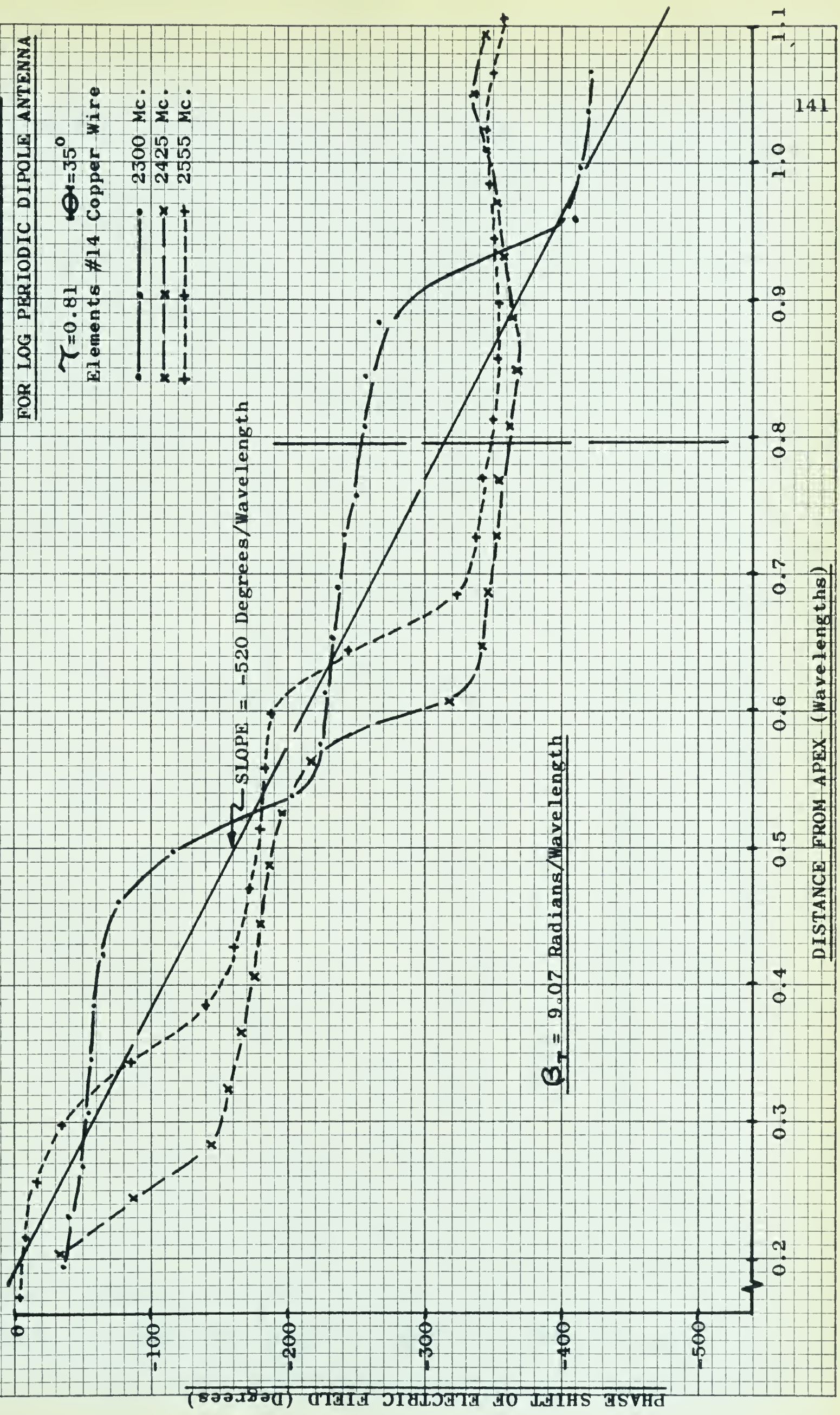






FIGURE 71

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\alpha$  SLOPE = -48.6 Decibels  
Per Wavelength

RELATIVE ELECTRIC FIELD STRENGTH (Decibels)

$\tau = 0.81$   $\theta = 30^\circ$

Elements #14 Copper Wire

- 2300 Mc.
- - - x - - - 2425 Mc.
- - - + - - - 2555 Mc.

$\infty \tau = 5.60$  Nepers/Wavelength

DISTANCE FROM APEX (Wavelengths)







FIGURE 72

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\tau = 0.81$   $\theta = 30^\circ$   
Elements #14 Copper Wire

- 2300 Mc.
- x-x-x- 2425 Mc.
- +--+ 2555 Mc.

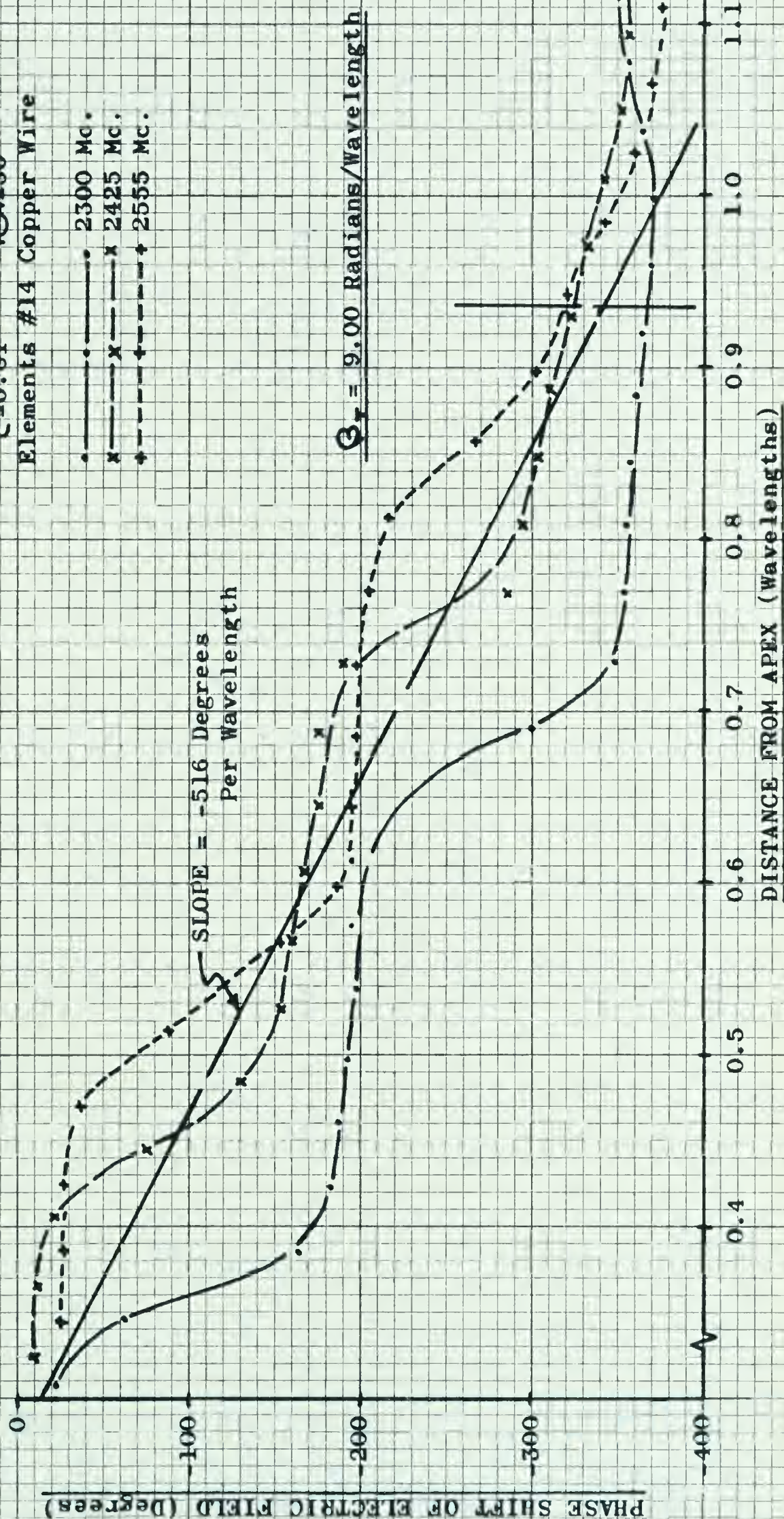








FIGURE 73

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$$\gamma = 0.81 \quad \theta = 25^\circ$$

Elements #14 Copper Wire

- x --- x 2425 Mc.
- --- • 2500 Mc.
- + --- + 2555 Mc.

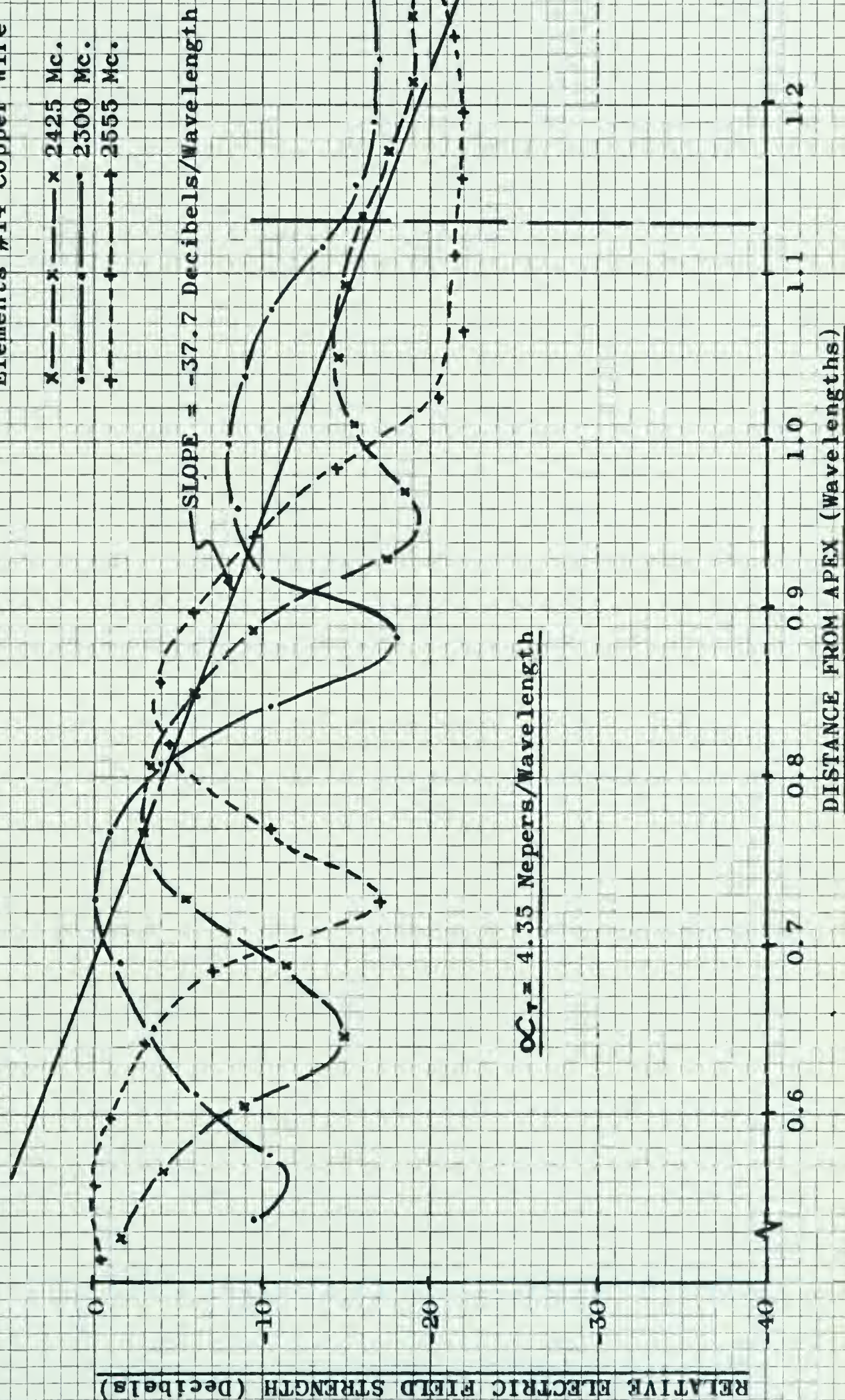








FIGURE 74

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.81$   $\theta = 25^\circ$   
Elements #14 Copper Wire

- 2300 Mc.
- x— 2425 Mc.
- +— 2555 Mc.

SLOPE = -525 Degrees/Wavelength

$\theta_T = 9.16$  Radians/Wavelength

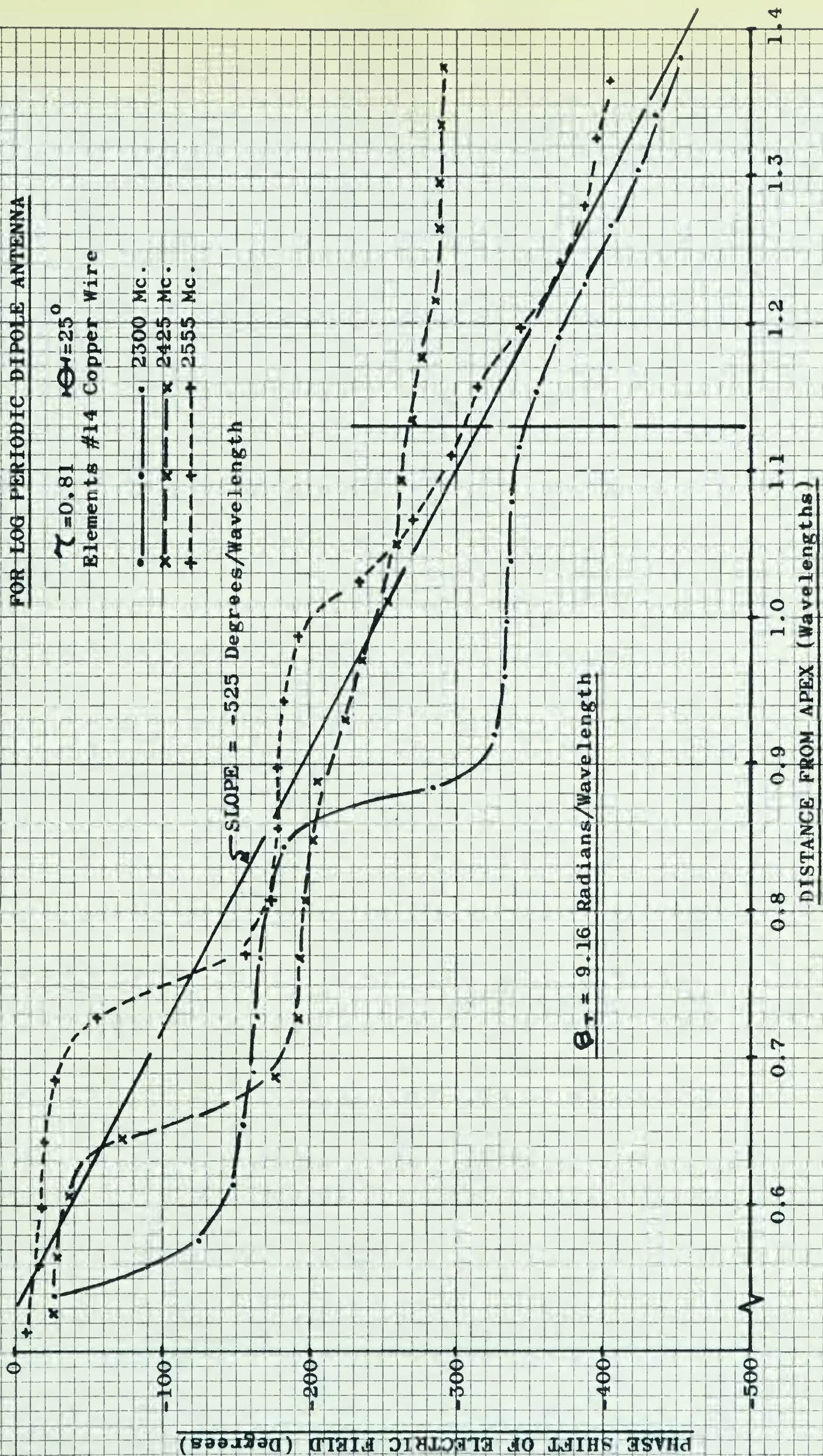






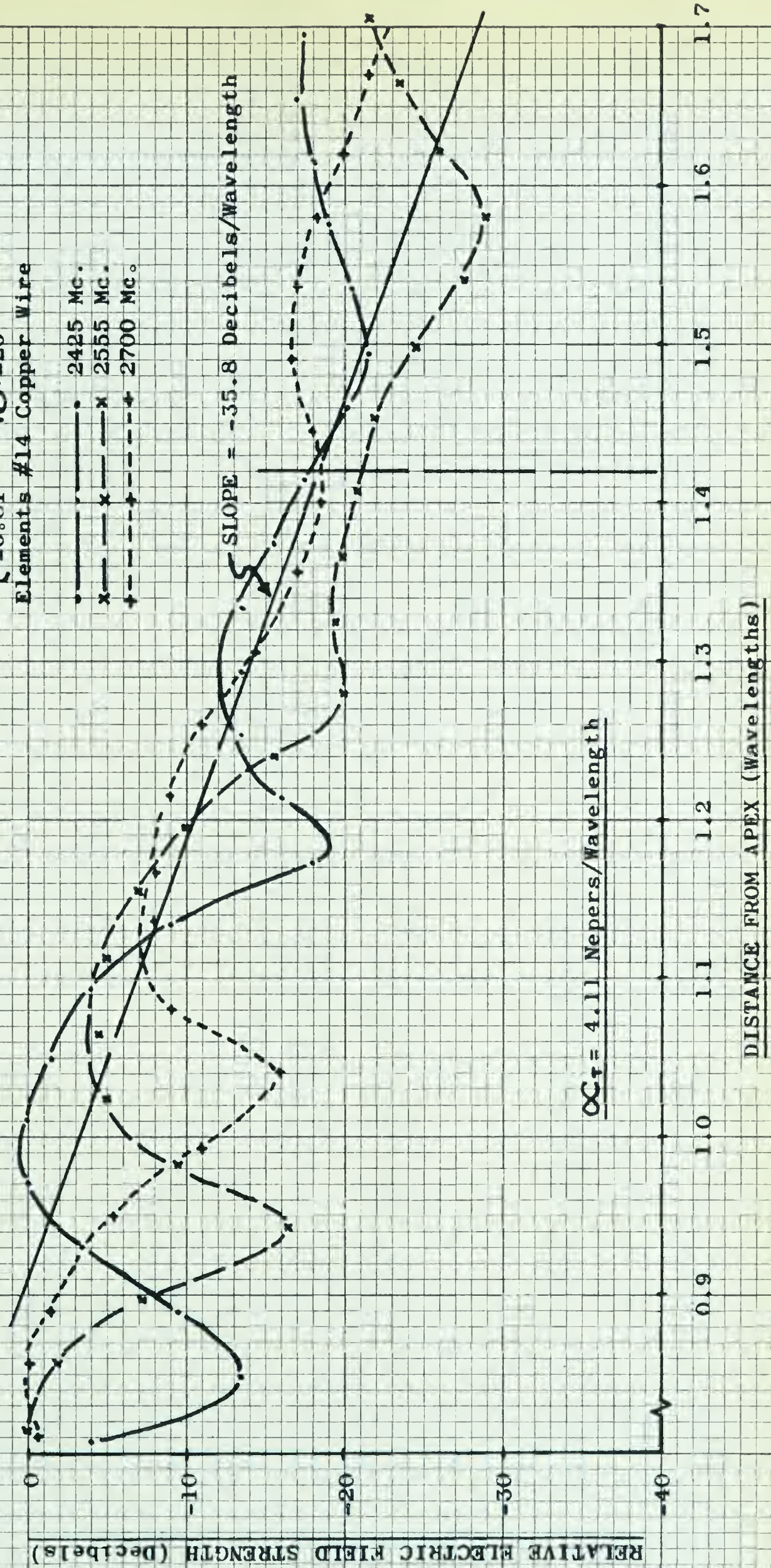


FIGURE 75

ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.81$   $\theta = 20^\circ$   
Elements #14 Copper Wire

- 2425 Mc.
- x— 2555 Mc.
- +— 2700 Mc.



$\alpha_r = 4.11$  Nepers/Wavelength

DISTANCE FROM APEX (Wavelengths)







FIGURE 76

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\alpha = 0.81$   $\theta = 20^\circ$   
Elements #14 Copper Wire

—•—	2425 Mc.
—x—	2555 Mc.
—+—	2700 Mc.

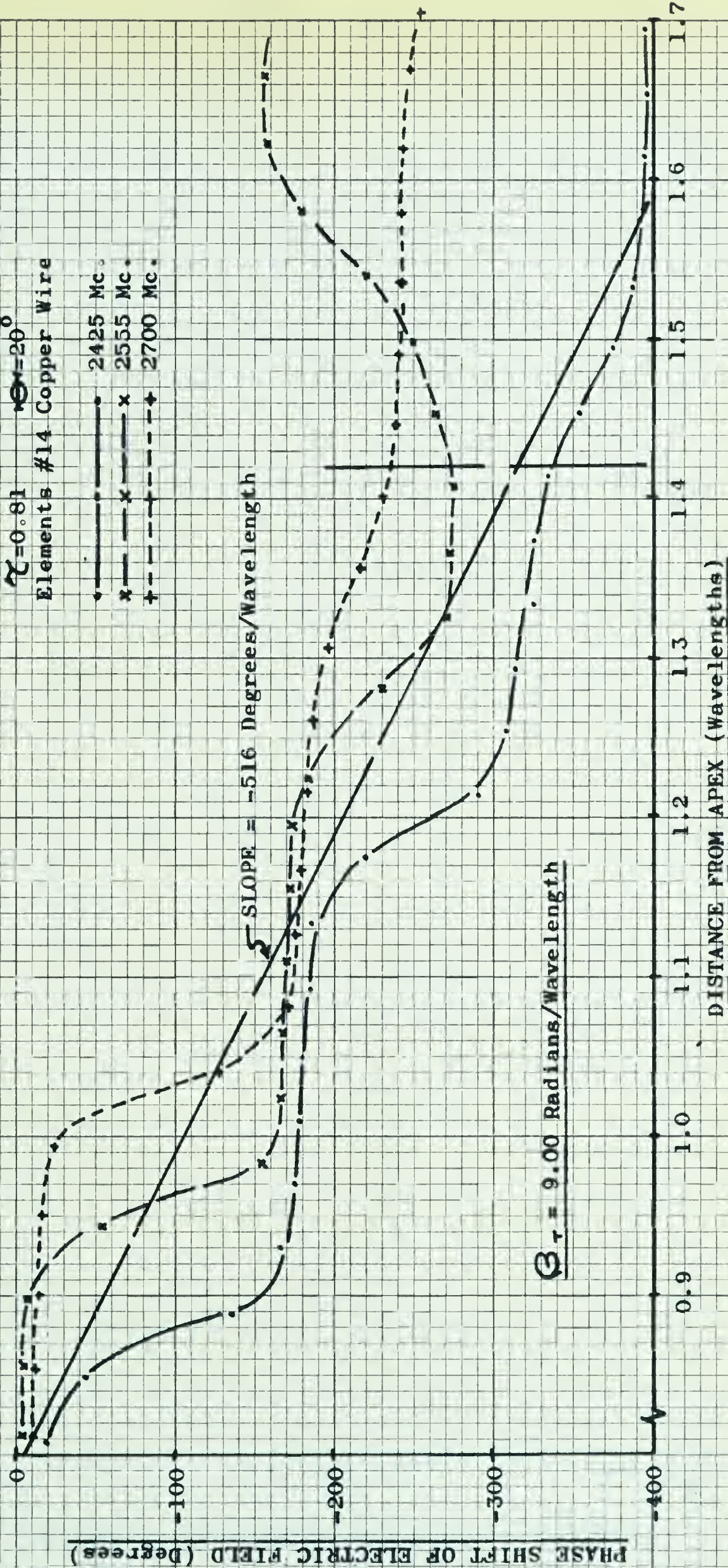








FIGURE 77

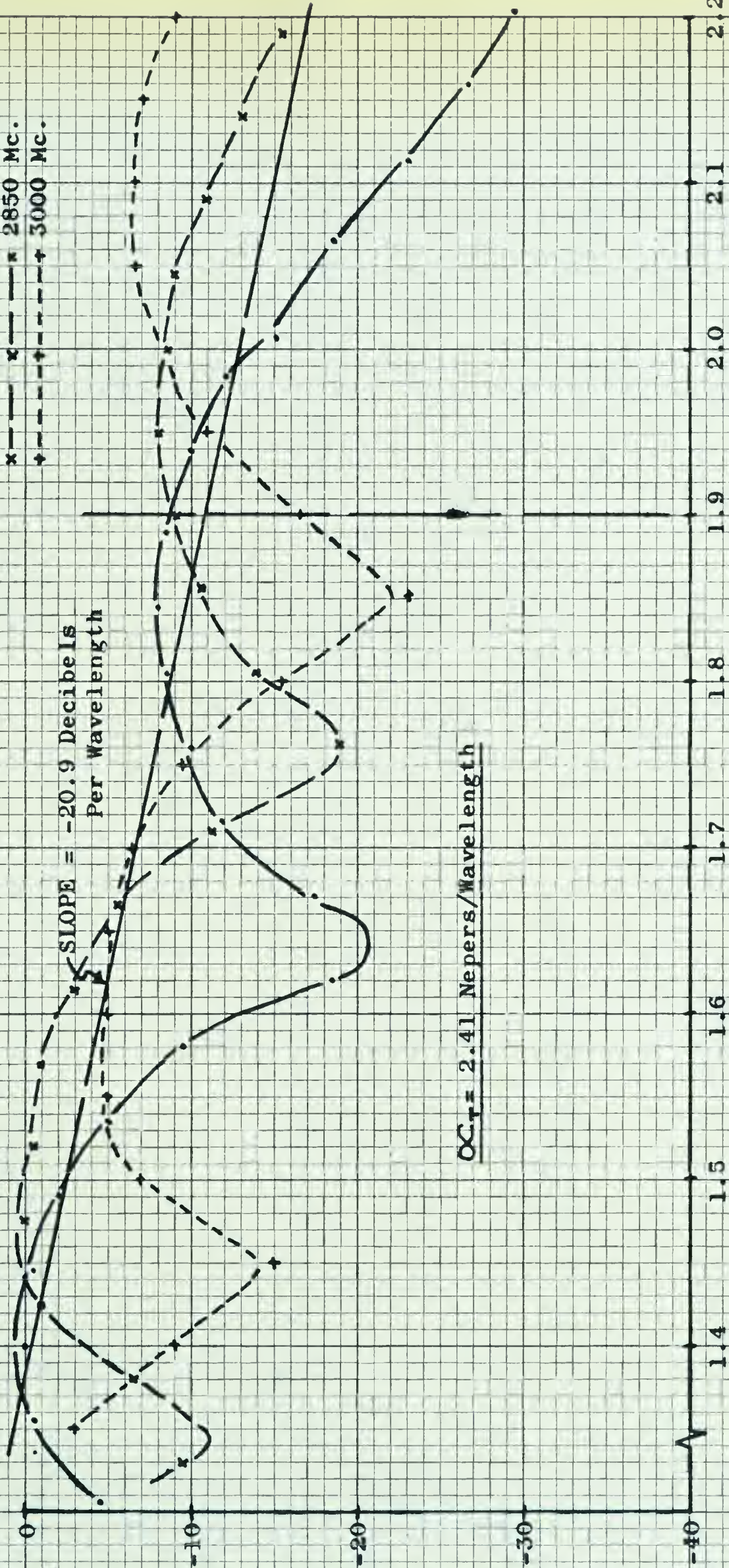
ATTENUATION CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.81$   $\theta = 15^\circ$

Elements #14 Copper Wire

- 2700 Mc.  
-x- 2850 Mc.  
-+-+ 3000 Mc.

RELATIVE ELECTRIC FIELD STRENGTH (Decibels)



DISTANCE FROM APEX (Wavelengths)







FIGURE 78

PHASE CHARACTERISTICS  
OF TRANSMISSION LINE FIELD  
FOR LOG PERIODIC DIPOLE ANTENNA

$\gamma = 0.81$   $\theta = 15^\circ$   
Elements #14 Copper Wire

- 2700 Mc.
- x— 2850 Mc.
- +— 3000 Mc.

SLOPE = -426 Degrees/Wavelength

$\beta_T = 7.43$  Radians/Wavelength

PHASE SHIFT OF ELECTRIC FIELD (Degrees)

DISTANCE FROM APEX (Wavelengths)



















**B29805**